AFAPL-TR-78-6 Part VIII

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ROTOR-BEARING DYNAMICS TECHNOLOGY DESIGN GUIDE
Part VIII A Computerized Retrieval System for Fluid
Film Bearings

SHAKER RESEARCH **BALLSTON LAKE, NEW YORK 12019**

OCTOBER 1980

TECHNICAL REPORT AFAPL-TR-78-6, Part VIII Interim Report for Period March 1979 - March 1980

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This technical report has been reviewed and is approved for publication.

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9. PERFORM & ORGANIZATION NAME AND ADDRESS	10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS
Shaker Research Corporation	3048 26 85
Northway 10, Executive Park	3040 30 03
Ballston Lake, N.Y. 12019 11. CONTROLLING OFFICE NAME AND ADDRESS	112. DEPONT-DATE /
Air Force Wright Aeronautical Laboratory/POSL	Octo-1980
Wright-Patterson Air Force Base, Ohio 45433	13. NUMBER OF PAGES
	239
14. MONITORING AGENCY NAME & ADDRESS(II dillerent from Controlling Office)	15. SECURITY CLASS. (of this report)
(12 V250	Unclassified
(12) 250	150. DECLASSIFICATION/DOWNGRADING SCHEDULE
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LUIRITY CLASSIFICATION OF THIS PAGE(When Date Entered) the dimensional or the dimensionless form or to generate data lines in the sequence and format directly usable as input to the rotordynamics software described in AFAPL-TR-78-6, Part F. Inertia, compliance and damping effects of the pedestal can be included in the retrieval dynamic characteristics of each bearing.

PREFACE

The work reported herein is for a partial fulfillment of USAF Contract No. AF33615-76-C-2038. Dr. Coda H. T. Pan, the Principal Investigator of the contract, was directly involved in the execution of the technical effort with the assistance of Mr. B. F. Geran, Ms. J. A. Bartlett, and Mr. S. Fiedler. The contract was initiated under Project 3048, "Fuels, Lubrication and Fire Protection," Task 304806, "Aerospace Lubrication," Work Unit 30480685, "Rotor-Bearing Dynamics Design."

The work reported herein was performed during the period March 1979 to March 1980 under the direction of John B. Schrand (AFWAL/POSL) and Dr. James F. Dill (AFWAL/POSL), Project Engineers. The report was released by Shaker Research Corporation in April 1980.

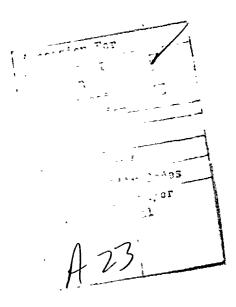


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NOMENCLATURE

a o	near field constant
bo	effective damping constant (lb-sec/in); far field constant
B _{xx} , B _{xy} , etc.	damping coefficients (1b-sec/in)
[B]	damping matrix (lb-sec/in)
\overline{B}_{xx} , \overline{B}_{xy} , etc.	$2\pi C \{\mu LD(R/C)^2\}^{-1} \times (B_{xx}, B_{xy}, etc.)$
[B]	$2\pi C\{\mu LD(R/C)^2\}^{-1} [B]$
С	radial arc clearance (in)
$c_{\mathbf{b}}$	radial bearing clearance (in)
D	journal diameter (in)
е	eccentricity (in)
f	vibrational frequency (Hz)
F _x , F _y	components of bearing reaction force (1b)
F', F'y	components of perturbation bearing reaction (1b)
$\tilde{\mathbf{F}}_{\mathbf{x}}$, $\tilde{\mathbf{F}}_{\mathbf{y}}$	complex amplitudes of F_{x}^{\prime} and F_{y}^{\prime} for simple harmonic motion (lb)
G_x , G_y	complex coefficients in the governing equation for the eigenvector
$\mathbf{J}_{\mathbf{p}}$	pitching mass moment of inertia of a tilting pad
J	$J_{p}N_{s}[\mu(R/C)^{2}(LD)R]$
k _o	effective stiffness constant (!b/in)
K _{xx} , K _{xy} , etc.	stiffness coefficients (lb/in)
[K]	stiffness matrix (lb/in)
\overline{K}_{xx} , \overline{K}_{xy} , etc.	$C\{\mu N_S LD(R/C)^2\}^{-1} \times (K_{xx}, K_{xy}, etc.)$
{ K }	$C\{\mu N_{S}LD(R/C)^{2}\}^{-1} [K]$
L	bearing length (in)

[M]	inertia matrix (lb-sec ² /in)
m	preload, 1-C _b /C
m O	consistent mass (1b-sec ² in)
Ns	rotor speed (rev/sec)
Q _{loss}	volume flow rate of lubricant lost by end leakage (in^3/sec)
Q required	volume flow rate of lubricant (in 3/sec)
Q _{loss} , Q _{required}	(N _s LDC) ⁻¹ (Q _{loss} , Q _{required})
R	journal radius, D/2 (in)
S	Sommerfeld number, $\mu N_S LD(R/C)^2/W$
T	bearing friction torque (in-lb)
Ŧ	$4TC/(\mu N_s D^3 L)$
u _o	amplitude of o
U _x , U _y	real parts of G_{x} and G_{y} , respectively
v _x , v _y	imaginary parts of G_{x} and G_{y} , respectively
<u>w</u>	vector symbol for (δ_x, δ_y)
<u>w</u> o	vector symbol for (\$\delta_{xo}, \delta_{yo})
W	static load magnitude (lbs)
v	1/S, static equilibrium load parameter
х, у	Cartesian coordinates of the journal bearing in the radial plane; x is along the direction of load, rotation is counterclockwise
х, ч	global Cartesian coordinates of the rctor in the radial plane
[Z]	[K] = i [B]

load scaling parameter for L/D variation argument of ô xo components of static displacement (in) components of perturbation displacement (in) components of perturbation velocity (in/sec) complex amplitudes of $\hat{\textbf{G}}_x$ and $\boldsymbol{\delta}_y$ for simple harmonic motion (in) e/C, eccentricity ratio of bearing arc e/C_h , eccentricity ratio of assembled bearing viscosity coefficient of lubricant (Reyns) frequency of oscillation (radians/sec) rotational speed (radians/sec) attitude angle (deg) $\frac{2}{\pi} \tan^{-1}(\overline{W})$, load parameter function angular location of pivot or preload line measured from the leading edge of a bearing pad arc angle of bearing pad arc angle of bearing pad (deg) χ

Subscripts

Arg { }

Argument of the complex quantity { }

b rigidly mounted bearing

o natural orbit

p pedestal

(') time derivative

SECTION I

INTRODUCTION

Reliable prediction of the dynamic behavior of a roter system supported by fluid film bearings (e.g., critical speeds and resonant frequencies, damped response to mass imbalance and other forms of dynamic excitation, and threshold of self-excited instability) depends on an accurate knowledge of the dynamic restraining characteristics of each bearing, which are represented by a set of eight dynamic perturbation coefficients (stiffness and damping coefficients with cross-coupling effects). Given the details of the bearing design (e.g., diameter, length, nominal operating clearance, gap geometry), lubricant viscosity and its temperature dependance, operating speed, and the lubricant film temperature, one can in principle compute steady state performance parameters (e.g., minimum gap, bearing friction, lubricant flow rate) and the dynamic perturbation coefficients. Typically, it is necessary to resort to some numerical technique in such computations. In the previous issue of the Rotor-Bearing Dynamics Technology Design Guide, one volume dealt with the calculation of performance parameters and perturbation coefficients for circular arc journal bearings operating with an incompressible lubricant [1]. Since then, other sources for such computations have also become available either as software furnished in its entirety or by allowing the user to access installed software at a computer center [2]. Some of the software have the capability of accommodating special features in the gap geometry. The proliferation efforts of such software has not yet reached a well-defined trend. Even with the availability of such software, the computation of a set of data needed as input for a rotor-bearing dynamic analysis is not a totally trivial matter. The storage memory requirement is typically quite large. Accuracy of computed data is often sensitive to details in the input preparation which may be quite obscure to an inexperienced user. For these two reasons, it was decided not to link up the bearing computer software with the rotor dynamics software. Instead, as a part of the present effort to update the Rotor-Bearing Dynamics

Technology Design Guide, it was decided that the immediate attention should be directed toward the esta! :hment of a procedure for the convenient extraction of the dynamic perturbation coefficients from a prepared data table. A basic data bank consisting of thirty-one tables for various incompressible fluid film bearings is described in another part of the new Design Guide [3]. The data bank can be readily appended with additional tables if so desired. The required procedures for extracting the desired dynamic perturbation coefficients and for installing additional data tables are described herein.

Gas lubricated fluid film bearings are also of interest in advanc J aircraft turbo-propulsion systems. Recent technological efforts in this area are primarily concerned with gas bearings of the "foil" variety. Another part in the new Design Guide will be exclusively devoted to gas bearings for machinery applications [4].

SECTION II

DYNAMIC CHARACTERISTICS OF A JOURNAL BEARING

2.1 Static Equilibrium Condition

The dynamic perturbation coefficients of a journal bearing contains the information regarding the mutual interactions between the rotor and the bearing. The perturbation analysis of a dynamic system presumes the existence of a static equilibrium condition, and deals with the response to a dynamics excitation and/or whether the equilibrium condition is stable. Typically, the dynamic perturbation coefficients are dependent on the static equilibrium state. Parameters which define the static equilibrium condition of a journal bearing are illustrated in Fig. 1, which depicts the cross-sectional view of the space between the journal, the portion of shaft which passes through the bearing bushing, and the bushing inner surface, which is shown as a circle for the present purpose.

In the lateral plane, the x-axis of a two-dimensional right-handed Cartesian coordinate system is oriented along the direction of the static load W. Sense of rotation is by convention required to be counterclockwise. Under load, the journal center is displaced from the bearing center. The amplitude of the displacement is called bearing eccentricity e. It is a peculiarity of the fluid film bearing that the displacement vector is not necessarily parallel to the load vector. The angle measured from the load vector to the displacement vector (in the counterclockwise direction) is called the attitude angle ψ .

The overall geometry of the journal bearing is described by three dimensions; namely, the length L, the diameter D, and the clearance C. There is some variance in the literature regarding the connotation of the clearance. In the present document, it represents the radial distance between the journal surface and the bushing surface (or a portion thereof) when the centers of curvature of the two surfaces are made coincident. The operating condition of the journal bearing

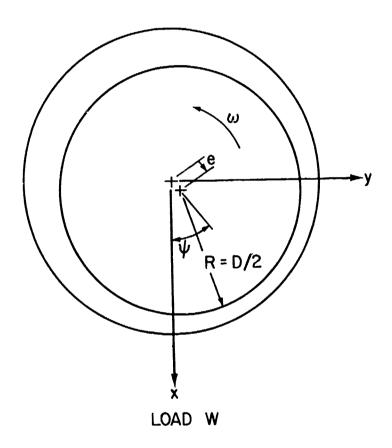


Figure 1 Static Equilibrium Condition of a Journal Bearing

is defined in terms of the rotational speed N $_{
m S}$ (in units of revolution per sec) and the lateral load W (in pounds).

It is also necessary to specify the lubricant viscosity, which in turn is temperature dependent. Actually the lubricant temperature varies not only across the film thickness, but also along the passage through the bearing gap. Since it is impractical to allow for actual temperature dependance of viscosity in the computation of bearing operation characteristics, it is necessary to exer ise a judicious judgment in specifying the effective temperature level. In principle, a heat balance can be performed among friction heat generation, convection cooling by the lubricant flow, and conduction cooling through journal and bushing surfaces. However, uncertainty in the operating temperature level itself is typically sufficient to obscure other considerations. Therefore, for the present purpose, it is recommended that high-low estimates of the operating temperature be used for the determination of the effective lubricant viscosity in lieu of a more thorough treatment of the thermal problem.

In order to allow general applicability to bearings possessing geometrical similarity irrespective of size, clearance to radius ratio, lubricant viscosity, and rotor speed, a data table is compiled in terms of dimensionless variables. Such dimensionless representation lends the data naturally to scaling in accordance with first principles. Among the parameters which represent the static equilibrium condition, specification of the load vector is sufficient. Other parameters are unique functions of the load vector (in the dimensionless representation). The directional parameter of the load vector relates to a reference axis which is peculiar to the bearing geometry. A plain circular bearing does not have such a reference axis due to its rotational symmetry; therefore, the load parameter is complete in terms of amplitude only. For other bearings (e.g., the two-lobe configuration) the direction of the load vector measured from a suitable reference

axis of the particular configuration is also a relevant parameter. The Sommerfeld number

$$S = \frac{\mu N_s LD(R/C,^2)}{W}$$
 (1)

which is a natural parameter in any solution of the lubrication equation for a journal bearing, can be directly utilized to represent the amplitude of the static load. The latter appears in the right-hand side of Eq. (1), while all other quantities are presumably known for a specific problem. Note that the journal diameter D is customarily represented in conjunction with the bearing length L to describe the projected area; at the same time, the journal radius R also appears in a ratio with the clearance C. In fact a bearing engineer often thinks of (LD) and (C/R) as independent design parameters. The reciprocal of the Sommerfeld number is customarily accepted as the dimensionless representation of the static load (amplitude) and is sometimes referred to as the load parameter:

$$\overline{W} = 1/S = \frac{W}{\mu N_S LD(R/C)^2}$$
 (2)

A brief comment on the applicable units is in order in view of the common addiction to "English units" of American and British engineers amidst the worldwide movement toward standardization via the SI units. In the "English" convention, W would be in lbs. while the consistent unit for μ is Reyns = lb-sec/in². If SI units are to be adhered to, then W should be in Newtons (1 N = 0.22482 lb) and μ in Pascal-sec (1 Pa = 1 N/m² = 1.45038 x 10^{-4} lb/in²). For the reader who is more used to cgs units, it may be of interest to note

1 centipoise =
$$1.45038 \times 10^{-7}$$
 Reyns = 0.001 Pa-sec

1 centipoise = $0.001 \text{ Pa-sec} = 1.45038 \times 10^{-7} \text{ Reyns}$

Other operating parameters for the static equilibrium condition are unique functions of S or \overline{W} . Although they do not directly contribute to the dynamic perturbation problems, they are included in the data table for They are defined in the dimensionless representation as follows without further elaboration.

Eccentricity

Pad Eccentricity Ratio
$$\varepsilon$$
 = e/C (3a)

Bearing Eccentricity Ratio
$$\varepsilon_b = e/C_b$$
 (3b)

 $\mathbf{C}_{\mathbf{b}}$ is the largest radius which the journal center can circumscribe in an assembled bearing.

Attitude Angle

ψ (customarily given in deg)

Torque

$$\overline{T} = \frac{4TC}{\mu N_c D^3 L} \tag{4}$$

T is the frictional torque experienced by the bearing.

$$\frac{\text{Lubricant Flow Rate}}{\overline{Q}_{\text{required}}} = \frac{\overline{Q}_{\text{required}}}{\overline{N}_{\text{S}}\text{LDC}}$$
 (5)

$$\overline{Q}_{loss} = \frac{Q_{loss}}{N_{s}LDC}$$

 $Q_{\scriptsize{ ext{required}}}$ is the volume flow rate which passes through the bearing inlet edge. $Q_{\mbox{loss}}$ is the volume flow rate leaving the bearing gap through its ends. The excess of $Q_{required}$ over Q_{loss} is the recirculation flow rate.

2.2 Dynamic Perturbations

The perturbation point of view requires that the deviation of the force of interaction between the rotor and the bearing from the equilibrium value be sufficiently small to permit use of the method of linear superposition in the dynamic analysis. The measure of "smallness" is that the amplitude of the perturbation displacement at any instant be a small fraction of the minimum film thickness. Dynamic analysis requires allowance for time-dependence. In the case of incompressible fluid film bearings, time-dependence is associated with squeeze film effects which correspond to velocity perturbations. Combined displacement and velocity perturbations are illustrated in Fig. 2. Displacement components $(\delta_x^{'}, \delta_y^{'})$ and velocity components $(\delta_x^{'}, \delta_y^{'})$ form a complete set of perturbation parameters. The instantaneous bearing reaction has the components

$$F_{x} = -W + F_{x}$$

$$\mathbf{F}_{\mathbf{y}} = \mathbf{F}_{\mathbf{y}}^{\mathbf{T}}$$

While the static equilibrium displacement is determined by $e=C\epsilon$ and $\psi,$ both ϵ and ψ being unique functions of $\bar{W}=W(C/R)^2/(\mu N_S LD)$, the perturbation hypothesis allows one to express the perturbation reaction components to be

$$F_{x}' = -K_{xx} \delta_{x}' - B_{xx} \delta_{x}$$

$$-K_{xy} \delta_{y}' - B_{xy} \delta_{y}$$
(7a)

$$F_{y}' = -K_{yx} \delta_{x}' - B_{yx} \delta_{x}'$$

$$-K_{yy} \delta_{y}' - B_{yy} \delta_{y}'$$
(7b)

The complete set of perturbation coefficients includes the stiffness coefficients (K_{xx} , K_{xy} , K_{yx} , K_{yy}) and the damping coefficients

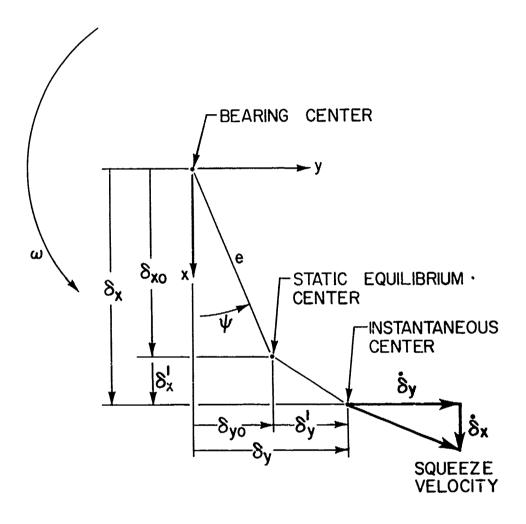


Figure 2 Displacement and Kinematic Perturbation Vectors

 $(B_{xx}, B_{xy}, B_{yx}, B_{yy})$. Double subscripts are necessary because the perturbation reaction is not necessarily co-linear with either the perturbation displacement or the perturbation velocity. The first subscript denotes the direction of the reaction, and the second subscript represents the direction of the perturbation parameter. The stiffness coefficients are associated with displacement perturbations while the damping coefficients are associated with kinematic (velocity) perturbations. It is customary to employ the notations of matrix algebra to write

$$\begin{bmatrix} K \end{bmatrix} = \begin{bmatrix} K_{xx} & K_{xy} \\ K_{yx} & K_{yy} \end{bmatrix}; \quad \begin{bmatrix} B \end{bmatrix} = \begin{bmatrix} B_{xx} & B_{xy} \\ B_{yx} & B_{yy} \end{bmatrix}$$

$$\begin{bmatrix} B \\ B \end{bmatrix} = \begin{bmatrix} B_{xx} & B_{xy} \\ B_{yx} & B_{yy} \end{bmatrix}$$
(8)

The perturbation coefficients are obtained by solving the lubrication equation which is linearized with respect to each of the eight perturbation parameters one at a time, then integrating the film pressure with the appropriate projection of bearing surface.

The perturbation coefficients can also be represented in the dimensionless form to allow dynamic scaling for geometrically similar bearings. Thus,

$$\begin{bmatrix} \overline{K} \end{bmatrix} = \begin{bmatrix} \frac{\mu N_S LD(R/C)^2}{C} & [K] \end{bmatrix}$$

$$\begin{bmatrix} \overline{B} \end{bmatrix} = \begin{bmatrix} \frac{\mu LD(R/C)^2}{2\pi C} & B \end{bmatrix}$$
(9a)
(9b)

$$\begin{bmatrix} \overline{B} \end{bmatrix} = \begin{bmatrix} \mu LD(R/C)^{2} & \overline{1} & \overline{1} \\ 2\pi C & B \end{bmatrix}$$
 (9b)

From Eq. (2) one can write

$$\mu N_S LD(R/C)^2 = WS$$

Therefore, it may also be written

$$\left[\overline{K}\right] = \left(\frac{C}{WS}\right) \left[K\right]; \quad \left[\overline{B}\right] = \left(\frac{\omega C}{WS}\right) \left[B\right]$$
 (10)

 $\lceil K \rceil$ and $\lceil B \rceil$ clearly are dependent on the static equilibrium parameter \overline{W} .

2.3 Symmetry Law for the Damping Matrix

It has been shown that the damping matrix is symmetrical provided the boundary conditions of the perturbation problem are time invariant [5]; that is

$$\overline{B}_{yx} = \overline{B}_{xy} \tag{11}$$

Bearings with fixed film boundaries or with a cavitation break up boundary determined by the Swift-Stieber condition [6, 7] inherently obey this symmetry relationship. Some approximate computation schemes may not conform to this requirement but otherwise yield fairly suitable results [3]. For this reason, the data retrieval system enforces this symmetry condition by substituting

$$(\overline{B}_{xy})_{adj} = (\overline{B}_{yx})_{adj} = \frac{1}{2} (\overline{B}_{xy} + \overline{B}_{yx})$$
 (12)

in place of $(\overline{B}_{xy}, \overline{B}_{yx})$ which are directly extracted from a data table and may not have identical values. Clearly, a data table which already satisfies Eq. (11) is not affected by this adjustement.

2.4 Natural Orbits and Effective Stiffness/Damping Constants

2.4.1 General Case:
$$Z_{xy} Z_{yx} \neq 0$$

Suppose the perturbation motion is simple harmonic so that

$$(\delta_{x}^{\prime}, \delta_{y}^{\prime}) = \operatorname{Re} \{(\tilde{\delta}_{x}^{\prime}, \tilde{\delta}_{y}^{\prime}) e^{i\nu t}\}$$
 (13)

The corresponding perturbation reactions are

$$(\mathbf{F}_{\mathbf{x}}^{\prime}, \mathbf{F}_{\mathbf{v}}^{\prime}) = \operatorname{Re} \{(\mathbf{F}_{\mathbf{x}}^{\prime}, \mathbf{F}_{\mathbf{v}}^{\prime}) e^{i\nu t}\}$$
 (14)

To make use of matrix notation, write

$$\underline{u} = \left\{ \begin{array}{c} \tilde{\delta}_{x} \\ \tilde{\delta}_{y} \end{array} \right\} : \underline{P} \left\{ \begin{array}{c} \tilde{F}_{x} \\ \tilde{F}_{y} \end{array} \right\}$$

$$(15)$$

Given the stiffness and damping matrices, the perturbation reaction can be computed as

$$\underline{\mathbf{P}} = -\mathbf{K} + \mathbf{i} \mathbf{v} \cdot \mathbf{B} \cdot \underline{\mathbf{w}} \tag{16}$$

One can seek the condition of diagonalization of the above equation; i.e. provided the components of \underline{w} assume a special relative amplitude and phase relationship, one can write

$$\underline{P}_{o} = -(k_{o} + i\nu b_{o}) \underline{w}_{o} \tag{17}$$

Subscript "o" designates the special relationship which describes a natural orbit. The physical interpretation of Eq. (17) is that the bearing behaves like a spring-dashpot restraint if the perturbed motion is a natural orbit, the spring and dashpot constants being respectively k_0 and b_0 .

Given [K] and [B] of a bearing, k_0 and b_0 can be found at any frequency ν as roots of a complex eigenvalue problem. Accordingly, \underline{w}_0 can be determined as the corresponding eigenvector. Upon first identifying \underline{P} and \underline{w} in Eq. (16) with \underline{P}_0 and \underline{w}_0 respectively, and then eliminating \underline{P}_0 between Eqs. (16) and (17), one obtains the homogeneous equation

$$\begin{bmatrix} K + iv B - (k_o + ivb_o) & I \\ W = 0 \end{bmatrix} \underline{w}_o = \underline{0}$$
 (18)

[I] is the identity matrix with unity as its diagonal elements and null elements elsewhere. Disregarding the possibility that \underline{w}_0 may be a null vector, the characteristic determinant should vanish. k_0 and b_0 are roots of the characteristic determinant and are found as sets which are consistent

with the null condition of both real and imaginary parts of the characteristic determinant. The characteristic determinant is

$$\{Z_{xx} - (k_o + ivb_o)\}\{Z_{yy} - (k_o + ivb_o)\} - Z_{xy}Z_{yx} = 0$$
 (19)

where, $Z_{xx} = K_{xx} + ivB_{xx}$, etc.; and an explicit solution can be found as

$$k_o + ivb_o = \frac{1}{2} (z_{xx} + z_{yy}) \div z_o$$
 (20)

where

$$Z_{o} = \frac{1}{2} \int_{J} (Z_{xx} - Z_{yy})^{2} + 4 Z_{xy} Z_{yx}$$
 (21)

By convention, Z_0 has a non-negative real part. The alternate signs in Eq. (20) yield two independent sets of $(k_0, \nu b_0)$. One may observe that the imaginary part of the right-hand side of Eq. (20) may not vanish with ν . Thus, it is more appropriate to keep ν and b together as a product. It is sometimes useful to define $\beta_0 = (\nu b_0)/(2k_0)$ as the critical damping ratio of the natural orbit, which is related to the amplitude decrement factor of an unrestrained natural motion per cycle.

Corresponding to the two sets of $(k_0^{}, \nu b_0^{})$, there are two eigenvectors, which are defined by

$$\{\frac{1}{2}(Z_{xx} - Z_{yy}) \pm Z_{o}\} \tilde{\delta}_{xo} + Z_{xy}\tilde{\delta}_{yo} = 0$$
 (22a)

or, alternately,

$$z_{yx}\tilde{\delta}_{xo} - \left[\frac{1}{2}(z_{xx} - z_{yy}) \pm z_{o}\right]\tilde{\delta}_{yo} = 0$$
 (22b)

$$\left|\tilde{\delta}_{xo}\right| \div \left|\tilde{\delta}_{yo}\right| = 1$$
 (23)

$$Arg\{\delta_{xo}\} + Arg\{\delta_{yo}\} = 0$$
 (24)

which sets the eigenvector to be

$$\underline{\underline{w}}_{o} = \begin{cases} u_{o} e^{i\gamma_{o}} \\ (1-u_{o}) e^{-i\gamma_{o}} \end{cases}$$
 (25)

 u_0 (which has a non-negative value bounded by unity) and γ_0 (which is one-half the relative phase between the two degrees of freedom) remain to be found. In order to avoid writing lengthy algebraic expressions, in place of Eq. (22a) or Eq. (22b), let the governing equation for the eigenvector be

$$G_{x}\tilde{\delta}_{xo} + G_{y}\tilde{\delta}_{yo} = 0$$
 (26)

where

$$G_{x} = U_{x} + iV_{x}; \quad G_{y} = U_{y} + iV_{y}$$
(27)

are complex quantities. Substitution of Eq. (25) into Eq. (26) then sets real and imaginary parts separately to zero, and one finds

$$\{v_{x_0} + v_y(1-v_o)\} \cos \gamma_o - \{v_{x_0} - v_y(1-v_o)\} \sin \gamma_o = 0$$

$$\{V_{x,0} + V_{y}(1-u_{0})\} \cos \gamma_{0} + \{U_{x,0} - U_{y}(1-u_{0})\} \sin \gamma_{0} = 0$$

They yield, after some straightforward algebraic manipulations

$$u_o = \frac{|G_y|}{|G_x| + |G_y|}$$
; $1 - u_o = \frac{|G_x|}{|G_x| + |G_y|}$ (28)

$$\gamma_0 = \tan^{-1} \frac{(v_x | G_y | + v_y | G_x |)}{(v_x | G_y | - v_y | G_x |)}$$
 (29)

The principal value of Eq. (29) would be used. The eigenvector as defined by Eq. (25, 28, 29) is thus far presented in terms of a Cartesian coordinate system. It describes an elliptical orbit with the sense of whirl governed by γ_0 . The sense of whirl is positive; i.e., same as the sense of shaft rotation if γ_0 is in either the first or the third quadrant; and the sense of whirl is negative or opposite to that of shaft rotation if γ_0 is in either the second or the fourth quadrant. In other words, the sense of whirl has the same sign as

$$tan\gamma_o = \frac{v_x |G_y| + v_y |G_x|}{v_x |G_y| - v_y |G_x|}$$

If either Z_{xy} or Z_{yx} or both should vanish, the formulations presented above are not workable because both G_x and G_y would also vanish, rendering Eqs. (28) and (29) indeterminate. These are special cases and require separate attention.

2.4.2 No Cross-coupling:
$$Z_{xy} = 0$$
, $Z_{yx} = 0$

The homogeneous system of Eq. (18) becomes

$$\left\{
\begin{array}{cccc}
Z_{xx} - (k_o + ivb_o) & 0 & \\
0 & Z_{yy} - (k_o + ivb_o) & \tilde{\delta}_{xo}
\end{array}
\right\} = 0$$

Since the two degrees of freedom are totally uncoupled, the desired eigenvalue/vector sets are simply

$$k_0 + ivb_0 = Z_{xx}; \quad \tilde{\delta}_{x0} = 1, \quad \tilde{\delta}_{y0} = 0$$
 (31a)

and

$$k_{o} + i\nu b_{o} = Z_{yy}; \quad \tilde{\delta}_{xo} = 0, \quad \tilde{\delta}_{yo} = 1$$
 (31b)

2.4.3 Pseudo Uncoupled Case

One of $\mathbf{Z}_{\mathbf{x}\mathbf{y}}$ and $\mathbf{Z}_{\mathbf{y}\mathbf{x}}$ is zero, the other one is not. Write

$$z_{ik} = 0; z_{ki} \neq 0$$
 (32)

The indices j or k may denote either x or y. They are always distinct. (Repeated indices designate a diagonal term; the implicit summation convention of indicial contraction is <u>not</u> used here.)

Eq. (18) is reduced to

$$\{Z_{jj} - (k_0 + ivb_0)\} \tilde{\delta}_{j0} = 0$$
 (33)

$$z_{kj} \tilde{\delta}_{jo} + \{z_{kk} - (k_o + ivb_o)\} \tilde{\delta}_{ko} = 0$$
 (34)

Eq. (33) can be satisfied by two conditions. The first one is

$$k_{o} + i\nu b_{o} = Z_{ij}$$
(35)

Consequently, substituting into Eq. (34)

$$Z_{kj} \tilde{\delta}_{jo} + (Z_{kk} - Z_{jj}) \tilde{\delta}_{ko} = 0$$
 (36a)

Following steps previously used to describe Eqs. (28) and (29), from Eq. (26), one finds

$$|\tilde{\delta}_{jo}| = \frac{|z_{kk}^{-2}j_{j}|}{|z_{kj}^{-1}|+|z_{kk}^{-2}j_{j}^{-1}|}; \quad |\tilde{\delta}_{ko}| = \frac{|z_{kj}^{-1}|}{|z_{kj}^{-1}|+|z_{kk}^{-2}j_{j}^{-1}|}$$
(37a)

$$Arg\{\tilde{\delta}_{jo}\} = -Arg\{\tilde{\delta}_{ko}\}$$

$$= tan^{-1} \left[\frac{K_{kj} |Z_{kk} - Z_{jj}| + (K_{kk} - K_{jj}) |Z_{kj}|}{\nu(B_{kj} |Z_{kk} - Z_{jj}| - (B_{kk} - B_{jj}) |Z_{kj}|)} \right]$$
(38a)

If Z_{kk} and Z_{jj} should happen to be equal, Eq. (38a) becomes indeterminate, at the same time δ_{jo} vanishes while $|\delta_{kg}|$ becomes unity. Actually, with δ_{jo} being zero, the argument of δ_{ko} has no physical significance and can be set to zero as an accepted convention.

The second condition which satisfies Eq. (33) is

$$\tilde{\delta}_{jo} = 0 \tag{37b}$$

Consequently, one can set

$$\bar{\delta}_{ko} = 1 \tag{38b}$$

Then, upon substitution into Eq. (34), one obtains

$$k_0 + ivb_0 = Z_{kk}$$
 (35b)

Note that the second eigenvalue/vector set is indistinguishable from the first set under the conditions of $(Z_{jk} = 0, Z_{kj} \neq 0, Z_{jj} = Z_{kk})$ which represent a truly degenerate state since only one eigenvalue/vector set can be found for a two-degrees-of-freedom system.

The parameters $(\tilde{\delta}_{xo}, \tilde{\delta}_{yo})$ are Cartesian components of the natural orbit. Since every simple harmonic orbit takes on the shape of an ellipse, the natural orbit can be represented by:

- ratio of minor/major radii of the orbit, and
- inclination of the major axis from the x-axis

The relationships between the Cartesian components and the geometric parameters are derived in another part of the Rotor-Bearing Dynamics Technology Design Guide [9].

SECTION III

RETRIEVAL SYSTEM

3.1 Contents of the Bearing Data Bank

There are presently thirty-one (31) data tables in the collection of the data bank. Each data table represents one combination of configurations, geometrical parameters, and preload setting (if applicable). A summary of the contents of the data bank is contained in Table 1.

For bearings made up of partial arc pads, there is a setup parameter called preload, which measures the distance between the arc center and the bearing center as a fraction of the nominal clearance (pad clearance). As shown in Fig. 3, the bearing clearance C_b is smaller than the original arc clearance by the amount of preload mC. In this particular illustration, the preload is centrally directed, i.e., the line of centers bisect the pad arc. All data tables thus far collected have centrally preloaded pads.

For partial arc configurations, rotational symmetry is disrupted. The direction of the load vector is to be defined with reference to the arc geometry. A lobed bearing is made up of equally spaced arcs. The direction of the load vector is specified relative to the loaded arc. A two-lobe bearing is usually parted in a horizontal plane. Thus the gravity load is commonly directed along the bi-sector of the bottom arc. Such an arrangement is said to have a load "on pad." With an odd number of lobes, the usual practice is to place one more arc on the bottom than the top. Thus the load is directed between pads for the three-lobe bearing.

The tilting shoe bearing is described similarly as the lobed bearing. The preload parameter deals with the radial displacement of the pivot point. The angular location of the pivot is equivalent to the preload direction relative to the arc. It is known what the optimum pivot location is about 55% from the leading edge. However, if the pads should be erroneously in-

TABLE 1

CONTENTS OF DATA BANK

Bearing Type	Arc Angle	L/D	Notes	Preload	Data Table Number
Plain	360	0.50			PJ-05-1
2 Lobe	160	0.50	(1),(5)	0.00	ML2-05-1
				0.25	ML2-05-2
				0.50	ML2-05-3
3 Lobe	100	0.50	(2),(5)	0.00	MI.3-05-1
				0.25	ML3-05-2
				0.50	ML3-05-3
4 Shoe	80	0.25	(2),(3),	0.00	TP4-02-1
Tilting Pad			(4)	0.20	TP4-02-2
				0.30	TP4-02-3
				0.50	TP4-02-4
		0.50		0.00	TP4-05-1
				0.20	TP4-05-2
·				0.30	TP4-05-3
				0.50	TP4-05-4
		1.00		0.00	TP4-10-1
				0.20	TP4-10-2
				0.30	TP4-10-3
				0.50	TP4-10-4

LEGEND:

- (1) Load on pad
- (2) Load between pads
- (3) Centrally pivoted
- (4) Synchronous frequency only, and shoe inertia is neglected
- (5) Centrally preloaded

TABLE 1 - CONTENTS OF DATA BANK (continued)

Bearing Type	Arc Angle	L/D	Notes	Preload	Data Table Number
5 Shoe Tilting Pad	55	0.25	(1),(3), (4)	0.00 0.20 0.30	TP5-02-1 TP5-02-2 TP5-02-3
				0.50	TP5-02-4
		0.50		0.00	TP5-05-1
				0.20	TP5-05-2
				0.30	TP5-05-3
				0.50	TP5-05-4
		1.00		0.00	TP5-10-1
				0.20	TP5-10-2
				0.30	TP5-10-3
				0.50	TP5-10-4

LEGEND:

- (1) Load on pad
- (2) Load between pads
- (3) Centrally pivoted
- (4) Synchronous frequency only, and shoe inertia is neglected
- (5) Centrally preloaded

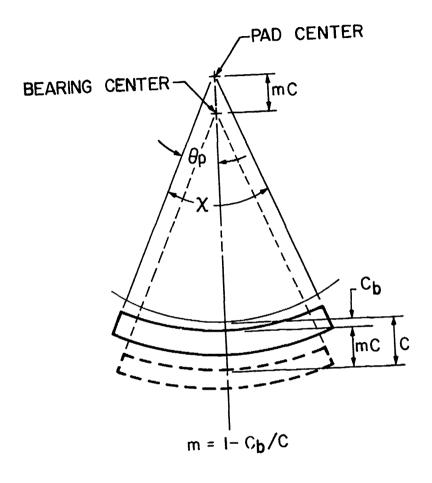


Figure 3 Centrally Preloaded Partial Arc ($\theta_p = \chi/2$)

stalled backwards, the load capacity of the bearing would be drastically smaller. The centrally pivoted arrangement is error safe since it acts identically if reversed. The tilting shoe bearings in the present data bank are all centrally pivoted.

An ideal pivot is incapable of exerting moment to the bearing pad. This condition adds two additional parameters for a complete definition of the dynamic data; namely,

$$\overline{J} = \frac{J_p N_s}{\mu(R/C)^2 (LD) R}$$
(39)

$$\overline{v} = f/N_{s} \tag{40}$$

which relate to the pitch moment of inertia of a bearing pad and the ratio of the vibrational frequency to the rotation frequency, respectively. In the present data, $(\overline{J} = 0 \text{ and } \overline{v} = 1)$ are imposed. The inertialess assumption is usually adequate since the inertia of each pad should be quite small in comparison with the rotor inertia. An exception is the unloaded shoe, which would become vulnerable to the pathological condition of shoe For this reason, tilting shoe bearings should always be pre-The data tables for the unloaded tilting shoe bearings are included mainly for reference. The nature of frequency dependance of the perturbation coefficients of tilting shoe bearings in effect reflects the phase relationships between the rotor motion and the pitching motions of the shoes. tunately, the total resultant of all shoes does not significantly vary with Therefore, the synchronous data (v = 1) can be used for nonsynchronous rotor dynamic studies with fair accuracy. If the precise data with allowance for shoe inertias and frequency dependance should be desired, the dynamic assembly procedure should be followed to synthesize the bearing characteristics from single pad characteristics [3].

All data tables in the present collection concern journal bearings which influence dynamics of the rotor system by reacting to lineal motions of the rotor. Some bearing support systems can react to angular motions of the rotor. A long journal bearing and a large thrust bearing runner are two such examples. In reacting to an angular motion of the rotor (in the radial plane), the bearing can provide a restraining moment. The required data parameters for such bearings are angular stiffness and angular damping coefficients which react to angular displacements and angular velocities, respectively. These coefficients also may induce anisotropic and crosscoupling features. The storage and retrieval methodologies described in this document can be readily adapted to handle angular bearings.

3.2 Interfacing Bearing Data with Rotor Dynamics Software

Each bearing data table is necessarily restricted to the following two conditions:

- The bearing data is presented in a coordinate system which is peculiar to static load vector acting on the particular bearing.
- 2. The pedestal of the bearing is assumed to be rigid.

The bearing data retrieval system incorporates an interfacing procedure which removes these restrictions.

The static load vectors of all bearings in a rotor system may not share the same direction. The following circumstances require that individual load directions be assigned to each bearing:

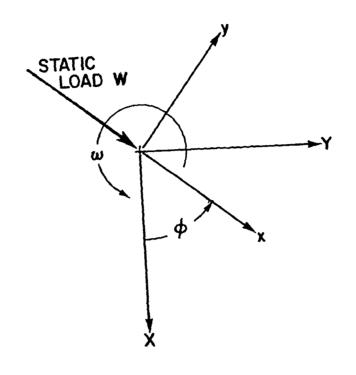
 For an overhung rotor, if its center of gravity is outboard, the load vectors at the two bearings would have opposite directions.

- Gyroscopic loads which occur during rapid maneuvers of a military aircraft, usually are oppositely directed at the two main support bearings.
- 3. Laterally coupled rotors (e.g., in a transmission box) exert force on each other and tend to cause load vectors at various bearings to assume distinct directions.
- 4. Misalignment loads (for rotors supported by three or more bearings) generally may not be co-planar with the gravity load.

For these reasons, it is necessary to allow a deviation of the static bearing load vector from an axis of the global coordinates of the rotor: Let (X, Y) and (x, y), respectively, be global (rotor) and local (bearing) coordinate axes in the Cartesian representation, with the x-axis directed along the static load. Let ϕ be the angle measured from X to x as illustrated in Fig. 4. The transformation of a vector represented in the bearing coordinates to that represented in the rotor coordinates is given by the relation

$$\begin{bmatrix} X & \vdots & \cos \phi - \sin \phi & \int x \\ Y & \vdots & \sin \phi & \cos \phi - y \end{bmatrix}$$
(41)

Transformation of the stiffness and damping matrices is achieved by the following formula



XY ROTOR COORDINATES
XY BEARING COORDINATES

Figure 4 Rotor and Bearing Coordinate Systems

Pedestal compliance and/or damping can contribute significantly to the rotor behavior. Pedestal compliance can be modelled in terms of lumped parameters.

$$Z_{p} = K_{p} + [M]_{p} \frac{d^{2}}{dt^{2}} + [B]_{p} \frac{d}{dt}$$
 (43)

This is regarded as an operator for the displacement of the pedestal. The subscript "p" labels the pedestal effects. For a simple harmonic motion with the time dependence factor exp{ivt}, Eq. (43) is reduced to

$$Z_{p} = K_{p} - v^{2} M_{p} + iv \left[B\right]_{p}$$
 (44)

Since the bearing reaction is directly transmitted to the pedestal

$$\underline{P} = -\begin{bmatrix} Z \\ b \end{bmatrix} \underline{w}_b - \underline{w}_p$$
 (45a)

$$= -Z \underset{-}{\overset{w}{=}} p$$
 (45b)

 $\begin{bmatrix} 2 \end{bmatrix}_b$ is the impedance matrix of the (rigidly mounted) bearing. \underline{w}_b and \underline{w}_p are displacement vectors of the bearing and of the pedestal in the matrix notation, respectively. Eliminating \underline{P} between Eqs. (45a) and (45b), one finds

$$-\left[2_{b} \underline{w}_{b} + 2_{p} + 2_{b} \underline{w}_{p} = 0\right]$$

Solving for $\frac{w}{p}$ then substituting back into Eq. (45b), one obtains

$$\underline{P} = -\left[z\right]_{p} \left[\left[z\right]_{p} + \left[z\right]_{b}\right]^{-1} \left[z\right]_{b} \underline{w}_{b}$$
(46)

which in essence defines the effective support impedance as

$$\begin{bmatrix} z \end{bmatrix}_{\text{effective}} = \begin{bmatrix} z \end{bmatrix}_{p} \begin{bmatrix} z \end{bmatrix}_{p} + \begin{bmatrix} z \end{bmatrix}_{b}^{-1} \begin{bmatrix} z \end{bmatrix}_{b}$$
(47)

The retrieval software provides the user with an option to allow for pedestal compliance. Upon selecting this option, the user can furnish data consisting of up to ten numbers which define the elements of the three matrices $\left[K\right]_p$, $\left[B\right]_p$, and $\left[M\right]_p$. Upon extracting $\left[Z\right]_b$ from the data bank, Eq. (47) is executed by the interfacing software.

Aside from performing required coordinate transformations and incorporating pedestal compliance effects, the bearing data retrieval system also organizes the results in a tabular form consistent with the input format of the rotor dynamics software [9].

3.3 The Retrieval Procedure

The Fluid Film Bearing Data Bank is designed to furnish numerical information of bearing support characteristics needed in the performance of a variety of rotordynamic studies. The software for data retrieval requires the user to furnish the following input:

1. Location of Data Table -- The user has the choice of selecting a data table from the index of the available data bank or submitting the data bank to the file management system of the computer center in advance of the retrieval run. In either case, the data table is a result of preparation for full range interpolation according to the method to be described in Section IV. 2. Design Parameters of the Bearing.

o lubricant viscosity (centipoise)

o bearing diameter (in)

o bearing length (in)

o clearance (in)

3. Operating Parameters of the Bearing.

o load magnitude (lbs)

o load direction defined by the angle measured from the vertical axis in the sense of shaft rotation (deg)

o shaft rotational speed (rpm)

o vibration frequency as may be needed in certain bearing support systems (Hz)

4. Pedestal Compliance Characteristics -- The user can elect to furnish lumped parameter data of the pedestal which define stiffness, damping, and mass matrices.

o pedestal weight (1bs)

o pedestal c.g. offsets in

three directions (in)

o pedestal angular weight moment of inertia in two

planes (lb-in²)

o pedestal stiffness con-

stants in two planes (lbs/in or in-lb/rad)

o pedestal damping constants in

two planes (lbs-sec/in or in-lb-sec/rad)

The output of the retrieval software includes in tabular forms

 Data sets compatible with the input format of the rotordynamics software [9],

- 2. Tabulation of the static bearing characteristics,
- 3. Tabulation of the dynamic bearing coefficients, and
- 4. Effective stiffness/damping constants and natural orbit parameters as described in Section 2.4.

Additional information pertaining to the use of the retrieval software is given in Appendix A.

3.4 Description of the Retrieval Table

The retrieval table of each bearing is furnished as a sequential FORTRAN file. Its contents include the following:

- Alphanumeric Index This contains the file designation, the file size or number of data points (including nearand far-field limits), L/D of the bearing, and the load scaling factor which is presently defaulted as 1.125.
- Array of Working Parameters This is stored as groups of seven floating point numbers, each occupying a field of eleven spaces.
- 3. Retrieval Data of Each of Thirteen Data Variables The thirteen variables appear in the following order: eccentricity ratio, attitude angle, friction, required flow, lost flow, \overline{K}_{xx} , \overline{B}_{xx} , \overline{K}_{xy} , \overline{B}_{xy} , \overline{K}_{yx} , \overline{B}_{yx} , \overline{K}_{yy} , and \overline{B}_{yy} .

The retrieval data of each variable consists of numbers which are also stored in succession in groups of seven floating point numbers each occupying a field of eleven spaces. They are stored without interruption for each variable and include: near-field exponent s_1 , far-field exponent s_2 , near-field constant s_3 , far-field constant s_4 , then for each data interval a set of spline coefficients for each of the data intervals corresponding to the array of working parameters (the comparison function and its three derivatives with respect to the working

parameter), and the comparison function for the far-field limit which should always be unity.

Table 2 shows the contents of a typical retrieval file. Table 3 is a printout of the contents with appropriate headings. The load is computed from the working parameter (TAU) according to Eq. (62) described in the next section.

A complete listing of all retrieval files of the present data bank is given in Appendix D.

TABLE 2

```
RETRIEVAL FILE NO
                   ML3-05-1
                            TYPICAL RETRIEVAL FILE
      FILE SIZE =
                    17
      LZD
                      0.5000
      ALFA
                      1.1250
0.00000-01 3.57680-02 5.84560-02 8.01380-02 1.27180-01 2 07210-01 2.86020-01
4.4623D-01 5.7427D-01 6.5943D-01 7.6051D-01 8.1662D-01 8.4578D-01 8.7580D-01
9.0640D-01 9.3739D-01 1.8000D 00
1.00000 00 0.00000-01 1.00010 00 1.00000 00 1.00000 00-1 65020 00 0.00000-01
1.9803D 02 9.4249D-01-1.5236D 00 7.0830D 00-1.3293D 02 9.0948D-01-1.3971D 80
4.0671D 00-6.6448D 01 8.8003D-01-1.3245D 00 2.6264D 00 1 2060D 01 8 2085D-01
-1.1876D 00 3.1937D 00 1.3868D-02 7.3602D-01-9.3197D-01 3.1948D 00-1 7265D 00
6.7236D-01-6.8557D-01 3.0587D 00-3.4465D 00 5.9942D-01-2.3977D-01 2.5066D 00
6.8959D-01 5.8950D-01 8.6833D-02 2.5949D 00 5.2071D 00 6.0385D-01 3.2671D-01
3.0383D 00 1.0659D 01 6.5722D-01 6.8826D-01 4.1157D 00 2 4396D 01 7 0303D-01
9.5757D-01 5.4845D 00 2.6015D 01 7.3340D-01 1.1286D 00 6.2432D 00 3.4971D 01
7.7026D-01 1.3318D 00 7.2931D 00 6.2869D 01 8.1471D-01 1 5843D 00 9.2165D 00
3.4883D 01 8.6841D-01 1.8867D 00 1.0298D 01-1.6446D 02 1.0000D 00
0.0000D-01-5.0000D-01 9.0000D 01 1.0007D 02 1.0000D 00-2.2698D 00 0.0000D-01
5.0412D 02 9.2266D-01-1.9473D 00 1.8031D 01-4.9054D 02 8.8216D-01-1.6645D 00
6.9020D 00 4.2378D 01 8.4777D-01-1.5049D 00 7.8208D 00-4.8412D 01 7 8479D-01
-1.1905D 00 5.5436D 00-2.1033D 01 7.0546D-01-8.1422D-01 3 8601D 00-1 5182D 01
6.5205D-01-5.5717D-01 2.6638D 00~6.8634D 00 5.9227D-01-2.1849D-01 1 5642D 00
9.3216D-01 5.7744D-01-1.0573D-02 1.6835D 00 4.1808D 00 5 8307D-01
                                                                   1 47960-01
2.0396D 00 1.5715D 01 6 1115D-01 4.3440D-01 3.6280D 00 4 4005D 01 6 4253D-01
7.0720D-01 6.09690 00 5.9916D 01 6.6600D-01 9.1050D-01 7.8443D 00 1.3540D 02
6.9748D-01 1.2070D 00 1.1909D 01 1.6682D 02 7.4078D-01 1.6494D 00 1.7013D 01
5.4840D 02 8.0278D-01 2.4400D 00 3.4008C 01-5.4315D 02 1.0000D 00
0.0000D-G1 5.0000D-01 1.6463D 01 1.7306D 01 1.0000D 00 1 2603D 00 0 0000D-01
-3.1207D 02 1.04270 00 1.0607D 00-1.1162D C1 3.5253D 02 1 0646D 00
                                                                   8 98190-01
-3.1636D 00-1.9538D 01 1.0933D 00 8.2500D-01-3.5872D 00 2.1639D 01 1 1185D 00
6.8020D-01-2.5694D 00 7.0205D 00 1.1653D 00 4.9704D-01-2 0075D 00 5.1224D 00
1.1987D 00 3.5475D-01-1.6038D 00-9.9900D-01 1 2342D 00 8 4981D-02-1 7639D 00
-9.9638D-01 1.2303D 00-1.4904D-01-1.8914D 00-3.7706D 00 1 2104D 00-3 2379D-01
-2.2126D 00-6.9700D 00 1.1651D 00-5.8304D-01-2.9171D 00 5.6679D 00 1.1280D 00
-7.3778D-01-2.5991D 00 4.7046D 01 1.1056D 00-7.9358D-01-1 2270D 00-9 1263D 01
1.0808D 00-8.7155D-01-3.9670D 00 4.5428D 02 1.0544D 00-7 8032D-01 9.9308D 00
-2.4352D 02 1.0338D 00-5.8950D-01 2.3839D 00-3.8074D 01 1 0000D 00
0.0000D-01 0.0000D-01 4.7133D 00 4.0711D 00 1.0000D 00 7 7286D-02 0.0000D-01
-2.9511D 01 1.0025D 00 5.8409D-02-1.0555D 00 1.0540D 01 1 0036D 00 3.7173D-02
-8.1640D-01 2.6434D 01 1.0043D 00 2.5686D-02-2.4323D-01-2 2245D 00 1.0052D 00
1.17830-02-3.47870-01 2.51140 00 1.00520 00-8.01560-03-1.46870-01 2 75070 00
1.00440 00-1.10480-02 6.98980-02-2 53570-01 1 00330 00-3 10420-03 2 92740-02
4.73360-01 1.00330 00 4.52440-03 8 38840-02-9.16810-01 1 00390 00 8.85450-03
1.1804D-02 3 9665D-01 1.0050D 00 1.2023D-02 5.0896D-02 3 5435D-01 1 00570 00
1.54350-02 7.17660-02-2.0676D 01 1.0061D 00 8.7061D-03-5 3224D-01 1 7294D 01
1.0062D 00 5.2080D-04-1.3025D-02-3.9162D 01 1 0060D 00-1 8204D-02-1 21110 00
2.4203D 00 1.0049C 00-5.4574D-02-1.1361D 00 1 8145D 01 1 0000D 00
1.00000 00 0.00000-01 1.49020 00 1.61700 00 1 00000 00-1 54800 00 0 00000-01
1.2753D 02 9.4560D-01-1.4665D 00 4.5616D 00-1.1867D 02 9.1327D-01-1 3935D 00
1.8691D 00 7.3618D 01 8.8362D-01-1.3357D 00 3 4653D 00-1 6203D 01 8.2435D-01
-1.1906D 00 2.7032D 00 7.5566D 00 7.3836D-01-9 5004D-01 3 3080D 00-4.6527D 00
6.7339D-01-7.0381D-01 2.9413D 00-2.4167D 00 5.9672D-01-2 6360D-01 2 5541D 00
7.0220D-01 5 8415D-01 6.9196D-02 2.6441D 00 4 6007D 00 6 00!1D-01 3 1106D-01
3.0359D 00 1.1362D 01 6.4901D-01 6 7596D-01 4 1843D 00 3 1211D 01 6 9444D-01
```

Table 2 - Typical Retrieval File (continued)

```
9.5984D-01 5.9354D 00-7.6498D 00 7.2493D-01 1.1297P C 5 7123D 00 5.4438D 01
7.6166D-01 1.3257D 00 7.3467D 00 1.2255D 01 8.057° 31 1 5562D 00 7.7216D 00
2.6260D 02 8.5896D-01 1.9216D 00 1.5860D 01-2.533f
                                                     2 1 00000 00
1 00000 00 2.50000 00-5.51020-01 9.01730-01-1.006 ^{\circ}
                                                       2.43230 00 0.00000-01
7.1487D 02-9.0755D-01 2.8896D 00 2.5569D 01-9.14;
                                                     UZ-8.3719D-01 3.2344D 00
4.82830 00 5.04370 02-7.6507D-01 3.4576D 00 1.576.5 01-2.2806D 02-5.8893D-01
3.9469D 00 5.0357D 00-2.7454D 00-2.5717D-01 4.7411D 00 4.8160D 00-2.0724D 02
8.3000D-02 4.0770D 00-1.1517D 01 5.9633D 01 6.. 25D-01 2.9972D 00-1.9630D 00
-1.1046D 02 9.5828D-01 1.8404D 00-1.6106D 01 1.0971D 02 1 0679D 00 8.6662D-01
-6.7639D 00-6.0643D 00 1.1199D 00 1.5194D-01-7.3769D 00-4.4883D-01 1.1168D 80
-2.6268D-01-7.4020D 00 4.8558D 02 1.1080D 00-2.7207D-01 6 7576D 00-6.3813D 02
1.1000D 00-3.5675D-01-1.2399D 01 3.6049D 02 1 0850D 00-5 6738D-01-1.3678D 00
-3.5481D 02 1.0650D 00-7.8015D-01-1.2363D 01 1.9747D 02 1 0000D 00
0.0000D-01 1.5000D 00 2.0058D 00 1.4450D 01 1.0000D 00-1 8609D 00 0.0000D-01
-9.6880D 02 9.2605D-01-2.4806D 00-3.4652D 01 2.2807D 03 8 6529D-01-2.6798D 00
1.7092D 01-1.9040D 02 8.1088D-01-2.3540D 00 1.2964D 01 2 1440D 01 7.1486D-01
-1.7204D 00 1.3973D 01-7.1428D 01 6.1582D-01-8.3091D-01 8 2564D 00-7.3256D 01
5.7000D-01-4.0772D-01 2.4831D 00-1.1667D 01 5.2855D-01-1 5964D-01 6.1388D-01
1.0713D 01 5.1689D-01 6.7780D-03 1.9855D 90-1.7847D 01 5.2283D-01 1.1115D-01
4.6571D-01 3.7505D 01 5.4290D-01 3.4982D-01 4.2567D 00-4 0585D 00 5.6911D-01
5.8228D-01 4.0290D 00 5.2698D 02 5.8998D-01 9.2381D-01 1 9396D 01 1.2142D 02
6,2700D-01 1.5608D 00 2.3041D 01 3.0426D 02 6.8700D-01 2.4083D 00 3.2351D 01
-4.3686D 02 7.7500D-01 3.2011D 00 1.8813D 01-3.0047D 02 1 0000D 00
0 0000D-01 2.0000D 00 1.0028D 00 2.4336D 00 1.0000D 00 2 5245D-01 0.0000D-01
6.9107D 02 1.0143D 00 6.9450D-01 2.4718D 01-3.6921D 02 1 0357D 00 1.1603D 00
1.63410 01-1.16740 02 1.06450 00 1 48720 00 1.38100 01-1 40590 02 1 14730 00
1.9813D 00 7.1964D 00-1.6511D 02 1.3148D 00 2.0284D 00-6.0173D 00-7.0334D 00
1.4554D 00 1.5324D 00-6.5716D 00-2.4457D 01 1 5998D 00 1.6564D-01-1.0490D 01
3 9955D 01 1.5490D 00-8.4997D-01-5.3741D 00 1.6620D 00 1 4573D 00-1 3016D 00
-5.2326D 00 5.4599D 01 1.3084D 00-1.5516D 00 2.8626D-01-3 7039D 01 1.2207D 00
-1.5938D 00-1.7920D 00 8.1851D 01 1 1738D 00-1.6113D 00 5.9479D-01 2.8899D 02
1.1270D 00-1.4632D 00 9.2704D 00-1.1846D 02 1.0860D 00-1.2350D 00 5.6455D 00
1 1328D 02 1.0510D 00~1.0057D 00 9.1561D 00-1 4624D 02 1 0000D 00
1 0000D 00 1.5000D 00 5.0960D 00 4 3765D 00 1.0000D 00 5 0622D-01 0 0000D-01
2 4829D 02 1.0200D 00 6.6504D-01 8.8804D 00-6.0311D 02 1 0362D 00 7 1129D-01
-4.80300 00 2.98320 02 1.05100 00 6.77270-01 1 66520 00-4 62670 01 1 08390 00
7.0441D-01-5.1129D-01-2.1501D 01 1.1368D 00 5.9464D-01-2.2320D 00 3.2866D 00
1.1770D 00 4.2894D-01·1.9730D 00-5.9810D 00 1.2163D 00 3 6082D-02-2.9312D 00
6 0249D 00 1.1990D 00-2.8985D-01-2.1399D 00 2.2492D 01 1 1688D 00-3.9222D-01
-2.4440D-01 5.4648D-01 1.1280D 00-4 1413D-01-1.8916D-01-2 9390D 01 1.1036D 00
-4.71010-01-1.8382D 00 1.0071D 02 1.0895D 00-4.8179D-01 1 0986D 00-3.3966D 02
1 0740D 00-6.0187D-01-9.0980D 00 5.6050D 02 1.0540D 00-6 1785D-01 8 0533D 00
-3 4674D 02 1.0370D 00-5 3478D-01-2.6920D 00 4 2997D 01 1 0000D 00
0 00000-01 2 50000 00-2 27990 00 2 24600-01-1 00000 00 8 05130-01 0.0000D-01
-3 2357D 02-9 7367D-01 5 9815D-01-1.1573D 01-1 7063D 02-9 6341D-01 2.9166D-01
-1 5445D 01 1.4284D 03-9.5829D-01 2 9254D-01 1.5526D 01-6 0409D 02-9.37870-01
3 54510-01-1 28920 01 2 33080 02-9 30830-01 6 92310-02 5 76220 00-1.04440 02
-9 1600D-01 1 9900D-01-2.4689D 00 9 5282D 01-8.5050D-01 1 0263D 00 1.2796D 01
-2 4658D 01-6.2283D-01 2.4626D 00 9 6391D 00 2.2912D 02-3 5458D-01 4.1143D 00
2.9151D 01-8.5809D 01 1.9544D-01 6.6225D 00 2.0477D 01 1 0861D 02 6.0246D-01
7.9424D 00 2.6572D 01-7.5738D 03 8 1406D-01 5.4972D 00-1 9428D 02 5 2019D 03
9.1500D-01 2.0089D 00-3.8119D 01 2.8757D 02 9.6000D-01 9 7712D-01-2 9319D 01
7 6561D 02 9.8000D-01 4.3616D-01-5 5929D 00 8 9329D 01 1 0000D 00
1.0000D 00 1.5000D 00 5.0960D 00 4 3765D 00 1.0000D 00 5 0622D-01 0 0000D-01
                                                     (Sheet 2 of 3)
```

Table 2 - Typical Retrieval File (continued)

```
2.48280 02 1.02000 00 6.65040-01 8.88040 00-6.03110 02 1 03620 00 7 11290-01
-4.8030D 00 2.9832D 02 1.0510D 00 6 77270-01 1 6652D 00-4 6267D 01 1 0839D 00
 7.0441D-01-5.1129D-01-2.1501D 01 1 1368D 00 5 9464D-01-2 232DD 00 3 2866D 00
 1.17700 00 4.28940-01-1.97300 00-5 98100 00 1.21630 00 3 60820-02-2 93120 90
6.02490 00 1.19900 00-2.8985D-01-2 15980 00 2.24920 01 1 16880 00-3 92220-01
-2.44400-01 5.46480-01 1.12800 00-4.14130-01-1 89160-01-2 93900 01 1 :0360 00
-4.7101D-01-1.8382D 00 1.0071D 02 1.0895D 00-4 8179D-01 1 0986D 00-3 3966D 02
1.07400 00-6.01870-01-9.09800 00 5 60500 02 1 05400 00-6 17850-01 8 05330 00
-3.4674D 02 1.0370D 00-5.3478D-01-2 6920D 00 4 2997D 01 1 0000D 00
 1.0000D 00 1.5000D 00 4.7203D 00 3.4007D 00 1 0000D 00 5 9625D-01 0 0000D-01
 1.14530 02 1.02220 00 6 69510-01 4 09650 00-7 93320 02 1 03690 00 5 58270-01
-1.39020 01 7.43670 02 1.04700 00 4 31640-01 2 22200 00-1 30490 02 1 06/50 00
 3.9179D-01-3.9165D 00 2.9114D 01 1.0888D 00 1.7159D-01-1 58650 00 7 4008D 00
 1.0980D 00 6.9544D-02-1.0032D 00-6.2259D 00 1 0920D 00-1 7108D-01-2.0007D 00
1.7163D 01 1.0597D 00-2.8655D-01 1.9695D-01-2.5369D 01 1 0334D 00-3 6177D-01
-1.9635D 00-4.6567D 00 9.8600D-01-5 8403D-01-2.4342D 00 1 2063D 02 9 5295D-01
-5 30720-01 4.33450 00 2.01530 02 9.40150-01-3.18640-01 1 02110 01-2 62950 02
 9.34000-01-1.30580-01 2.3174D 00 4.0015D 02 9 3300D-01 1 2767D-01 1 4562D 01
 1.02650 01 9.44000-01 5.83880-01 1.48800 01-2 37660 02 1 00000 00
  0.00000 - 01 \  \, 1.00000 \  \, 00 \  \, 4.51430 \  \, 00 \  \, 7 \  \, 02810 \  \, 00 \  \, 1 \  \, 00000 \  \, 00 - 2 \  \, 73140 \  \, 00 \  \, 0 \  \, 60000 - 01 
 6.6960D 02 9.0741D-01-2.3031D 00 2 3950D 01-4 6342D 02 8 6042D-01-1 8790D 00
1.34360 01-1.52250 02 8.22580-01-1 62340 00 1.01350 01-1 99080 01 7 57080-01
-1.1687D 00 9.1987D 00-7.0346D 01 6 8700D-01-6 5777D-01 3 5690D 00-8 1482D 00
 6.4558D-01-4.0180D-01 2.9268D 00-9 8761D 00 6 1200D-01-5 9646D-02 1 3445D 00
-4.3085D 00 6.1387D-01 7 7029D-02 7.9033D-01-9.4781D 00 6 2232D-01 1 0996D-01
-1 6820D-02 4.7353D 01 6.4150D-01 3 5017D-01 4 7696D CO-1 0745D 01 6 6834D-01
 6.0088D-01 4.1667D 00 3.5252D 02 6.8909D-01 8 7226D-01 1 4446D 01-3 9581D 02
 7.2000D-01 1.12760 00 2 5641D 00-3.5695D 02 7.5400D-01 1 0389D 00-8 3584D 00
 2.3823D 03 7.9400D-01 1.9239D 00 6.5469D 01-1 0457D 03 1 0000D 00
```

TABLE 3

CONTENTS OF RETRIEVAL FILE WITH HEADING

(Sheet 1 of 7)

VERIFICATION	CATION OF DATA	TA TABLE	J.E.					
RETRIEVE	.VAL FILE NO FILE SIZE = L/D = ALFA =	#L3- 17 0.5	3-05-1 .5000 .1250					
ECC RAT	10 EAR FIELD AR FIELD	CON 1.00001	CONSTANT EXPONEN 1001000000 00 1.0000000 1000000000 00.0000000	EXPONENT 0000000000 000000000000000000000000				
POINT	1.090		TAU	COMPARISON FCN	DERIVATIVE :	DERIVATIVE 2	Œ	
	.0000000	0-0	•	1.0000000000000000000000000000000000000	502000000	-00000000000	œ	
~	Ψ.	0-0	. 5768000000 -0	9.4249000000-01	-1.5236000000 00	3000000E	9300000	
m	.208	0-0	.845600000D-0	9 09480000000-01	-1.397100000D 00	4 067100000D 00	44800000	
4	654961	4	8 0138000000-02	ω.	-1.3245000000 00	2.626400000D 00	0	
Ŋ	.0247463	0-0	1.2718000000-01	8.2085000000-01	-1.1876000000 00	0000002	86800000D-0	
9	374875	4	072100	.36	-9.3197000000-01	3 194800000D 00	.726500000	
~	9216	20-0	2 8602000000-01	۲.	. 85	0287000000	.446500000D O	
ω	438904	0-01	462300	Q.	-2.3977000000-01	.5066000000	895900000	
6	265506	0 02	5.7427000000-01	5 8950000000-01	8.6833000000-02	0000006	2071000000 0	
10	874	0 09	6.5943000000-01	. 06	3.2671000000-01	0383000000	29000000	
11	.5316352	٥	7.6051000000-01	6.5722000000-01	6.8826000000-01	7000000	4396000000	
12	.37503490	0	•	7.0303000000-01	9.5757000000-01	2000000	.6015000000	
13	.0469303	30 00	.4578000	7.33400000000-01	1.1286000000 00	.2432000000	۵	
14	.06056648	40 00		7.7026000000-01	1.3318000000 00	7.293100000D 00	6.2869000000 01	
15	1367	00 09	•	8.1471000000-0:	1.5843000000 00	9 2165000000 00	3 4883000000 01	
16	.01352169	0	.3739000	8.6841000000-01	1.8867000000 00	1.0298000000 01	-1.644600000D 02	
1.7	INFINITE		.0000000	1.0000000000 00				

(Sheet 2 of 7)

			18811000110001188
		M	
		00000000000000000000000000000000000000	
		L 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	> r R 8 e R 4 0 8 6 0 e 8 u 8 c 4 4 6 6 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
		\$\begin{align*} 4.0 \text{ \$\mathcal{\text{\$\to}\$}}}} \end{equal}}}}}} \end{picture}}} \$\$ \$\text{\$\mathcal{\text{\$\mathcal{\text{\$\mathcal{\text{\$\mathcal{\text{\$\to}\$}}} \end{picture}}}}} \end{picture}} \$\$ \$\mathcal{\text{\$\mathcal{\text{\$\mathcal{\text{\$\to\$}\end{picture}}}} \end{picture}}}} \$\$ \$\mathcal{\text{\$\mathcal{\text{\$\mathcal{\text{\$\mathcal{\text{\$\mathcal{\text{\$\to\$}}}} \end{picture}}}} \$\$ \$\mathcal{\text{\$\mathcal{\text{\$\to\$}}} \end{picture}}} \$\$ \$\mathcal{\text{\$\to\$}}} \\ \text{\$\to\$} \\ \text{\$\to\$} \\ \text{\$\to\$}} \\ \text{\$\to\$} \\ \text{\$\to\$}} \\ \text{\$\to\$} \\	RUURAUURAFFAOU4TO
		N 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4	######################################
		1 1 1 1 1	71000000000000000000000000000000000000
		600000000000000000000000000000000000000	
		W 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	> 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
		00000000000000000000000000000000000000	70000000000000000000000000000000000000
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		######################################	00000000000000000000000000000000000000
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7		20000000000000000000000000000000000000	20000000000000000000000000000000000000
-	100		X 0 L 3 W W W V V V W 4 L 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
(continued)	1000	40000 TO 00	70000011-110000000000000000000000000000
	7000 7000 7000	0 - 0 0 0 V F 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	0
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74 16	မ်ာင်သ	00000000000000000000000000000000000000	
_			
11000	0 0	140000 156800	+0000m -00000 + 500 6000000000000000000000000000
400	000 000	000 000 000 000 000 000 000 000 000 00	0 10 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
Ċ	0000 0000 0700	00000000000000000000000000000000000000	
Contonta	000	000000000000000000000000000000000000000	
i,	5 -	000 3340 1120 1200 1200 1400 1400 1600 1600 1600 1600 1600 16	0 4 4 2 4 2 4 2 4 2 4 2 4 2 4 2 4 2 4 2
1	20	10000 100000 100000 100000 100	00446479000WWW9-7H
٠,		00000000000000000000000000000000000000	10000000000000000000000000000000000000
Tablo	π <u>α</u> α. α.	ON C = 0 W 4 C = 4 C W 4 C W 4 C W 8 4 C M W W C O C O C O C O C O C O C O C O C O	O N O → O M ★ O → → O M ★ N O → → O → O → O → O → O → O → O → O →
-	A STATE	_ 3	
	-	X - 0 m 4 0 2 5 5 0 0 0 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	$\begin{array}{c} \bullet \\ \Xi & \rightarrow 0 \\ \downarrow \bullet \\ \bullet \\$
	€	04	2

8 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	LOSS LOSS	NSTANT EXPONDED TO 0.0000 100000 00 0.0000 00 0.0000 00 0.0000 00 0	000000 11 00000 11 00000 0000 0000 000	DERIVETIVE 1 7. 7286000000-02 3. 7173000000-02 2. 5686000000-02 1. 1783000000-02 -3. 1042000000-03 4. 5244000000-03 1. 2023000000-03 1. 2023000000-03 5. 2080000000-02 8. 7061000000-02 8. 7061000000-02 9. 7061000000-02 -1. 3935000000 -1. 3935000000 -1. 3935000000 -2. 636000000-03 9. 5984000000-03 9. 5984000000-03	DERIVATIVE 2 1. 0555000000-01 -3. 4787000000-01 -3. 4787000000-01 -1. 4687000000-01 -1. 4687000000-02 -1. 4687000000-02 -1. 3025000000-02 -1. 3025000000-02 -1. 3025000000-02 -1. 3025000000-02 -1. 3025000000 -1. 3025000000 -1. 3025000000 -1. 3025000000 -2. 3413000000 -2. 9413000000 -3. 358000000 -3. 358000000 -3. 358000000 -3. 358000000 -3. 358000000 -3. 358000000 -3. 358000000 -3. 358000000 -3. 358000000 -3. 358000000 -3. 358000000 -3. 358000000 -3. 358000000 -3. 358000000 -3. 3580000000 -3. 3580000000 -3. 3580000000 -3. 358000000000000000000000000000000000000	DERIVATIVE 3 1-2.9511000000 01 2.6434000000 01 2.6434000000 01 2.5245000000 01 2.7507000000 01 4.7336000000 01 -2.5357000000 01 -3.91681000000 01 1.7294000000 01 1.2753000000 01 1.2753000000 01 2.4203000000 01 -3.9162000000 01 -3.9162000000 01 -3.9162000000 01 -3.9162000000 01 -3.9162000000 01 -3.9162000000 01 -3.9162000000 01 -3.9162000000 01 -3.9162000000 01 -3.9162000000 01 -3.9162000000 01 -3.9162000000 01
11 11 22 11 12 4 10 4 10 17	.0469303930 .0605664840 .7524136760 .0135216960 INFINITE	8 4578000000-0 8 7580000000-0 9 0640000000-0 9 3739000000-0 1 0000000000	2493000000-0 6166000000-0 0572000000-0 5896000000-0	1297000000 3257000000 5562000000 9216000000	7123000000 3467000000 7216000000 5860000000	5 4438000000 0 1.2255000000 0 2.6260000000 0 2.5330000000 0
			-			

(continued)

Table 3 - Contents of Retrieval File . . .

Table 3 - Contents of Retrieval File . . . (continued)

	1VE 1 DERIVATIVE 2 DERIVATIVE 10000 00 0 0 0 0000000000000000000000	## PERIVATIVE 1 DERIVATIVE 2 DERIVATIVE 3 000000 00 0.000000000001 1-9.6880000000 02 000000 00 1.7092000000 01 -1.904000000 03 000000 00 1.2964000000 01 -2.280700000 01 000000-01 3973000000 01 -7.1428000000 01 000000-01 2.8564000000 01 -7.1428000000 01 000000-01 2.8564000000 00 -1.166700000 01 000000-03 1.985000000 00 -1.7847000000 01 000000-01 4.557000000 00 -1.7847000000 01 000000-01 4.2567000000 00 -1.7847000000 02 000000-01 1.939600000 00 -1.214200000 02 000000 00 2.3041000000 01 1.2142000000 02 000000 00 3.2351000000 01 -4.3686000000 02 000000 00 1.8813000000 01 -3.0047000000 02
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SECTION IV

FOUNDATIONS OF THE STORAGE RETRIEVAL METHOD

4.1 Choice of the Operating Parameter

The state of static equilibrium of a fluid film journal bearing may be specified either in terms of a static load vector or a static displacement vector. Traditionally, the bearing designer tends to work with the displacement vector not only because the eccentricity ratio is an easy to use design parameter in bearing analysis but also because the bearing designer is concerned with the minimum film thickness which is directly related to the eccentricity ratio. A rotor dynamicist, on the other hand, starts with a known bearing design. Furthermore, the static load condition is usually known. Therefore; the load parameter, as readily computed according to Eq. (2), is the natural choice to define the data point in the table.

The complete range of the load parameter should span from zero to infinity. Goviously it is not possible to treat unbounded numbers in a data table. As a practical matter, some arbitrary cutoff point is accepted when a table is being prepared. At the time of data retrieval, if the requested data point should lie outside the available data range, a judicial extrapolation procedure must be used. The most commonly used extrapolation procedure is the power law, which amounts to a straight line extension of a log-log plot. The desired retrieval method requires the inclusion of such an extrapolation procedure which also retains overall smoothness within the data range.

The graphical extrapolation process can be emulated by a computer program. The power law extrapolation for an infinitesimal load can be expressed as

The new rotor dynamics software [9] has a feature to calculate the load vector of each bearing.

$$\overline{Z} = \alpha \overline{N}^{S} L \qquad \text{for} \qquad \overline{N} << 1 \qquad (48)$$

Its log differential form is

$$\frac{d\overline{Z}}{\overline{Z}} = s_1 \frac{d\overline{W}}{W}$$

Or,

$$s_1 = \frac{d\overline{Z}}{d\overline{W}} \frac{\overline{W}}{\overline{Z}} \qquad \text{for} \qquad \overline{W} <<1$$
 (49)

With the differentials replaced by finite differences and $(\overline{w}, \overline{z})$ approximated by average values, Eq. (49) can be used to calculate s₁ from two data points corresponding to the smallest values of \overline{w} in the data table.

Similarly, the proper law extrapolation for the load parameter tending to infinity is

$$\overline{z} = b\overline{w}^{2}$$
 for $\overline{w} > 1$ (50)

Or,

$$s_2 = \frac{d\overline{Z}}{d\overline{W}} \frac{\overline{W}}{\overline{Z}} \qquad \text{for} \qquad \overline{W} >> 1$$
 (51)

 \mathbf{s}_2 can thus be calculated with the two data points which represent the largest values of $\overline{\mathbf{W}}$ in the data table.

The values of s_1 and s_2 as computed numerically can have either sign and certainly may not always be integers. As illustrated in the π -film short bearing analysis, the factor $(1-\epsilon^2)^{-1/2}$ shows up prominently in various coefficients. Therefore, the values of s_1 and s_2 are rounded off to the nearest half-integer.

Upon fixing \mathbf{s}_1 and \mathbf{s}_2 , the near-field and far-field data behavior can be more accurately represented by

$$\frac{1 \text{im}}{\overline{W} \Rightarrow 0} \quad \overline{Z} = \overline{W}^{S} \mathbf{1} \{ a_{o} + a_{1} \overline{W} + a_{2} \overline{W}^{2} \}$$
 (52)

$$\frac{1}{\overline{W}} = \overline{Z} = \overline{W}^{1} \left\{ b_{0} + b_{1} \overline{W}^{-1} + b_{2} \overline{W}^{-2} \right\}$$

$$(53)$$

and the coefficients a_0 and b_0 , respectively, can be determined numerically from three consecutive data points in each end of the data table. The other coefficients (a_1, a_2, b_1, b_2) are not explicitly required to define the extreme field asymptotes which are

$$\frac{1 \text{im}}{\overline{W} \to Q} \overline{Z} = a_0 \overline{W}^{S} 1; \quad \frac{1 \text{im}}{\overline{W} \to \infty} \overline{Z} = b_0 \overline{W}^{S} 2$$
 (54)

4.2 Full Range Interpolation

An important requirement for a foolproof full range interpolation procedure is to conform to the asymptotic properties uniformly in both the near field and in the far field. This very stringent requirement is satisfied by comparing the data point with a reference function \overline{Z}_0 which has the following properties:

$$\frac{1}{\overline{W}} = \frac{1}{\overline{Z}} = 1$$
 (55)

$$\frac{1 \text{im}}{\overline{W} \to \infty} \frac{|\overline{Z}|}{Z_0} = 1 \tag{56}$$

(3)
$$\overline{Z}_0$$
 (0\infty)>0 and is bounded. (57)

Many simple functions can be concocted to satisfy these conditions for a specific combination of s_1 and s_2 . After some experimentation, the reference function is chosen to have one of three forms depending on the relative values of s_1 and s_2 :

(a)
$$s_1 > s_2$$
 then
$$\overline{Z}_0 = \frac{\left| \frac{a_0 b_0}{\overline{w}^{S_1} + s_2} \right| (1 + \overline{w})}{\left| \frac{a_0}{\overline{w}^{S_1} + 1} + \left| \frac{b_0}{\overline{w}^{S_2}} \right|} \tag{58a}$$

(b)
$$s_2 - 1 < s_1 \le s_2$$
 then
$$\overline{Z}_0 = \frac{|a_0|\overline{W}^{s_1} + |b_0|\overline{W}^{s_2} + 1}{1 + \overline{W}}$$
(58b)

$$\overline{Z}_{0} = |a_{0}|\overline{w}^{1} + |b_{0}|\overline{w}^{2}$$
(5)

(c) $s_2 - 1 > s_1$ then

The interpolation operation is then to be performed on the comparison function, which is simply the ratio between the data point and the rei rence function at the same value of the load parameter; i.e.

(58c)

$$\overline{Z}_{C}(\overline{V}) = \overline{Z}(W)/\overline{Z}_{O}(\overline{W})$$
 (59)

The near-field and far-field values of the comparison function are simply

$$Z_c(W \rightarrow 0) = sg\{a_o\}; \qquad Z_c(W \rightarrow \infty) = sg\{b_o\}$$
 (60)

 \overline{Z}_c is bounded for all values of \overline{W} . Since \overline{Z}_o is everywhere smooth, smooth-

ness, or the lack of it, in the data would be directly reflected by the comparison function. Because \overline{Z}_c has a limited range, its maximum magnitude is usually near unity; its smoothness can be readily verified. Examination of \overline{Z}_c is thus a very effective way to discover inaccuracy in the data table. Furthe, discussion of this question will be pursued in Section 4.4.

Obtaining a $\overline{Z}_{_{\mbox{\scriptsize C}}}$ (\overline{W}) at discrete points of \overline{W} , interpolation can still be problematic if the desired value of \overline{W} exceeds the largest data point. This difficulty is overcome by mapping the semi-infinite range of the load parameter into a finite domain of a working parameter. To assure one-to-one transformation between the load parameter and the working parameter, a monotonic differential relationship is desired. The selected transformation is

$$\tau = \frac{2}{\pi} \tan^{-1}(\overline{W}) \tag{61}$$

The inverse transformation is

$$\overline{W} = \tan(\frac{\pi}{2} \tau) \tag{62}$$

Differentiation of Eq. (61) yields

$$\frac{\mathrm{d}\tau}{\mathrm{d}\overline{\mathrm{W}}} = \frac{(2/\pi)}{1 + (\overline{\mathrm{W}})^2} \tag{63}$$

which is always positive and bounded. One may again note the near-field and far-field asymptotes of Eq. (61)

$$\frac{\lim_{\overline{W}\to\Omega}\tau=\frac{2}{\pi}\overline{W};\qquad \frac{\lim_{\overline{W}\to\infty}\tau=1-\frac{1}{(\pi/2)\overline{W}}$$
(54)

These simple relationships allow one to examine the data points in <u>all</u> ranges of \overline{W} .

With the aic of Eq. (62), the comparison function can now be described in the domain $0 \le \tau \le 1$. $\overline{Z}_c(\tau)$ is in fact assured to be bounded and smooth. Established numerical interpolation schemes can be readily implemented on it. The selected approach is the third order spline function [9], which represents $\overline{Z}_c(\tau_i \le \tau \le \tau_{i+1})$ as

$$\overline{Z}_{c} = \overline{Z}_{c}(\tau_{i}) + a_{i}(\tau - \tau_{i}) + \frac{1}{2}b_{i}(\tau - \tau_{i})^{2} + \frac{1}{6}c_{i}(\tau - \tau_{i})^{3}$$

i spans 1 to N if the full range is divided into N intervals (which may not be uniform). There is some flexibility in the choice of the coefficients at each of the end points where one free condition can be specified. The common convention is to let the outboard second derivative vanish at each end; i.e.

$$b_1 = 0$$
; and $b_N + (\tau_{N+1} - \tau_N)c_N = 0$

Sometimes, if justified by other arguments, one can elect to null the first derivative • either end; i.e.,

$$a_1 = 0$$
; or $a_N + (\tau_{N+1} - \tau_N)b_N + \frac{1}{2}(\tau_{N+1} - \tau_N)^2c_N = 0$

4.3 Correction for L/D Variation

The length to diameter ratio is a major configuration parameter in the design of a fluid film bearing. Typically, in industrial practice, it is in the range of 0.4 - 0.75; occasionally it may be less than 0.25. Unfortunately, there is no standardization of its value. If the data bank were to include a range of L/D, its total storage requirement would be very large indeed. It is, therefore, useful to develop an approach to correct any deviation in the value of (L/D) from that of an available data table.

The short bearing analysis described in Appendix B contains a scaling law which is applicable to the dimensionless load as well as the dimensionless dynamic coefficients:

$$\overline{W}$$
; \overline{K}_{xx} , etc.; \overline{B}_{xx} , etc. $\sim \left(\frac{L}{D}\right)^2$ (66)

For the plain journal bearing, Lund [12] showed that this scaling yields reasonable results even for L/D=1.0 for modest eccentricity ratios (up to about 0.3). It is also clear that such a scaling law would become increasingly unsatisfactory as L/D becomes very large since all such coefficients remain finite for an infinitely long bearing. To improve the situation one requires the scaling factor to assume a finite asymptote as $L/D \rightarrow \infty$ and that a load level dependence be suitably included. A scaling factor which possesses these two properties can be derived by comparing the "half Sommerfeld" solution with the short bearing solution. The "half Sommerfeld" solution is described in Appendix C.

Suppose the scaling factor is defined as

$$\Sigma = \overline{W}/\overline{W}_{sh} \tag{67}$$

 $\overline{\overline{W}}$ is the dimensionless load of any L/D and

$$\overline{W}_{sh} = \frac{1 \text{im}}{L/D + 0} \frac{\overline{W}}{(L/D)^2}$$
(63)

The requirements indicated previously are satisfied if Σ is to be dependent on both L/D and \overline{W}_{\bullet} . Thus

$$\Sigma = \Sigma (L/D, \overline{W});$$

and

$$\lim_{L/D\to 0} \quad \Sigma = (L/D)^2$$

$$\lim_{L/D\to\infty} \Sigma = \Sigma_{\infty} < \infty \tag{69}$$

An empirical formula which is consistent with these conditions is

$$\Sigma = \frac{3}{\alpha^2} \left[1 - \frac{\tanh(\alpha L/D)}{\alpha(L/D)} \right]$$
 (70)

It is seen that

$$\lim_{L/D\to\infty} \quad \Sigma = \Sigma_{\infty} = \frac{3}{\alpha^2} \tag{71}$$

One can compute, with the aid of $\overline{\mathbb{W}}_{\operatorname{sh}}$ and $\overline{\mathbb{W}}_{\infty}$ derived in Appendices B and C,

$$\alpha = \sqrt{\frac{3\overline{W}_{sh}}{\overline{W}_{\omega}}}$$

$$= \sqrt{\frac{(2+\epsilon^2)}{2(1-\epsilon^2)}} - \sqrt{\frac{1 - \{1 - (4/\pi)^2\}\epsilon^2}{1 - \{1 - (2/\pi)^2\}\epsilon^2}}$$
(72)

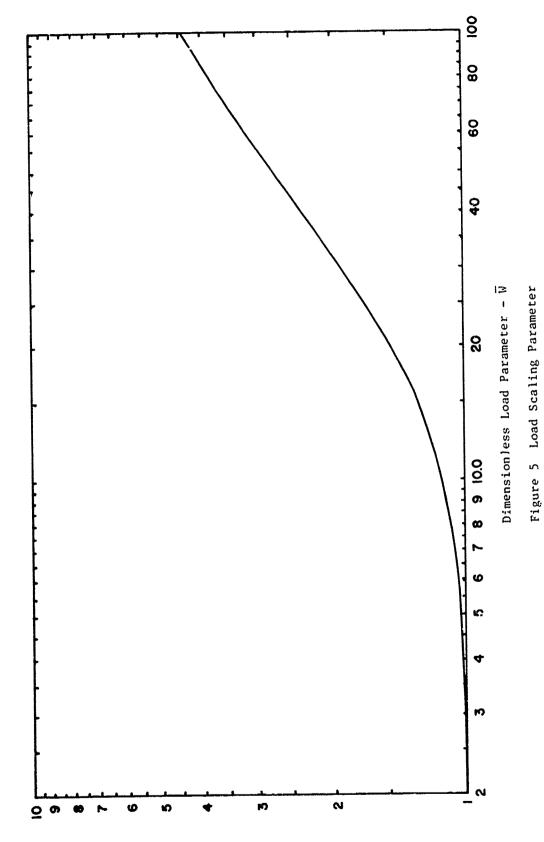
 ϵ , in turn can be regarded as a function of \overline{W}_{∞} . Since Eq. (70) is intended to exclude the effect of L/D, Eq. (72) can be construed to define the load scaling factor

$$\alpha(\overline{W}) = \alpha(\varepsilon(\overline{W}_{m} = \overline{W})) \tag{73}$$

as shown in Fig. 5. It is seen that for a wide range of the load parameter, \overline{W} , α is hardly distinguishable from unity. Consequently, the value of $\alpha = 1.125$ (at $\overline{W} = 10.0$) should be quite satisfactory; thus

$$\Sigma = \frac{\overline{W}}{\overline{W}_{sh}} \approx 2.3704 \left[1 - \frac{\tanh (1.125 \text{ L/D})}{1.125 \text{ (L/D)}} \right]$$
 (74)

which would be used to scale the load parameter as well as the dimensionless dynamic coefficients.



Load Scaling Parameter c

For instance, if a data table for L/D = 0.5 is available, then

$$(\Sigma)_{\text{table}} = 0.22195$$

If the design under consideration is for L/D = 1.0, i.e.,

$$(\Sigma)_{\text{design}} = 0.66518$$

the effective load for the available table should be

$$(\widetilde{W})_{\text{table}} = \frac{0.22195}{0.66518} (\widetilde{W})_{\text{design}} = 0.3367 (\widetilde{W})_{\text{design}}$$

which would be used to interpolate in the data table to obtain

Finally,

$$= \frac{0.66518}{0.22195} (\overline{K}_{xx}, \text{ etc.}; \overline{B}_{xx}, \text{ etc.})_{\text{table}}$$

= 2.9970
$$(\overline{K}_{xx}, \text{ etc.}; \overline{B}_{xx}, \text{ etc.})_{table}$$

In this manner, a data table of a particular value of L/D can be reasonably used for designs of somewhat different L/D. This scaling procedure should not be used if there is significant preloading.

4.4 Data Smoothing

The full range interpolation procedure is sensitive to the presence of inaccurate data points. Numerically computed data points can be inaccurate for a variety of reasons; the most common ones are:

- · faulty algorithm
- · improper mesh (or function) setup, and
- · inadequate convergence control

Unfortunately, such inaccuracies often show up even with "proven" softwares as run by an "experienced" user. Because the full range interpolation procedure is sensitive to inaccuracies in the data table, it can be used as the tool to uncover bad data points which should be adjusted.

Upon compiling a retrieval table, the comparison function and its spline interpolation coefficients (first, second, and third derivatives with respect to the working parameter) in each data interval are printed out. Lack of smoothness can be recognized at a glance by inspection of the sign of the first derivative. An isolated change of sign or a succession of sign reversals prominently mark the presence of bad data points.

Very often the inaccuracy associated with each data point is quite modest. But the error may be of alternate signs for a group of such data points in close range. In this case, the table is readily "fixed" by omitting some of the original data points. In some other instances, neighboring groups of data points depict an abrupt change. This is most likely caused by the necessity of employing a different input setup to obtain the data points of the two groups. It is then necessary to adjust one or more data points to permit smooth interpolation over the full range.

Fig. 6(a) shows an example of how "ripples" would be generated by the full range interpolation procedure. In this particular case, a number of data points of questionable accuracy were contained in the original data table as furnished [3]. "Data clutter" in a range of moderately high load level is probably due to loss of accuracy associate... with the use of finite difference operation on "trajectory" points to compute the impedance

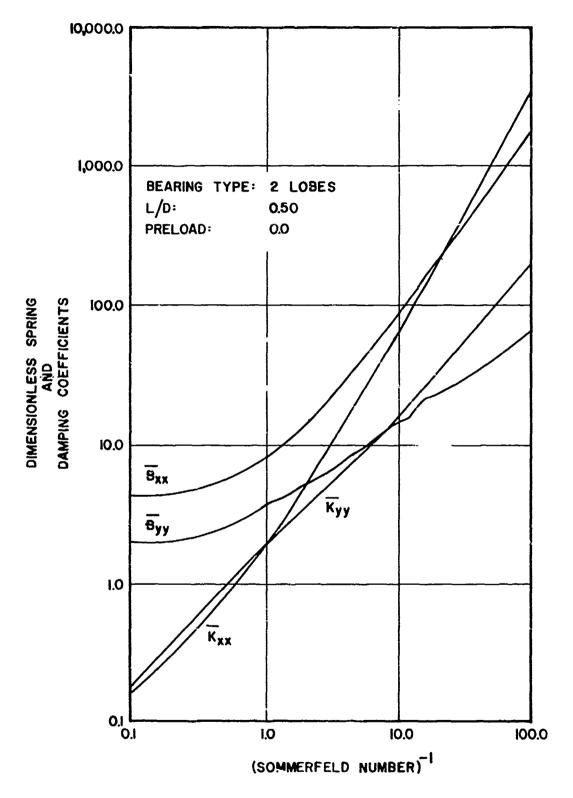


Figure 6(a) Ripples in Original Data

coefficients. Removal of "data clutter" by omitting six of the original seventeen data points permits smooth graphing by interpolation as shown in Fig. 6(b).

A thorough screening of each data table was performed. Adjustments and/or omissions are made where necessary. Therefore, the retrieval files as listed in Appendix D no longer correspond fully to the original data tables.

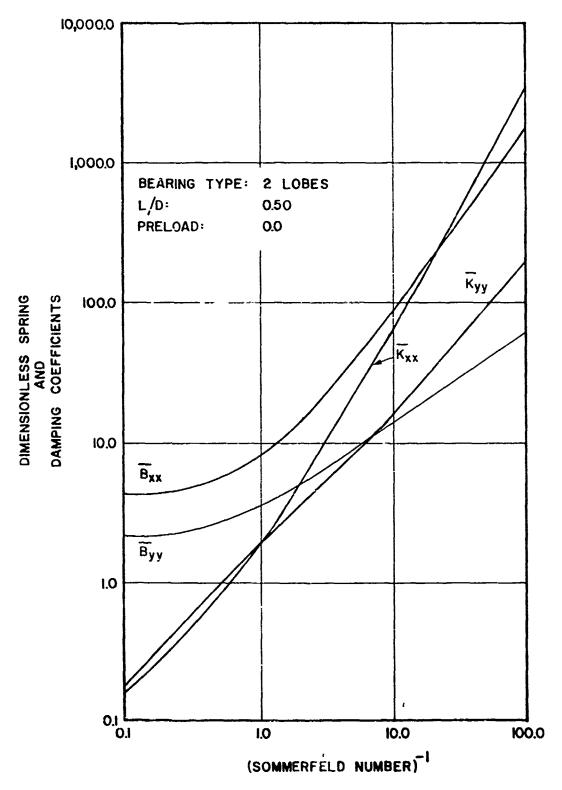


Figure 6(b) Smooth Curves upon Removal of Data Clutter

APPENDIX A

SAMPLE RUNS

The data retrieval software is designed primarily to prepare dynamic data of fluid film bearings for use as part of the input for running the Rotor System Vibrations Program (RSVP) [9]. In addition to extracting the necessary information from a stored data table, at user's command, it performs the supplementary functions of

- · shifting the direction of load vector,
- adjusting the data for any deviation of L/D from that of the data table, and
- incorporation of foundation compliance with allowance for inertia and dissipation effects.

It may also be used simply to prepare a listing of the data table in either the dimensional or the dimensionless form in a load range specified by the user.

The software is coded in FORTRAN and is written in the interactive mode. It can thus be run on any computer system which provides the FORTRAN option and can be commanded from an interactive terminal. The bank of data tables may be located on either tape or disk type mass storage device. Thirty-one data tables are listed in Appendix D. The user may install additional data tables according to the procedure described in Section IV. The format of the data table is described in Section 3.4. The source listing of the software is given in Appendix E.

Several sample runs are shown on the following pages. Each sample includes:

- a record of the interactive session with the prompting messages enclosed in boxes;
- a complete output of bearing data for the applicable operating condition(s), including file identification, static characteristics, table of dynamic characteristics, effective stiffness/damping representation with descriptive headings; and

as required, a listing of the data lines suitable for incorporation into the input setup for running the Rotor System Vibration Program.

The last listing can be punched out into data cards.

Brief descriptions of these sample runs with commentary on notable features are given below:

Sample 1 Synchronous Data with Anisotropic Foundation Compliance in Two Speed Groups, L/D Adjustment Accepted, Inclined Load

L/D of specified bearing (0.6667) is different from that of data table (0.50).

Load vector of bearing is inclined to the vertical direction by 30 degrees.

Foundation data includes inertia (weight) stiffness, and damping coefficients; the latter have distinct values in vertical and horizontal planes.

Sample 2 Synchronous Data with Rigid Foundation in Two Speed Groups, No L/D Adjustment

Same bearing as Sample I, but L/D adjustment is suppressed. Load direction coincides with vertical direction. Same speed points in two groups as Sample I.

Sample 3 Synchronous Data with Rigid Foundation in Two Speed Groups, L/D Adjustment Accepted

By comparison with Sample II, this sample shows the effect of L/D adjustment.

Synchronous Data with Rigid Foundation in Three Speed Groups, L/D Adjustment Accepted, Inclined Load.

By comparison with Sample III, this sample shows that the dynamic coefficients vary with the load direction, but the static characteristics as well as the stiffness/damping representation are independent of the load direction.

This run also shows the capability of generating data for a third speed group.

Sample 5 Asynchronous Data with Rigid Foundation, Two Frequency Groups at the Same Speed, L/D Matched with Data Table

A single data line of dynamic characteristics is shown for each frequency group because such data for the bearing on rigid foundation is independent of frequency.

Sample 6 Asynchronous Data with sotropic Foundation Compliance

Foundation has an isotropic elastic compliance, (reciprocal of stiffness) but no inertia and damping effects. The overall dynamic coefficients become frequency dependent and are thus generated at each frequency.

SAMPLE 1

RECORD OF INTERACTIVE SESSION

```
ENTER DATA TARLE NAME: PI-05-1
FRIFR DUTPUT FILE NAME: DUTPUT
SELECT:
          (1) PREPARE DATA FOR RSVP INPUT
          (2) DIMENSIONAL TABLE
          (3) DIMENSIUNLESS TABLE
          (4) 0011
SELECT: (1) SYNCH FREW, UR
          (2) ASYNUH FREQ?
ENTER BEARING LENGTH (IN)
ENTER BEAPING DIAMETER (IN)
ENTER BEARING CLEARANCE (IN)
 0.001
ENTER VISCOSITY (CENTI-POISE)
 40.0
ENTER BEARING LOAD (LBS)
 750.0
ENTER BEARING COEFS FILE MET RSVP
          (I) USE DEFAULT L7D =
SELECT:
          (2) ADJUST DATA FOR L/D =
 SELECT: (1) LOAD VECTOR IN VERTICAL PLANE, UR
          (2) INCLINED LOAD VECTOR?
      ENTER LOAD INCLINATION ANGLE (VEG)?
 SELECT: (1) RIGID FOUNDATION
          (2) ISOTROPIC FOUNDATION COMPLIANCE: OR
          (3) ANISOTROPIC FOUNDATION COMPLIANCE?
         (I) RADIAL BEARING, OR
 SELECT:
          (2) ANGULAR BEARING?
      ENTER PEDESTAL WEIGHT (LB)?
      ENTER RECESTAL STIFFHESS IN VERTICAL PLANE?
      (LB/IN) FOR RADIAL BRG, OR (IN-LB/RAD, FOR HNG BRG
      ENTER PEDESTAL STIFFNESS IN HURIZUNTAL PLANE
      (LB/IN) FOR RADIAL BRG, OR (IN-LB/RAD) FOR HMG BPG
 1.00+04
      ENTER PEDESTAL DAMPING IN VERTICAL PLANE?
      (LB-SEC/IN) FOR RAD BRG, OR (IN-LB-SEC/RAD) FOR ANG BPG
```

Sample 1 - Record of Interactive Session (continued)

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ENTER PEDESTAL DAMPING IN HURIZUNTAL PLANE?
     (LB-SEC/IN) FOR RAD BRG, OR (IN-LB-SEC/RAD) FOR ANG BPG
ENTER LOWEST SPEED (RPM)
6000.0
ENTER HIGHEST SPEED (RPM)
9000.0
ENTER NUMBER OF SPEED POINTS (NOT MORE THAN 11)
 SELECT - (1) ANOTHER SPEED GROUP
          (2) QUIT
ENTER LOWEST SPEED (RPM)
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ENTER HIGHEST SPEED (RPM)
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ENTER NUMBER OF SPEED POINTS (NOT MORE THAN 11)
 SELECT - (1) ANOTHER SPEED SROUP
          1100 (57
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0.6667	FRICT(HP)	8.60800-04	9.3257D-04	1.01050-03	1.09510-03	1.1871D-03	1.2869D-03	1 39530-03	1.51300-03	1.6410D-03	1.7799D-03	1.93090-03
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Sample 1 - Tabulation of Bearing Data (continued)

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SAMPLE 1

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RECORD OF INTERACTIVE SESSION

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ENTER DATA TABLE NAME: PJ-05-1
ENTER OUTPUT FILE NAME: OUTPUT
          (1) PREPARE DATA FOR RSYP INPUT
SELECT:
          (2) DIMENSIONAL TABLE
          (3) DIMENSIONLESS TABLE
          (4) QUIT
SELECT:
          (1) SYNCH FREQ, OR
          (2) ASYNCH FREQ?
ENTER BEARING LENGTH (IN)
ENTER BEARING DIAMETER (IN)
ENTER BEARING CLEARANCE (IN)
0.001
ENTER VISCOSITY (CENTI-POISE)
 40.0
ENTER BEARING LOAD (LBS)
 750.0
ENTER BEARING COEFS FILE NAME: RSVP
          (1) USE DEFAULT L/D = 0.5000
          (2) ADJUST DATA FOR L/D =
          (1) LOAD VECTOR IN VERTICAL PLANE, OR
SELECT:
          (2) INCLINED LOAD VECTOR?
SELECT: (1) RIGID FOUNDATION
         (2) ISOTROPIC FOUNDATION COMPLIANCE, OR
         (3) ANISOTROPIC FOUNDATION COMPLIANCE?
ENTER LOWEST SPEED (RPM)
6.000.0
ENTER HIGHEST SPEED (RFM)
9000.0
ENTER NUMBER OF SPEED POINTS (NOT HORE THAN 11)
```

Sample 2 - Record of Interactive Session (continued)

```
SELECT - (1) ANOTHER SPEED GROUP

(2) QUIT

1
ENTER LOWEST SPEED (RPM)
9000.0
ENTER HIGHEST SPEED (RPM)
12000.0
ENTER NUMBER OF SPEED POINTS (NOT MORE THAN 11)
11
SELECT - (1) ANOTHER SPEED GROUP

(2) QUIT
```

		11 PTS
<u>18 2</u>	BEARING DATA	SPEED RANGE (RPM) 0 6000.00 TO 9000.00
SAMPLE 2	TABULATION OF BEARING DATA	VISC (CP) W (LBS) 4.0000 01 750.00
PJ-05-1 17	0.5000	N) C (1N)
RETRIEVAL FILE NO	; ; ; • • • • •	L (IN) DIA (IN)

A-LOST(GPM) 5.14030-02 5.23910-02 5.33720-02 5.43430-02 5.53040-02 5.62500-02 5.71810-02 5.89870-02 5.89870-02 5.98620-02 5.98620-02
0-RED(GPM) 8.63990-02 8.94330-02 9.25670-02 9.58070-02 9.91550-02 1.02610-01 1.06190-01 1.09880-01 1.17650-01
FRICT(HP) 8.8661D-04 9.5796D-04 1.0353D-03 1.1191D-03 1.2099D-03 1.3084D-03 1.4152D-03 1.5312D-03 1.5312D-03 1.9422D-03
ATT ANGLE 6.1276D 01 6.1907D 01 6.2539D 01 6.3172D 01 6.3804D 01 6.4434D 01 6.5060D 01 6.5682D 01 6.6299D 01 6.6299D 01
ECC(IN) 4.5916D-04 4.3975D-04 4.3975D-04 4.2030D-04 4.1056D-04 4.0082D-04 3.9108D-04 3.9167D-04 3.6203D-04
RPH 6.00000 03 6.24830 03 6.7610 03 7.05650 03 7.34850 03 7.65250 03 7.96920 03 8.29900 03 8.64240 03

8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8
8-77 5650 4870 4120 2210 2040 1410 0210 9650
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K-YY 5280 53310 53310 53310 15340 15450 15450 15450 15510 15510 15510
00000000000000000000000000000000000000
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8-4X 9160 79460 5660 2560 2530 1660 9910 9010
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X-4X 0250 0250 1250 1270 2310 2440 4040 5300 5300
003-1 003-1 003-1 003-1 03-1
8-XY 9160 7940 5660 2590 1660 9910 9910
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8-XX 7-7460 7-4460 6-9870 6-7620 6-3500 6-3500 6-3500 6-3500 6-3500 6-3500 6-3500 6-3500 6-3500
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000000000000000000000000000000000000000
RPM 6.2480 6.2480 6.2480 7.3480 7.3480 7.6530 7.6530 8.2990 9.0000

Sample 2 - Tabulation of Bearing Data

11 PTS
SPEED RANGE (RPM) 9000 00 TO 12000.00
750 00
(IN) C (IN) VISC (CP) W (L83) 5000 0.001000 4.0000 01 750 00
C ( IN ) 0.001000
010 (IN) 1.5000
( IN )

3 3 1	D 7 . B.X	5.30470-01	21150-		3176	32290-		32/00/10	33180-		33260-	0000		34250-	3456		5 6485U-U1	
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0-L0ST(GPK)	6 07170-02	13120		2001	24710	70750		32890	41200		46640	2000		34000	0 0	3.10	
-REQCGP	1 21730-01	04710		.2777D	1 30900-01		10-01146 1	1 37390-01	TO THE TOTAL	3	1.44220-01		7010000	1 51400-01		1 55130-01	
FRICT(HP)	94220-	2 4 4 4	101000	17480-	20170-03		43620-	-06825		73010-	89040-		06030-	24066-	3	4317D-	
AMG	75200 0		00050.	83770		0 0	.92270	96480		08900.	04850		00060	2110	7177	17200	) }
ECC(1N)	0-02000	0 40040	55230-0	48470-0	1000	D-00/14.	35070-0	0.0440.		21860-0	0-07621		0-08880		0.08420	0-141-0	2
Σ 0	2020	00000.	2627D	4 4 0 0	000000	81120	00000		02250	9		2	12290		1.16600 04	6	0000

Sample 2 - Tabulation of Searing Data (continued)

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99999999999999999999999999999999999999
K-YY 1 5600 1 5600 1 5600 1 5600 1 5700 1 5720 1 5770 1 5770
00000000000000000000000000000000000000
8-4% 1 9090 1 8530 1 7980 1 6940 1 5970 1 5970 1 5050 1 4610
00000000000 0000000000
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000000000000000000000000000000000000000
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0000000000 99990000000
2.9510 2.9510 3.9530 3.0280 3.0280 3.0520 3.1520 3.1540
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8-XX 6600 3550 3550 3550 3550 3550 1760 0910 9320 8590 7880
00000000000000000000000000000000000000
X-XX-X-X-X-X-X-X-X-X-X-X-X-X-X-X-X-X-X
00000000000000000000000000000000000000
8 110 0000 0330 0330 0330 0330 1010 1010

PARAMETERS F RATIO S 34860-01 5 35240-01 5 35240-01 5 35420-01 5 35600-01 5 35780-01 5 36320-01 5 36430-01 5 36430-01
STABILITY CP MASS 1 04250 04 1 04410 04 1 04890 04 1 05040 04 1 05180 04 1 05320 04 1 05590 04 1 05590 04 1 05590 04
******  MAJ DIES  2.13810 00  2.24150 00  2.25600 00  2.35250 00  2.47150 00  2.53420 00  2.53420 00  2.53420 00  2.53420 00  2.53420 00  2.53420 00
11STICS***********************************
RG CHARACTER MIN DHMF 7 59860-01 7 73720-01 8 02570-01 8 17570-01 8 32980-01 8 48790-01 8 8 8 9 9 00-01 9 16490-01
**SYNCH B STIFF 0284D 06 0404D 06 0524D 06 0524D 06 0524D 06 07005 06 07005 06 0815D 06 0872D 06
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(Page 3 of 3)

SAMPLE 2

# DATA LINES IN RSVP FORMAT

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## RECORD OF INTERACTIVE SESSION

```
ENTER DATA TABLE NAME: PJ-05-1
ENTER OUTPUT FILE NAME: OUTPUT
SELECT:
          (1) PREPARE DATA FOR RSYP INPUT
          (2) DIMENSIONAL TABLE
          (3) DIMENSIONLESS TABLE
          (4) QUIT
SELECT:
          (1) SYNCH FREQ, OR
          (2) ASYNCH FREQ?
ENTER BEARING LENGTH (IN)
ENTER BEARING DIAMETER (IN)
ENTER BEARING CLEARANCE (IN)
001
ENTER VISCOSITY (CENTI-POISE)
 40.Ū
ENTER BEARING LOAD (LBS)
 750.0
ENTER BEARING COEFS FILE NAME: RSVP
SELECT:
          (1) USE DEFAULT L/D =
                                    D 5000
          (2) ADJUST DATA FOR L/D =
SELECT:
          (1) LOAD VECTOR IN VERTICAL PLANE, OR
          (2) INCLINED LOAD VECTOR?
         (1) RIGID FOUNDATION
SELECT:
          (2) ISOTROPIC FOUNDATION COMPLIANCE, OR
          (3) ANISOTROPIC FOUNDATION COMPLIANCE?
ENTER LOWEST SPEED (RPM)
 6000.0
ENTER HIGHEST SPEED (RPM)
 9000 0
ENTER HUMBER OF SPEED POINTS (NOT MORE THAN 11)
```

## Sample 3 - Record of Interactive Session (continued)

m	ARING DATA	SPEED RANGE (RPM) 6000.00 TO 9000 00
SAMPLE 3	TABULATION OF BEARING DATA	4 (LBS) 756.00
	TABUI	L (IN) DIA (IN) C (IN) \$1SC (CP) W (LBS) 1 0000 1.5000 0.001000 4.0000 01 756.00
PJ-05-1	0.5000 1.1250	C (IN) 0.001000
FILE NG		DIA (IN) 1.5000
RETRIEVAL	FILE SIZE = 17 L/D = 0.5000 ALFA = 1.1250	L (1N) 1 0000

	0-L0ST(GPM 3.82000-02 3.86850-02 3.96140-02 4.0590-02 4.04890-02 4.05050-02 4.153050-02 4.15880-02 4.15880-02
	0-REQ(GPM) 8 00480-02 8.28270-02 8.57050-02 8.97750-02 9.49760-02 9.82940-02 1.01730-01 1.05300-01 1.05300-01
0.6667	FRICT(HP) 8.6080D-04 9.3257D-04 1.0105D-03 1.0951D-03 1.2869D-03 1.3953D-03 1.5130D-03 1.6410D-03 1.7799D-03
FOR L/D =	ATT ANGLE 6.88050 01 6.94020 01 6.99940 01 7.05820 01 7.11640 01 7.23120 01 7.28730 01 7.34250 01 7.34660 01 7.44940 01
IS ADJUSTED	ECC(IN) 3.41710-04 3.32320-04 3.23020-04 3.04760-04 2.95810-04 2.86980-04 2.78290-04 2.61280-04
LOAD PARAMETER IS ADJUSTED FOR L/D =	RPM 6.00000 03 6.24830 03 6.77610 03 7.05650 03 7.65250 03 7.65250 03 7.65250 03 8.29900 03 8.29900 03

20000000000000000000000000000000000000
8-47 6.2190 6.1410 6.0660 5.9930 5.8560 5.7920 5.7920 5.6160 5.6160
K-YY 03 1.5660 03 1.5660 03 1.5720 03 1.5720 03 1.5760 03 1.5830 03 1.5850 03 1.5850
8-4% 8540 62340 51770 7150 7150 7210 7210 7210 7210 7210 7210 7210 721
99999999
74-48 8899 88999 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 11506 1
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8-37 8740 7370 6-250 5170 41140 22210 1310 0440 8840
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K-XY 0010 0400 0820 1280 1770 2300 2870 2870 3480 4130 4130 4130
003
8-xx 7600 3500 3500 1630 9890 8240 6710 5260 3920 2650 1480
00000000000000000000000000000000000000
000000000000000000000000000000000000000
K-XX 1640 11420 11220 1020 0840 0490 0490 0170 00170 00250
000000000000000000000000000000000000000
8 2480 6 2480 6 2480 7 3480 7 3480 7 8530 8 2990 8 6420 9 0000
000000000000

2993 0

LOAD AND DYNAMIC COEFFICIENTS ARE ADJUSTED FOR L/D

3 (Page 2 of

5.35920-01 5.36180-01 5.36410-01 5.36680-01 5.36950-01 5.37220-01 PARAMETERS 5.35420-01 37720-01 5.35680-01 F PATIO 4 0 4 0 4 4 49 40 STABILITY 1.05540 06270 1.04740 .05160 COULT GO 1.04950 1.05350 02860 00090 .06140 06410 2.29630 00 2.3764D 00 00 00 00 00 000 00 MAJ DAMP 2.64230 2.74230 2.84270 2.95030 3.06310 2.46060 2.5492D 18130 90 90 90 90 90 STIFF 90 1.55700 .68360 .65660 .60520 . 58060 63040 .51230 .49130 47140 8.68880-01 8.92720-01 9.17420-01 0 9.43110-01 9.69890-01 97860-01 8.02640-01 8.23900-01 8.45970-01 MIN DAMP .02710 1 04650 06 1.0549D 06 37175 90 90 90 90 90 1.06320 1.07140 1.07950 1.08750 1.09530 1029D 11040 .11770 12470 03 6.77610 7.65650 7.34650 7.65250 00000 6.2483D 6.50680 8.29900 64240 00000

Sample 3 - Tabulation of Bearing Data (continued)

)	2	4.26380-02	. 28640	30810	င္တ	9	.36740	<u>م</u>	.40220	1830	4.43340-82
	Q-REQ(GPM)	1.12830-01	1	۵	ا دې	1.24490-01	1.27600-01	۵	ا ا ا	7430-	1.40880-01	1,44420-01
0.6667	O:	930	-06	.16790-	.29730-	.4345D-	.58010-	. 734	89850-	07230-	. 256	45240-0
FOR L/D =	ATT ANGLE	44940 0	48600	2190	. 55	9140	.62490	7.65750 01	8930	_	_	7.7793D 01
IS ADJUSTED	ECC(IN)	0	47180-0	41450-	•	30220-0	24720-	0-00	139	780	C	984
LOAD PARAMETER	æ Q	9	26270	53300	81120	008800	03.650	1 19	10080	13290	009	O

0.6667

Sample 3 - Tabulation of Bearing Data (continued)

	8-Y8	. 5630	. 5280	4940	4620	4310	.4020	3750	3490	3258	3020	.2800	
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		03 1	03.1		03 1	03 1			03 1	03	03	03	
	8-78	4 ت	<u>ت</u>	0622	ದ	.6810	30	90	20	20		0.9	
567		90	90	90	90	90	90	90	90	90	90	90	
ယ ထ	K- Y &	6390	2125	1962	8780	9620	0490	1380	2300	3240	4210	5210	
= 0/7		1	1	- 1	ŧ	1	t	1	1	1	03-3	1	
ADJUSTED FOR	%×-8	1.8840	1.8310	1.7790	. 72	1.6810	S. W	59	1.5470	1.5050	1.4650	1.4260	
Sn		90	90	90	90	90	90	90	0.5	90	90	90	
ARE ADJ	K-X4		9	. 67	. 73	. 79	. 86	. 93	4.0060	. 08	4.1610	4.2430	
3		03	03	03	03	03	03	63	03	03	03	03	
FFICIENT	8-8	1.4	90	9	.92	8	62.	٠. د	. 67	.61	6.5630	. 51	
COE		05	05	03	0.5	0.5	05	0.5	02	0	05	0.5	
LOAD AND DYNAMIC	×××××××××××××××××××××××××××××××××××××××	88.6	9.79	9 70	9.62	9.54	9.47	9.40	9.33	9.27		9 16	
۵ ۲											40		
L0AD A	a.			9.5330				•	1.1010	1.133D	1.1660	1.2000	

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(Page 3 of 3)

SAMPLE 3

DATA LINES IN RSVP FORMAT

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RECORD OF INTERACTIVE SESSION

```
ENTER DATA TABLE NAME: PJ-05-1
ENTER OUTPUT FILE NAME: OUTPUT
SELECT:
          (1) PREPARE DATA FOR RSVP INPUT
          (2) DIMENSIONAL TABLE
          (3) DIMENSIONLESS TABLE
          (4) QUIT
SELECT:
          (1) SYNCH FREQ, OR
          (2) ASYNCH FREQ?
ENTER BEARING LENGTH (IN)
ENTER BEARING DIAMETER (IN)
ENTER BEARING CLEARANCE (IN)
0.001
ENTER VISCOSITY (CENTI-POISE)
40.0
ENTER BEARING LOAD (LBS)
750.0
ENTER BEARING COEFS FILE NAME: RSVP
SELECT:
          (1) USE DEFAULT L/D =
                                    0.5000
          (2) ADJUST DATA FOR L/D =
                                        0.6667
SELECT:
          (1) LOAD VECTOR IN VERTICAL PLANE, OR
          (2) INCLINED LOAD VECTOR?
     ENTER LOAD INCLINATION ANGLE (DEG)?
30.0
         (1) RIGID FOUNDATION
         (2) ISOTROPIC FOUNDATION COMPLIANCE, OR
         (3) ANISOTROPIC FOUNDATION COMPLIANCE?
ENTER LOWEST SPEED (RPM)
6000.0
ENTER HIGHEST SPEED (RPM)
9000.0
ENTER NUMBER OF SPEED POINTS (NOT MORE THAN 11)
 SELECT - (1) ANOTHER SPEED GROUP
          (2) QUIT
ENTER LOWEST SPEED (RPM)
9000.0
ENTER HIGHEST SPEED (RPM)
12000 0
ENTER NUMBER OF SPEED POINTS (NOT MORE THAN 11
```

Sample 4 - Record of Interactive Session (continued)

```
SELECT - (1) ANOTHER SPEED GROUP

(2) QUIT

ENTER LOWEST SPEED (RPM)

12000.0

ENTER HIGHEST SPEED (RPM)

15000.0

ENTER NUMBER OF SPEED POINTS (NOT MORE THAN 11)

S

SELECT - (1) ANOTHER SPEED GROUP

(2) QUIT
```

			11 PTS
	SAMPLE 4	TABULATION OF PEARING DATA	(IN) C (IN) VISC (CP) W (LES) SPERD PHREE (N. N. STAND OF 11 PTS 50.00 00 TO 10 PTS
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	0.1 L0.0 T C C C C C C C C C C C C C C C C C C
	0-8E0(6PH) 8 28270-02 8 86860-02 9 17750-02 9 82940-02 1 05300-01 1 08990-01
0 6667	FPICTCHF - 8 . 60800-04 9 . 32570-04 1 01550-03 1 18710-02 1 28690-03 1 . 39520-03 1 . 64100-03 1 . 64100-03 1 . 64100-03 1 . 53090-03 1 . 53090-03 1 . 53090-03
IS ADJUSTED FOR LZD =	######################################
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Sample 4 - Tabulation of Bearing Data (continued)

STABILITY PARAMETERS CR MASS F RATIO 1.04740 04 S 35420-01 1.04950 04 S 35680-01 1.05160 04 S 35920-01 05350 04 S 35680-01 05540 04 S 36680-01 1.05710 04 S 36680-01 1.05860 04 S 36950-01 1.06000 04 S 37220-01 1.06140 04 S 37720-01 1.06270 04 S 37720-01 1.06410 04 S 37720-01
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Sample 4 - Inbuletion of Bearing Data (continued)

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	99999999999999999999999999999999999999
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	RPM 0000 2630 5330 6110 0100 0390 0700 1010 1330
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PARAMETERS F RATIO 5.37940-01 5.38090-01 5.38220-01 5.38320-01 5.38510-0 5.38620-0 5.38640-0 5.38680-0
444444444
STABILITY CR MMSS 1.0651D 0 1.0650D 0 1.0669D 0 1.0690D 0 1.0690D 0 1.0723D 0 1.0723D 0 1.0723D 0 1.0733D 0
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1STICS** NAJ STI 1.4524D 1.4235D 1.4271D 1.4152D 1.4152D 1.3825D 1.3825D 1.3630D 1.3630D 1.3540D
CTER 000 000 000 000 000
RG CHARA MIN DAM 1.0577D 1.1038D 1.1038D 1.1280D 1.1531D 1.1531D 1.2061D 1.2630D 1.2630D 1.2630D 1.2630D
11 00 00 00 00 00 00 00 00 00 00 00 00 0
******SYN 1.12470 1.12470 1.13430 1.13430 1.14350 1.14350 1.15210 1.15620 1.16400
# 000000 # WWWW444444
***** RPH 9.00000 9.26270 9.53300 9.81120 1.00980 1.03920 1.06960 1.10080 1.16600 1.16600

(Page 3 of 4)

Sample 4 - Tabulation of Bearing Data (continued)

5 PTS
) SPEED RHNGE (RPM) 0 12000 00 TO 15000.00
W (LBS) 750 00
N) C (1N) VISC (CP) W (LBS) 00 0,001000 4,0000 01 750 00
C < 182 0.001000
01A (IN) 1 5000
C 8 8 9

3	Q-LOST(GPM) 4.43340-02 4.48520-02 4.50790-02 4.50790-02
	0-REQ(GPM) 1 44420-01 1.51560-01 1.59090-01 1.67030-01
299910	FRICT(HP) 3,45240-03 3,86640-03 4,33040-03 4,85050-03 5,43320-03
FOR L/0 =	ATT ANGLE 7.77930 01 7.83360 01 7.88570 01 7.93610 01 7.98530 01
IS ADJUSTEC	ECC(IN) 1.98430-04 1.88820-04 1.79570-04 1.62200-04
LOAD PARAMETER	RPM 1 20000 04 1 26880 04 1 34160 04 1 41860 04 1 50000 04

	00000
	8-44 6.8230 6.7090 6.6040 6.5050
	9000 900 900
	K- YY 1.7390 1.7230 1.6930 1.6930
	00000
	B - Y X 1 . 24 70 1 . 18 90 1 . 13 60 1 . 08 70 1 . 04 30
_	
S S	90000
	K-YX 03-3.996D 03-4.194D 03-4.404D 03-4.626D 03-4.861D
	Waaaa
1/1	m m m m m
F08 L/D	8-XY 1 2470 1 1890 1 1360 1 0870 1 0430
<u>ت</u>	
Ω	00000 9999
704	X-XY 3.768D 3.940D 4.126D 4.325D 4.5385D
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S	00000
PFICIENTS ARE ADJUSTED	8-XX 4 9700 4 9560 4 9450 4 9310 4 9310
COE	0 0 0 0 0 0 0
CAR AND DYNAMIC	K-XX 04 7.7380 04 7.8110 04 7.8840 04 7.9560
Z 0 0 0 0	RFM 1 2690 1 2690 1 3420 1 4190

STABILITY PARAMETERS CF MASS F PATIO 1.07450 04 5.38640-01 1.07680 04 5.38640-01 1.07680 04 5.38620-01 1.08040 04 5.38650-01 1.08140 04 5.38760-01	(Page 4 of 4)
**************************************	

## A LINES IN RSVP FORMAT

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## RECORD OF INTERACTIVE SESSION

```
ENTER CATA TABLE NAME: PJ-05-1
ENTER OUTPUT FILE NAME: OUTPUT
SELECT:
          (1) PREPARE DATA FOR RSVP INPUT
          (2) DIMENSIONAL TABLE
          (3) DIMENSIONLESS TABLE
          (4) QUIT
          (1) SYNCH FREQ, OR
SELECT:
          (2) ASYNCH FRE0?
     ENTER NUMBER OF FREQ GROUPS?
ENTER BEARING LENGTH (IN)
ENTER BEARING DIAMETER (IN)
2.0
ENTER BEARING CLEARANCE (IN)
ENTER VISCOSITY (CENTI-POISE)
40.0
ENTER BEARING LOAD (LBS)
800.0
ENTER BEARING COEFS FILE NAME: RSVP
SELECT:
          (1) LOAD VECTOR IN VERTICAL PLANE, OR
          (2) INCLINED LOAD VECTOR?
SELECT:
         (1) RIGID FOUNDATION
         (2) ISOTROPIC FOUNDATION COMPLIANCE, 3R
         (3) ANISGTROPIC FOUNDATION COMPLIANCE?
ENTER LOWEST SPEED (RPM)
 7500.0
ENTER HIGHEST SPEED (RPM)
 7500.0
ENTER NUMBER OF SPEED POINTS (NOT MORE THAN 11)
     ENTER FREQ DATA (HZ) FOR FREQ GROUP
                                            1 AT 7.5000D 03 RPM
           LOWEST FREQ?
 25 0
          HIGHEST FREQ?
 50 n
          NUMBER OF FREQ POINTS?
     ENTER FREQ DATA (HZ) FOR FREQ GROUP
                                            2 AT 7.5000D 03 RPM
           LOWEST FREQ?
 50.0
          HIGHEST FREQ?
100.0
          LOWEST FRER?
   HIGHEST FREQ?
           NUMBER OF FRED POINTS"
  SELECT - (1) ANOTHER SPEED GROUP
           (2) QUIT
```

	9	
	1 4 1	Q-LOST(GFM) 4.25400-02
	SPEED RANGE (RPM) 10.00 TO 7500.00	0-L0 -25
	D RANG	0-REQ(GPM) 1.2313D-01
ING DATA	SPEED RANGE (RPM) 7500.00 TO 7500.00	
SAMPLE 5 TABULATION OF BEARING DATA	00.008 800.00	FRICT(HP) 3.1841D-03
TABUI	VISC (CP) W (LBS) 4.0000 01 800.00	ATT ANGLE 7.60220 01
PJ-05-1 17 0.5000 1.1250	C (IN) 0.001000	ECC(IN)
112E NO	01A (1N) 2.0000	
m at		c
RETRIEVAL FILE NO FILE SIZE ** L/D **	L (1N)	7
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1 PTS

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3590
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90
X-X- €69
0.3
B-XX K-XY B-XY K-YX B-YX K-YY B-YY 9.9050 03 4.0730 06 2 4150 03-3 1890 06 2.4150 03 1.6990 06 7 8590 03
90
K- YX 1890
03-3
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O
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K-XY . 0736
4
03
8-XX 9.9050
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5000 K
7.5 01
RPM FREG 5.0000

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RPH 7.50000 03

Sample 5 - Tabulation of Bearing Data (continued)

0 3
8-YY 7 8590
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90
K-YY 1.6990
63
8-XX K-XY 8-XY K-7X 8-YX K-YY 8-9.9050 03 4.0730 06 2 4150 03-3.1890 06 2.4150 03 1.6990 06 7 88
26 82.
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63-3
8-XY 4150
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90
K-XY . 0730
4
ő
8-XX 9050
90
7.5000 03 K-XX 01 1.016D
000 000 X.0
7.5 01
RPM = 7.9 FREQ 2.5000 01

* 00001100110011011011011011011
****** 0RIENT 2.5180 2.5180 2.5240 2.5250 2.6250 2.6250 2.7660 2.7600
10R MODE****** 1P ELLIPTIC 00-7.703D-01 00-7.643D-01 00-7.579D-01 00-7.511D-01 00-7.362D-01 00-7.194D-01 00-7.103D-01 00-7.103D-01 00-7.0060-01
JOR MRP EI 000-7 000-7 000-7 000-7 000-7 000-7 000-7 000-7 000-7 000-7 000-7 000-7 000-8
**************************************
*"00000000000
**************************************
00000000000000000000000000000000000000
******** 0RIENT 1.7600 1.7310 1.6620 1.6620 1.6520 1.5500 1.5500 1.5500 1.5500 1.5500 1.5500 1.5500
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HSYNCHRONDUS BEARING CHARACTERISTICS

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## APPENDIX B

## SHORT JOURNAL BEARING

The short bearing analysis of Dubois and Ccvirk [10] makes available explicit analytical expressions for the dynamic perturbation coefficients of circular and journal bearings valid for  $L/D \le 0.25$ . Although these results are not precise enough for direct use at larger values of L/D, they still yield valid qualitative trends between the dynamic perturbation coefficients and static operating condition. The essential elements of this important work are summarized here.

Under the assumptions of an isoviscous lubricant and negligible misalignment, the governing equation for the film pressure of a short circular arc bearing is reduced to

$$\frac{\partial p}{\partial z^2} = \frac{6\mu}{h^3} \left(\omega \frac{\partial}{\partial \theta} + 2 \frac{\partial}{\partial t}\right) h \tag{B-1}$$

The film thickness as illustrated in Fig. B-1, is given by the equation

$$h = C\{1-\varepsilon \cos (\theta-\psi)\}$$
 (B-2)

Separating static equilibrium and dynamic perturbation parts, one can write

$$\varepsilon = \varepsilon_{o} + \delta \varepsilon$$
;  $\psi = \psi_{o} + \delta \psi$ ;  $h = h_{o} + \delta h$ 

$$h_{o} = C\{1 - \epsilon_{o} \cos (\theta - \psi_{o})\}$$
 (B-3)

$$\delta h = -\delta \epsilon \cos (\theta - \psi_0) - \epsilon_0 \delta \psi \sin (\theta - \psi_0)$$

Integration of Eq. (B-1) with respect to z twice, one finds

$$p = \frac{3\mu}{h^3} \left(z^2 - \frac{L^2}{4}\right) \left(\omega \frac{\partial}{\partial \theta} + 2\frac{\partial}{\partial t}\right)h \tag{B-4}$$

Integrating over the full bearing length, one obtains

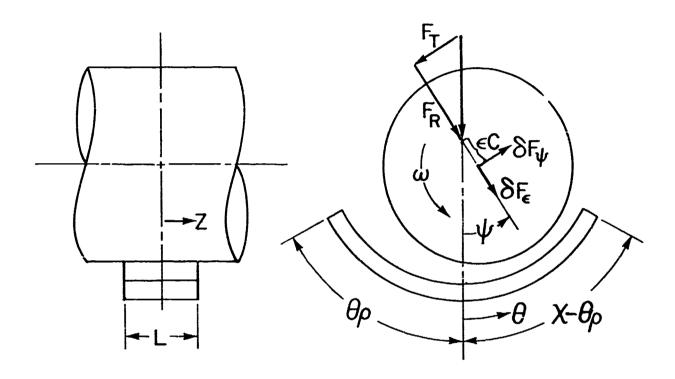


Figure B-l Description of a Short Circular Arc Bearing

$$q = \int \frac{L}{2} p dz = -\frac{\mu L^3}{2h^3} \left(\omega \frac{\partial}{\partial \theta} + 2 \frac{\partial}{\partial t}\right) h$$
 (B-5)

Substituting Eq. (B-3) into Eq. (B-5) and separate static equilibrium and dynamic perturbation parts, one obtains

$$q = q_0 + \delta q \tag{B-6}$$

$$\underline{a}_{o} = -\left(\frac{\lim \omega L^{3}}{2c^{2}}\right) \frac{\varepsilon_{o} \sin \left(\theta - \psi_{o}\right)}{\left\{1 - \varepsilon_{o} \cos \left(\theta - \psi_{o}\right)\right\}^{3}}$$
(B-7)

$$\delta q = -\left(\frac{\mu_{\omega}L^{3}}{2c^{2}}\right) \left\{ \left(f_{1} + \epsilon_{5}f_{3}\right)\delta\epsilon - \left(f_{2} - \epsilon_{6}f_{4}\right)\epsilon_{6}\delta\psi \right\}$$

$$+\left(\frac{\mu_{L}^{3}}{c^{2}}\right) \left(f_{2}\delta\dot{\epsilon} + f_{1}f_{6}\delta\psi \right)$$
(B-8)

$$f_{1} = \frac{\sin(\theta - \psi_{o})}{\{1 - \epsilon_{o} \cos(\theta - \psi_{o})\}^{3}} ; \qquad f_{2} = \frac{\cos(\theta - \psi_{o})}{\{1 - \epsilon_{o} \cos(\theta - \psi_{o})\}^{3}}$$

$$f_{3} = \frac{3 \sin(\theta - \psi_{o}) \cos(\theta - \psi_{o})}{\{1 - \epsilon_{o} \cos(\theta - \psi_{o})\}^{4}} ; \qquad f_{4} = \frac{3 \sin^{2}(\theta - \psi_{o})}{\{1 - \epsilon_{o} \cos(\theta - \psi_{o})\}^{4}}$$
(B-9)

Global effects are obtained by integrating over projected areas through appropriate limits  $\theta_1$  and  $\theta_2$ . That is

$$(F_R, F_T) = \begin{cases} \theta_2 \\ \theta_1 \end{cases} q_0 \{\cos(\theta - \psi_0), -\sin(\theta - \psi_0)\} Rd\theta$$
 (B-10)

$$\frac{\partial (F_{\varepsilon}, F_{\psi})}{\partial e} = \left(\frac{u_{\omega} L^{3} R}{2C^{3}}\right) \int_{\theta_{1}}^{\theta_{2}} (f_{1} + \varepsilon_{0} f_{3}) \left\{\cos (\theta - \psi_{0}), \sin (\theta - \psi_{0})\right\} d\theta$$
(B-11)

$$\frac{\partial (\mathbf{F}_{\varepsilon}, \mathbf{F}_{\psi})}{\mathbf{e}_{o} \partial \psi}$$

$$= -(\frac{\mu_{o} L^{3} R}{2c^{3}}) \int_{\theta_{1}}^{\theta_{2}} (\mathbf{f}_{2} - \varepsilon_{o} \mathbf{f}_{4}) \{\cos (\theta - \psi_{o}), \sin (\theta - \psi_{o})\} d\theta \qquad (B-12)$$

$$\frac{\partial (\mathbf{F}_{\varepsilon}, \mathbf{F}_{\psi})}{\partial \mathbf{e}}$$

$$= -\left(\frac{\mu \mathbf{L}^{3} \mathbf{R}}{c^{3}}\right) \int_{\theta_{1}}^{\theta_{2}} \mathbf{f}_{2} \left\{\cos \left(\theta - \psi_{0}\right), \sin \left(\theta - \psi_{0}\right)\right\} d\theta \tag{B-13}$$

$$\frac{\partial (F_{\varepsilon}, F_{\psi})}{e_{0}\partial \psi}$$

$$= -\left(\frac{\mu L^{3}R}{c^{3}}\right) \int_{\theta_{1}}^{\theta_{2}} f_{1} \left\{\cos \left(\theta - \psi_{0}\right), \sin \left(\theta - \psi_{0}\right)\right\} d\theta$$
 (B-14)

Closed form integrals are available:

$$\int \frac{\sin(\theta - \psi_0) \cos(\theta - \psi_0) d\theta}{\left\{1 - \varepsilon_0 \cos(\theta - \psi_0)\right\}^3}$$

$$= \frac{1 - 2\varepsilon_0 \cos(\theta - \psi_0)}{2\varepsilon_0^2 \left\{1 - \varepsilon_0 \cos(\theta - \psi_0)\right\}^2} \tag{B-15}$$

$$\int \frac{\sin^2(\theta-\psi_o) d\theta}{\left\{1-\varepsilon_o \cos (\theta-\psi_o)\right\}^3}$$

$$=\frac{\chi - \sin\chi \cos\chi}{2(1-\epsilon_0^2)^{3/2}}$$
(B-16)

where

$$x = 2 \tan^{-1} \left[ \sqrt{\frac{1+\epsilon_0}{1-\epsilon_0}} \tan \left( \frac{\theta-\psi_0}{2} \right) \right]$$
 (B-17)

$$\int f_{1} \cos (\theta - \psi_{0}) d\theta = \int f_{2} \sin (\theta - \psi_{0}) d\theta$$

$$= -\frac{1}{2} \left[ \frac{\cos (\theta - \psi_{0})}{1 - \epsilon_{0} \cos (\theta - \psi_{0})} \right]^{2}$$
(B-13)

$$\int_{2}^{4} \cos (\theta - \psi_{0}) d\theta$$

$$= (1 - \epsilon_{0}^{2})^{-5/2} \{(\frac{1}{2} + \epsilon_{0}^{2}), \chi + \sin\chi(2\epsilon_{0} + \frac{1}{2}\cos\chi)\}$$
(B-19)

$$\int_{1}^{\infty} \sin (\theta - \psi_0) d\theta$$

$$= \frac{1}{2} (1 - \varepsilon_0^2)^{-3/2} \{ \chi - \sin\chi \cos\chi \}$$
(B-20)

$$\int_{f_3} \cos (\theta - \psi_0) d\theta = -\left[\frac{\cos(\theta - \psi_0)}{1 - \epsilon_0 \cos (\theta - \psi_0)}\right]^3$$
(B-21)

$$\int_{4}^{4} \cos (\theta - \psi_{o}) d\theta = \int_{3}^{4} \sin (\theta - \psi_{o}) d\theta$$

$$= (1 - \varepsilon_{o}^{2})^{-5/2} \left\{ \frac{3}{2} \varepsilon_{o} (\chi - \sin\chi \cos\chi) + \sin^{3}\chi \right\}$$
(B-22)

$$\int_{4}^{6} \sin (\theta - \psi_{o}) d\theta = \frac{\cos^{3} (\theta - \psi_{o}) - \frac{1}{\varepsilon_{o}}}{\left\{1 - \varepsilon_{o} \cos (\theta - \psi_{o})\right\}^{3}}$$
(B-23)

Consider now the full cylindrical journal bearing. Accepting the Gümbel cavitation condition and the assumption of film initiation at the maximum gap, then for the static equilibrium problem,

$$\theta_1 = -\pi + \psi_0; \qquad \theta_2 = \psi_0$$
 (B-24)

Consequently,

$$F_{R} = (\frac{\mu\omega L^{3}R}{c^{2}}) \frac{\varepsilon_{o}^{2}}{(1-\varepsilon_{o}^{2})^{2}} ; \quad F_{T} = (\frac{\mu\omega L^{3}R}{c^{2}}) \frac{\pi\varepsilon_{o}}{4(1-\varepsilon_{o}^{2})^{3/2}}$$

or

$$\overline{W} = (\pi \frac{L}{D})^2 \frac{\varepsilon_0 \sqrt{1 - \varepsilon_0^2 + (4 \frac{\varepsilon_0}{\pi})^2}}{(1 - \varepsilon_0^2)^2} \qquad \psi_0 = \tan^{-1} (\frac{\pi \sqrt{1 - \varepsilon_0^2}}{4\varepsilon_0})$$
(B-25)

represent the dimensionless static equilibrium operating parameters previously defined in Section 2.

Note that

$$\varepsilon_{o}^{\text{im}} = (\pi_{\overline{D}}^{L})^{2} \varepsilon_{o}$$

$$\lim_{\varepsilon_{0} \to 1} \quad \overline{W} = \left(\frac{\underline{L}}{D}\right)^{2} \frac{4\pi}{\left(1-\varepsilon_{0}^{2}\right)^{2}}$$

Thus

$$\lim_{\overline{W} \to 0} \varepsilon_0 = (\pi_{\overline{D}}^L)^{-2} \overline{W}$$

$$\lim_{\overline{W} \to \infty} 1 - \varepsilon_0 = 2(\frac{L}{D}) \sqrt{\frac{\pi}{\overline{W}}}$$

respectively depict near-field and far-field asymptotic behaviors.

For the perturbation problems,  $\theta_1$  would remain unchanged but  $\theta_2$  would be shifted so the sum  $q_o$  +  $\delta q$  vanishes. Therefore,

$$\theta_{2} = \theta_{20} + \delta\theta_{2}$$

$$(\frac{\partial q_{0}}{\partial \theta})_{\theta_{20}} \delta\theta_{2} + (\delta q)_{\theta_{20}} = 0$$

$$\delta\theta_{2} = -(\frac{\delta q}{\partial q_{0}/\partial \theta})_{\theta_{20}} (B-26)$$

From Eqs. (B-8 and B-9), since at  $\theta$  =  $\theta_{\mbox{20}}$  =  $\psi_{\mbox{o}}$ 

$$(f_1, f_3, f_4)_{\theta_{20}} = 0$$

$$f_2 = (1-\epsilon_0)^{-3}$$
(B-27)

 $\delta\theta_2$  is zero for perturbations with respect to  $\delta\epsilon$  and  $\delta\psi$  and needs to be evaluated only for perturbations with respect to  $\delta\psi$  and  $\delta\epsilon$ . Differentiating Eq. (B-7) with respect to  $\theta$ , one finds

$$\frac{\partial q_o}{\partial \theta} = -\left(\frac{\mu \omega L^3}{2C^2}\right) \frac{\varepsilon_o}{\left\{1 - \varepsilon_o \cos \left(\theta - \psi_o\right)\right\}^3} \left[\cos \left(\theta - \psi_o\right) - \frac{3\varepsilon_o \sin \left(\theta - \psi_o\right)}{\left\{1 - \varepsilon_o \cos \left(\theta - \psi_o\right)\right\}\right]}$$

Therefore,

$$\delta\theta_2 = \delta\psi + \frac{2}{\varepsilon_0} \delta\varepsilon \tag{B-28}$$

Accordingly, allowing for the perturbed shift of  $_{2}$ , from Eq. (B-10, one finds

$$(F_R, F_T) = -(\frac{\mu\omega L^3 R}{2c^2}) \left[ (\frac{\delta\theta_2^2}{2})^2, -(\frac{\delta\theta_2^3}{3})^3 \frac{\epsilon_0}{(1-\epsilon_0)^3} \right]$$
 (B-29)

Clearly, in comparison with terms given by Eqs. (B-11) through (B-14), these are higher order effects and can be omitted from a dynamic perturbation analysis.

Upon substituting Eq. (B-24) for the limits of integration of Eqs. (B-18) through (B-23), Eqs. B-11) through (B-14) become

$$\frac{c^2}{\mu\omega L^3} \begin{bmatrix} \delta F_{\varepsilon} \\ \delta F_{\psi} \end{bmatrix} = -\left[ \overline{2}_{\varepsilon\psi} \right] \begin{bmatrix} \delta \varepsilon \\ \varepsilon_{o} \delta \psi \end{bmatrix}$$

$$= -\left\{ \frac{\frac{2\varepsilon_{o}(1+\varepsilon_{o}^{2})}{(1-\varepsilon_{o}^{2})^{3}} + \frac{\pi(1+2\varepsilon_{o}^{2})}{2(1-\varepsilon_{o}^{2})^{5/2}} \frac{d}{\omega dt} - \frac{\pi}{4(1-\varepsilon_{o}^{2})^{3/2}} - \frac{2\varepsilon_{o}}{(1-\varepsilon_{o}^{2})^{2}} \right] \frac{d}{\omega dt}}{\left\{ \frac{-\pi(1+2\varepsilon_{o}^{2})}{4(1-\varepsilon_{o}^{2})^{5/2}} - \frac{2\varepsilon_{o}}{(1-\varepsilon_{o}^{2})^{2}} \right\} \frac{d}{\omega dt} - \frac{\varepsilon_{o}}{(1-\varepsilon_{o}^{2})^{2}} + \frac{\pi}{2(1-\varepsilon_{o}^{2})^{3/2}} \frac{d}{\omega dt} \right\}$$
(B-30)

Using the nomenclature and coordinate system defined in Section 2, one obtains

$$= \begin{cases} \cos\psi_{o} & \sin\psi_{o} \left(\frac{2\varepsilon_{o}(1+\varepsilon_{o})^{2}}{(1-\varepsilon_{o}^{2})^{3}} \frac{\pi}{4(1-\varepsilon_{o}^{2})^{3/2}} \right) \begin{cases} \cos\psi_{o} & \sin\psi_{c} \\ \sin\psi_{o} & \cos\psi_{o} \right) \left(\frac{\pi(1+2\varepsilon_{o}^{2})}{4(1-\varepsilon_{o}^{2})^{5/2}} \frac{\varepsilon_{o}}{(1-\varepsilon_{o}^{2})^{2}} \right) \begin{cases} -\sin\psi_{o} & \cos\psi_{o} \end{cases}$$

(B-31)

$$\{4\pi \left(\frac{L}{D}\right)^{2-1} \omega \left[\overline{B}\right]$$

(B-32)

Numerical results for Eqs. (B-25), (B-) and (B-32) are given in Table B-1. These results can be used to illustrate the natural motion analysis discussed in Section 2.4. At each static equilibrium condition as may be fixed by  $\varepsilon_0$  of  $\overline{\mathbb{W}}$ , the set of eight (dimensionless) dynamic perturbation coefficients can be used to calculate the stiffness and damping constants  $(\overline{\mathbb{K}}_0, (\nu/\omega) \overline{\mathbb{b}}_0)$  as functions of the frequency ratio  $(\nu/\omega)$ . At each  $(\nu/\omega)$  there are two sets of  $(\overline{\mathbb{K}}_0, (\nu/\omega) \overline{\mathbb{b}}_0)$  corresponding to the alternate signs in Eq. (20). One can define

$$\overline{m}_{o} - \overline{k}_{o}(v/\omega)^{-2}$$

which may be called the consistent (dimensionless) mass of the natural motion. Accordingly, one can further define the consistent dimensionless gravitational acceleration as

$$\overline{g}_{o} = \overline{W}/\overline{m}_{o} = (v/\omega)^{2} \overline{W}/\overline{k}_{o}$$

TABLE B-1

STATIC AND DYNAMIC CHARACTERISTICS OF SHORT JOURNAL BEARING

K-yy  0.0000E+00  1.0042E-02  1.0042E-01  1.0143E-01  2.0343E-01  2.0343E-01  2.0343E-01  3.0343E-01  3.0343E-01  3.0343E-01  4.3166E-01  4.3169E-00  5.3166E-00  5.3166E-00  6.3236E-00  6.3236E-00  6.3236E-00  6.3236E-00  6.3236E-00  6.3236E-00  6.3236E-00  6.3236E-00  6.3236E-00  6.3436E-00  6.3236E-00  6.4653E-00  6.3236E-00  6.4653E-00  6.3236E-00  6.4653E-00  6.3236E-00  6.4653E-00  6.3236E-00  6.4653E-00  6.3236E-00  6.4653E-00  6.3336E-00  6.4653E-00  6.3336E-00  6.4653E-00
8. 8849E + 000
K-yx  85.39E-01  -7.8853E-01  -7.8853E-01  -7.8853E-01  -7.8853E-01  -7.8853E-01  -7.7599E-01  -7.7789E-01  -7.7789E-01  -7.7789E-01  -7.7789E-01  -7.789E-01  -7.789E-01  -7.789E-01  -7.789E-01  -7.789E-01  -7.789E-01  -7.885E-01
8.40 0.0006+00 1.0044E-01 1.0044E-01 1.0044E-01 2.1703E-01 3.6059E-01 4.7332E-01 1.9338E-01 1.9338E-01 1.9338E-01 1.9338E-01 1.9338E-01 1.9338E-01 1.9338E-01 1.9338E-01 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1.9338E-00 1
K-X 7.8839E-01 7.9747E-01 8.1275E-01 8.3446E-01 9.4426E-01 1.12836E-01 1.12836E-01 1.12836E-01 1.2103E-00 1.3103E-00 1.3103E-00 1.4243E-00 1.5565E-00 1.5565E-00 1.5565E-00 1.7103E-00 1.9560E-00 1.9565E-00 1.9565E-00 1.9565E-00 1.9565E-00 1.9565E-00 1.9565E-00 1.9565E-00 1.9649E-01 1.3649E-01 1.3649E-01 1.3649E-01 1.3649E-01 1.3649E-01 1.3649E-01 1.3649E-01 1.3649E-01 1.3649E-01 1.3649E-01 1.3649E-01 1.3649E-01 1.3649E-01 1.3649E-01 1.3649E-01 1.3649E-01 1.3649E-01 1.3649E-01 1.3649E-01 1.3649E-01 2.3567E-02 3.39026E-02 3.39026E-01
8-xx 1. 57076 + 00 1. 57576 + 00 1. 6468 + 00 1. 6468 + 00 1. 89946 + 00 1. 98997 + 00 2. 1017 + 00 2. 1017 + 00 2. 1017 + 00 2. 1017 + 00 2. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 + 00 3. 1017 +
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### ##################################
194 5.9663E-00 1.9663E-00 1.01963E-00 1.01968E-01 1.7419E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01 1.9563E-01

Geometrical characterization of the natural orbit can be described in terms of the ratio of the minor radius to the major radius and the orientation angle of the major axis measured from the static equilibrium load vector. Tables B-2 (a) and B-2 (b) list results of natural orbit analysis performed for the short journal bearing at two static equilibrium conditions. Mode 1 and Mode 2, respectively, correspond to the lower and upper signs of Eq. (20). Symbols used in the headings of these tables are defined as follows:

ecc	static equilibrium eccentricity, ratio, $\epsilon_{o}$
freq.	ν/ω
G	dimensionless consistent gravitational acceleration, $\text{g}_{\text{o}}/(\text{C}\omega^2)$
K	dimensionless stiffness constant, $k_0 C_0^3/(\mu\omega L_0^3 R)$
N*C	dimensionless stiffness constant, $k_0 c^3/(\mu\omega L^3 R)$ dimensionless damping constant, $\nu b_0 c^3/(\mu\omega L^3 R)$
b/a	minor/major radius ratio
orient	orientation angle in degrees of major axis measured
	from load vector

Table B-2 (a) is for  $\epsilon_{\rm o}$  = 0.5. Negative damping is indicated in the range

$$0 < \frac{v}{\omega} < 0.5$$

for Mode 1 which has a positive value of minor/major radius ratio at all frequencies, indicating a forward whirling mode. A lower bound of consistent gravitational acceleration for stable operation is

$$(\frac{g_0}{C\omega^2})$$
 = 0.15  $C\omega$  lower bound

The corresponding consistent mass is often called the critical mass for instability. Mode 2 has a positive damping at all frequencies and has a negative minor/major radius ratio, indicating a backward whirl orbit. Table B-2 (b) is for  $\varepsilon_0$  = 0.8. The damping constant is positive at all frequencies. Thus, by static loading, it is possible to suppress instability.

TABLE B-2

# NATURAL ORBIT PARAMETERS OF SHORT JOURNAL BEARING

			NATURAL UKBIT PAKAMETEKS UF SI	HOK! JOU	KNAL BEAKING		
	<b>3 3 4</b>	A	<pre>&lt;&lt;&lt;&lt;&lt;&lt;&lt;&lt;&lt;&lt;&lt;&lt; and of the state of the st</pre>	>>>>>>>>>>>>>>>>>>>>>>>>>>>>>>>>>>>>>>	<pre> &lt;&lt;&lt;&lt;&lt;&lt;&lt; models = 0</pre>	<<<<<<<<	>>>> rent
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### APPENDIX C

# THE HALF SOMMERFELD SOLUTION

The classical solution for an infinitely long journal bearing is remembered by the name of the noted mathematician Sommerfeld [11]. The differential equation to be solved is

$$\frac{\mathrm{d}}{\mathrm{d}\theta} \left[ \frac{\mathrm{h}^3}{12\mu} \frac{\mathrm{d}p}{\mathrm{R}\mathrm{d}\theta} - \left( \frac{\omega R}{2} \right) \mathrm{h} \right] = 0 \tag{C-1}$$

where  $h = C (1-\epsilon \cos \theta) = CH$ . Upon integrating once,

$$\frac{\mathrm{dp}}{\mathrm{d\theta}} = \frac{6\mu\omega R^2}{c^2} \left(\frac{1}{H^2} - \frac{H^*}{H^3}\right) \tag{C-2}$$

 *  is the value of the dimensionless gap at the pressure peak. Integrating again

$$p = \frac{6\mu\omega R^2}{c^2} \left\{ \left( \int \frac{d\theta}{H^2} + H^* \int \frac{d\theta}{H^3} \right) + A \right\}$$

$$=\frac{6\mu\omega R^2}{c^2}\{(1-\epsilon^2)^{-3/2}(\chi+\epsilon\sin\chi)$$

+ H^{*} 
$$(1-\varepsilon^2)^{-5/2} \left[ (1+\frac{\varepsilon^2}{2})\chi + 2\varepsilon \sin\chi + \frac{\varepsilon^2}{4} \sin2\chi \right] + A$$

where  $\chi = 2 \tan^{-1} \left[ \sqrt{\frac{1+\epsilon}{1-\epsilon}} \right] \left( \tan \frac{\theta}{2} \right) \right]$  A is a second integration constant. Requiring p to be periodic, one finds

$$H^* = \frac{2(1-\varepsilon^2)}{2+\varepsilon^2} \tag{C-3}$$

A is set to zero since it only represents a pressure level. Thus

$$\frac{2\pi p c^{2}}{\mu \omega R^{2}}$$

$$= \frac{-12\pi \sin \chi (2-\epsilon^{2} + \epsilon \cos \chi)}{(1-\epsilon^{2})^{3/2} (2+\epsilon^{2})}$$

$$= -\frac{12\pi \epsilon (1+H) \sin \theta}{(2+\epsilon^{2}) H^{2}}$$
(C-4)

This is seen to be an odd function of  $\theta$ , being positive in the converging gap but negative in the divergent gap. Gümbel [8] suggested that the lubricant film would not sustain the negative pressure so that Eq. (C-4) is valid only in the convergent half of the circumference. This is known as the half Sommerfeld solution.

Dimensionless force components based on half Sommerfeld solution are

$$\overline{w}_{R} = \frac{\pi c^{2}}{\mu \omega R^{2}} \int_{-\pi}^{0} p \cos \theta \ d\theta$$

$$= \frac{\pi c^{2}}{\mu \omega R^{2}} \left\{ p \sin \theta \Big|_{-\pi}^{0} - \int_{-\pi}^{0} \sin \theta \ \frac{dp}{d\theta} \ d\theta \right\}$$

$$= -6\pi \int_{-\pi}^{0} \left( \frac{1}{H^{2}} - \frac{H^{*}}{H^{3}} \right) \sin \theta \ d\theta$$

$$= \frac{12\pi \epsilon^{2}}{(1-\epsilon^{2})(2+\epsilon^{2})} \tag{C-5}$$

$$\overline{W}_{T} = \frac{\pi c^{2}}{\mu \omega R^{2}} \int_{-\pi}^{0} -p \sin \theta \ d\theta$$

$$= -6\pi \int_{-\pi}^{0} \left(\frac{1}{H^{2}} - \frac{H^{*}}{H^{3}}\right) \cos \theta \ d\theta$$

$$= \frac{6\pi^{2} \varepsilon}{\sqrt{1-\varepsilon^{2}}(2+\varepsilon^{2})}$$
(C-6)

Or, the dimensionless load and the attitude angle are

$$\overline{w} = \frac{2\pi wc^{2}}{\mu\omega R^{2}LD}$$

$$= \sqrt{\overline{w}_{R}^{2} + \overline{w}_{T}^{2}}$$

$$= \frac{6\pi^{2}\varepsilon}{(1-\varepsilon^{2})(2+\varepsilon^{2})} \sqrt{1 - (1-\frac{4}{\pi}^{2})\varepsilon^{2}}$$

$$= \tan^{-1}(\frac{\pi\sqrt{1-\varepsilon^{2}}}{2\varepsilon})$$
(C-7)

### APPENDIX D

### LISTING OF RETRIEVAL FILES

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       ALFA
                      1.1250
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 8 7489D-01 9.0123D-01 9.3233D-01 9.4160D-01 9 5931D-01 9 6382D-01 9 7737D-01
 9 81300-01 9 85510-01 1 00000 00
 1 00000 00 0 00000-01 4 65530-01 9 91330-01 1.00000 00-1 50710 00 0 00000-01
 3 58860 01 7 99170-01-1 12750 00 5 22010 00-2.95270 01 6 76450-01-6 82180-01
9 7697D-01 6 0402D 00 5 9128D-01-4 7862D-01 1 8476D 00 8 7404D-01 5 4556D-01
-2 3577D-01 1.9591D 00 1.2226D 01 5 3827D-01 1 7875D-01 2 7382D 00 2 3834D 01
 5.6444D-01 5 3215D-01 5.5516D 00 6.7419D 01 6 4616D-01 1 3728D 00 1 1780D 01
  5560D 02 6 8584D-01 1.6972D 00 1 5880D 01 2 4609D 02 7 4752D-01 2 3100D 00
 2 35320 01 4 19340 92 7 70010-01 2 54620 00 2 74120 01 7 09800 02 8 20060-01
 3 14290 00 3.99820 01 5.97390 02 8 34650-01 3.32930 00 4 26760 01 1 80710 03
 8 8443D-01 4.0735D 00 6 7162D 01 5 1491D 03 9 0364D-01 4 4304D 00 9 0477D 01
 1.26940 04 9.20340-01 4 84020 00 1 36350 02-9 41180 03 1 00000 00
 0 00000-01-5 00000-01 9.00000 01 1 88700 02 1 00000 00-1 53910 00 0 00000-01
 3 71560 01 7.95170-01-1.14600 00 5 40500 00-2 96650 01 6 71620-01-6 75590-01
 1 14200 00 1 43990 00 5.86830~01-4 96030-01 1 34950 00-2.57290 00 5 33630-01
-3 4479D-01 1.0213D 00 6.0473D 00 4 9738D-01-1 3215D-01 1 9012D 00 1 1734D 01
 4 9369D-01 4.8455D-02 2 7940D 00 7.6173D 01 5.1991D-01 6 2969D-01 9 8316N 00
 1.9367D 02 5 4051D-01 9 559SD-01 1 4934D 01 6 93730 02 5 80935-01 1 75580 00
 3 6507D 01 7 4281D 02 5 9888D-01 2 1262D 00 4 33950 01 3 2007D 03 6 46300-01
 3 39670 00 1 00090 02 1 45500 03 6 62660-01 3 86280 00 1 04640 02 1 47490 04
 7 30910-01 6.66170 00 3.06490 02 3.58290 04 7.64770-01 8 41680 00 4 68720 02
 6 33200 04 7 98740-01 1.05240 01 6 97540 02-4 8:490 04 1 00000 00
0 0000D-01 5 0000D-01 1 9576D 01 6 7158D 00 1.0000D 00 1 0700D 00 0 00000-01
~1 6629D 01 1 1471D 00 8 9403D-01~2 4189D 00 1 7137D 01 1 2591D 00 7 2337D-01
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 € 41730-01-5 02500-01-2.23970 01 1 52450 00 3 31470-01-3 76170 00-4 76120 01
 1 5354D 00-9.2527D-02-7.3841D 00-1 3930D 02 1 4770D 00-1 3692D 00-2 0254D 01
-2 4728D 02 1 4331D 00-1.9887D 00-2 6769D 01-4 7387P 02 1 3560D 00-3 0502D 00
-4 1504D 01-8 5760D 02 1 3258D 00-3 4720D 00-4 9457D 01-4 2966D 02 1 2861D 00
-4 41520 00-5 7066D 01-3.3754D 03 1.2356D 06-4 7069D 00-7 2289D 01-3 8134D 03
 1.16360 00-6.03650 00-1 23960 02 9 62230 03 1 13520 00-6 40910 00-8 03880 01
-2 0158D 04 1 1110D 00-6 9212D 00-1 5323D 02 1 0577D 04 1 0000D 00
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   3800D-02 6 3687D-02 1.9457D 00 9 2651D-01 5 36686-02 3 4682D-01-2 6398D 60
  31410-01 7 24150-02 1 45980-01 1 63860 01 9 46900-6: 1 56690-01 1 67830 00
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 9 72740-01 6.24410-01 4 00070 00 1 47400 04 9 75670-01 7 89590-01 5 89350 01
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 5 57210-01 5 17830-01 5 62600 00 6 51980 01 6 37646-01 1 31546 00 1 16560 01
2 23705 02 6 77030-01 1 76050 06 1 75440 01 1 62480 02 7 39210-01 2 32460 00
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        01-5,04940 02 1,39850 00-2.57400
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是是一种,我们是一个人,我们是一个人,我们是一个人,我们是一个人,我们是一个人,我们是一个人,我们是一个人,我们是一个人,我们是一个人,我们是一个人,我们是一个人, 第二十二章 第二十二二章 第二十二章 第二十二章 第二十二章 第二十二章 第二十二章 第二十二章 第二十二

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 4.14880 01 8.80690-01 1.46240-01 2 21820 00-1 03410 01 8 85950-01 2 07400-01
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                                               CANTELL & TSITEPPE : BRITE G.
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            3 31440-01 1 09220 00 6 09250 01 5 58371-01
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1 -8660 30 0 0100 01 1 53110 07
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4 00000-01 0 00000-01 0 60000-01 0 00000-01 1 00000 (1 6 00000-0) 0 00000-0;
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 9.3277D-01 9.4381D-01 9 5386D-01 9 62740-01 8 70550-01 8 7728D-01 1 0000D 00
 1 0000D 00 0.0000D-01 3 0349D-01 9 3273D-01 1 00000 00-1 4247D 00 0 00000-01
 4 59270 01 8 59750-01-1 17290 00 4 80516 00-0 01146 01
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-1 0423D 03 7.3128D-01 2.8408D 00 3 0010D 01 2 6164D 03 7 61615-01 3 2746D 90
 5 6262D 01 8 4516D 02 7 9210D-01 3 9001D 00 6 17616 05 0 625 B 97 8 19.56-61
 4 41560 00 9 20700 01 7 40410 03 8 57260-01 5 20700 00 t 4 950 02-6 24040 01
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HB 66290-01-2.52050 01-2 14130 02 2 26970 00-1 Peter 00-7 UZITO 01-1 19945 03
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   and from a lawde fello green-like rough-like benearing beneather a gordo he is beneather.
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 4.3975D 00 1.3339D 02 1.2828D 04 8.2010D-01 5.3839D 00 2 0760D 02 3.6555D 04
8.5012D-01 6.8639D 00 3.8898D 02 2 4825D 04 8 8213D-01 8 6852D 00 4 9166D 02
-4.3885D 04 1.0000D 00
0.00000-01 0.00000-01 7.2014D 00 1 3309D 01 1 0000D 00-1 0463D 00 0 0000D-01
 3.41780 01 8.75120-01-7.49330-01 4 50560 00-2.87750 01 8 0757p-01-4 11580-01
 9.2895D-01 1.8586D CO 7.6762D-01-2 9541D-01 1 1379D 00-3 8702D 00 7 4346D-01
-2.02250-01 7.57430-01 1 7949D 00 7 2899D-01-1
                                                 3001D-01 9 1272D-01 2.1861D 00
7 22050-01-5.70320-02 1 07340 00 2 74150 00 7 20700-01 1 6.85230 00 7 23640-01 9 47210-02 1 62240 00 2 53566 01 7
                                                             58990-02
                                                                      1 24580 00
                                                            30480-01 2 01230-01
2.83430 00-9.37010-01 7 41040-01 3 161:0-01 2.79610 00 :
                                               2.7961D 00 1 5874D 02 7 5491D-01
7 04400-01 5 3651D 00 3 5741D 02
5.1054D-01 8.3396D 00-1 0514D 02 7 72300-01
7.92460-01 9.49430-01 1 42800 01 2 89890 02 8 15040-01 1 31230 00 2.03550 01
9.1745D 02 8.2894D-01 1 5375D 00 2 8766D 01 1 324:D 03 8 4285D-01 1 8322D 00
 4.2245D 01 2.2527D 03 8 57770-01 2 2079D 00 5 8969D 01 2 8246D 03 8 7378D-01
2.6591D 00 7.7630D 01 9.2060D 03 8 9076D-01 3.2623D 00 1 3089D 02 2 0629D 04
9.0890D-01 4.16570 00 2 3325D 02 1 6250D 04 9 28390-0: 5 2695D 00 3 0046D 02
-2.6819D 04 1.0000D 00
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                      1.1250
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8.4404D-01 8.7311D-01 8.9714D-01 9 :708D-01 9 3402D-0: 9 4829D-0: 9 6045D-0:
9 70670-01 9.75180-01 9 79190-01 9 82820-01 9 86060-01 9 88920-01 1 00000 00
1.0000D 00 9 0000D-01 1 0946D-01 9 4761D-01 1 0000D 00-1 1892D 00 0 0000D-01
9.09470 00 7.07380-01-8.53760-01 2 470:0 00-1 17070 01 5 69850-01-5 93660-01
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-3 1927D-01 2 1766D 00 2 8411D 01 4 4360D-01-1 2450D-01 3 9755D 00 8 9768D 01
4.4369D-01 1.6139D-01 8.1934D 00 7 126:0 01 4 5576D-0: 5 0246D-0: 1 0758D 01
  7624D 02 4 7523D-01 8 8965D-01 1 5881D 01 4 5642D 02 5 0224D-01
  68470 01 2.16530 02 5 35840-01 1 98130 00 7 11650 01 1
                                                          29510 03
                                                                   5 74960-01
2,69580 00 5 31210 01 5.80110 02 6 19100-01 3 51250 00 6 13960 01
                                                                   3 50:10 03
  67400-01 4.51790 00 1 03970 02 7 93040 02 7 19:40-01 5 62:90 00
                                                                   1 12070 02
  7179D 04 7.4606D-01 6.4042D 00 2 3470D 02-9 0961D 03 7 7350D-01 7 2713D 00
1.9826D 02 2.6240D 04 8.0137D-01 8 1625D 00 2 9340D 02 2 8973D 04 8 2957D-01
9 26710 00 3 87410 02 6.92980 04 8 57960-01 1 06610 01 5 85860 02-5 28940 04
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0.0000P-01 1.0000D 00 0.0000D-01 0 0000D-01 0 0000P-01 : 0000D 00 0 0000P-01
0 0000b-01 0.0000b-01 1.0000b 00 0 0000b-01 0 0000b-01 0 0000b-01 1 0000b 00
0.0000D-01 0.0000D-01 0.0000D-01 1 0000D 00 0 0000D-0: 0 0000D-01 0 0000D-01
1.0000D 00 0 0000D-01 0 0000D-01 0 0000L-01 : 0000D G0 0 0000D-01 0 0000D-01
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0.0000D-01 0.0000D-01 1 0000D 00 0 0000D-01 0 0000E-0; 0 0000D-01 1 0000D 00
0.0000p-01 0.0000p-01 0.0000p-01 1 0000b 00 0 0000p-01 0 0000p-01 0 0000p-01
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0.0000D-01 5.0000D-01 2.9506D 01 6 8439D 00 1 0000D 00 1 3631D 00 0 0000D-01
-3 0310D 00 1.3601D 00 1.2513D 00-8 2322D-01 4 3279D 00 1 6008D 00 1 1733D 00
4 6703D-02-1.0624D 01 1.7555D 00 1 0825D 00-1 3896D 00-3 4546D 01-1 8431D 0D
  1773D-01-4.4970D 00-8.7598D 01 1 982:0 00 3 5738D-0:-: 0044D-01-1 50570 02
1.88520 08~2.8074D~01~1.7119D 01~2 85775 02 1 86180 00~1 0820D 00~2 74050 01
-4 5041D 02 1 8169D 00-2.0689D 00-4 0490D 01-4 8510D 02 1 7544D 00-3 1819D 00
-5 2154D 01-1.1717D 03 1 6790D 00-4.4548D 00-7 5517D 01-8 1817D 02 1 5920D 00
-5.8526D 00-8.9388D 01-2.4484D 03 1 4982D 00-7 3768D 00-1 2431D 02-3 8169D 03
1 3982D 00-9 1707D 00-1 7073D 02 1 5152D 03 1 2958D 00-1 0836D 01-1 5524D 02
-1 3875D 05 1 2432D 00-1 2949D 01-7 8125D 02 7 8194D 05 1 1935D 00-9 7572D 00
2 37490 03-2 00200 06 1 15790 00-1 43040 01-4 80000 03 2 38230 06 1 09930 00
-1 7608D 01 2.8468D 03-4 3413D 05 1 0509D 00-1 1236D 01 1 6006D 03-1 4478D 05
1 00000 00
0.0000D-01 0.0000D-01 4.0900D 00 3 1590D 00 1 060DD 00 2 9798D-01 0 0000D-01
-1.3835D 00 1.0763D 00 2 4695D-01-3 7574D-01 2 4476D 00 1 1217D 00 2 2087D-01
1 16220-01 7.81130-01 1.15290 00 2 43720-01 2 2:630-0: 5 48080-0: 1 17580 00
  65890-01 2.71130-01 5 31580 00 1 19340 00 2 977:0-01 6 07720-01-7 65860 00
1 20770 00 3 13810-01 2 47870-01 2 05570 01 1 2:040 00 3 36050-0: 9 87810-01
-3 6989D 01 1.2294D 00 3 4914D-01-8 742:D-02 5 5736D 0: : 2379D 00 3 63:2D-0:
1.2517D 00-9.8980D 01 1 2452D 00 3 6841D-91-7 2194D-01 4 8896D 01 1 2514D 00
3 6319D-01 1.0703D-01-4 5316D 02 1 2564D 00 3 1862D-01-6 3570D 00 6 3637D 02
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1 2600D · 00 2.8836D - 01 1 3812D 00 5 9161D 02 1 2631D 00 3 3337D - 01 7 4271D 00
-7.4<mark>293D 03 1.2646D 00 2 9</mark>126D-01-2 6093D 01-3 3363D 04 : 2652D 00-8 0892D-02
-1 5973D 02-1.5572D 04 1.2637D 00-7.6232D-01-2.16180 02-5 9671D 05 1 2567D 00
-4.6051D 00-2.1524D 03-1.6462D 05 1 2340D 00-1.1444D 01-2 6258D 03 2.3689D 05
1.00000 00
0.0000D-01 5.0000D-01 2.6704D-01 3 3247D-02 1 00000 00 1 4422D 00 0 0000D-01
1 1735D 00 1.3956D 00 1 4855D 00 3 10720-01 1 0699D 01 1 7151D 00 1 7657D 00
  4693D 00-9.0295D 00 1.9727D 00 2 0170D 00 1 2485D 00 7 8634D 09 2 1601D 00
2 1611D 00 1.9558D 00-2 2640D 02 2 2913D 00 1 83:1D 00-1 2380D 01-1 1793D 02
2.3616D 00 1.1192D 00-1.7921D 01-9 8381D 02 2.3826D 00-1 6312D-01-5 3352D 01
-1.51210 03 2.34920 00-2.35230 00-9 72880 01-4 34800 02 2 26360 00-4 81510 00
-1.0773D 02-4.6639D 03 2.1400D 00-7 8904D 00-2 0673D 02-4 1316D 03 1 9740D 00
-1 1887D 01-2.7077D 02-3.1890D 03 1 7754D 00-1 6674D 01-3 1626D 02-1 6264D 04
   5516D 00-2 1122D 01-5.1404D 02 2 95890 04 1 3:42D 00-2 483:D 0:-2.1106D 02
  7990D 05 1,1973D 00-2 7616D 01-1 02335 03 4 40560 05 1 0832D 00-2 81810 01
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-1.8985D 01 3 9913D 03 2.9846D 05 8 7196D-01-6 3313D 00 4 8459D 03-4 3751D 05
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0 0000D-01 1.5000D 00 1 2414D 01 2 7039D 00 : 0600D 00-: 3854D-0; 0 0000D-01
-2.4277D 00 9.5427D-01-2.2808D-01-6 5937D-01 1 2688D 00 8 9682D-01-3 3498D-01
-4 0433D-01 1.1098D 00 8.4829D-01-3 7950D-01-2 5430D-01 1 1452D 01 8 1452D-01
-3.5605D-01 7.7576D-01 1 6451D 01 7 9423D-01-2.7395D-01 1 8174D 00 1 0240D 02
  8513D-01-7.5531D-02 6.6286D 00-6 3807D 00 7 8666D-0. 1 5892D-01 6 3989D 00
-8.37700 01 7.93640-01 3 0954D-01 3 9639D 00 6 0607D 02 8 0362D-01 5 7969D-01
1 85250 01-1,27660 03 8,17170-01 6 95230-01-6 94980 00 2 3021D 03 8 2983D-01
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8 56170-01 1.1644D 00 5.5639D 01-3 0:03D 04 8 6562D-01 1 6111D-01-2 5199D 02
1.59050 05 8 66220-01 6.43060-01 4 65630 02-1 48090 05 8 70940-01 ; 32010 00
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3.0135D 00 8 5509D 02 6.2393D 04 8 9440D-01 5 7179D 00 1 037PD 03-9 7330D 04
 1 00000 00
 0 0000D-01 1.0000D 00 1 6909D 01 8 7990D 00 1 0000D 00-7 050:D-0: 0 0000D-01
2 49560 00 8.16850-01-6 12970-01 5 77795-01-4 35090 00 7 0:450-0:-5 64636-0:
   9675D-01 1.1554D 00 6.2379D-01-5 3067D-01-4 0548D-02 4 7981D 00 5 7:98D-01
-5.6490D-01 3 9103D-01 1 1850D 01 5 3749D-01-5 1639D-01 1 141DD 00 5 7893D 01
5 15490-01-3 98860-01 3.86150 00 3 71660 01 5 03930-01-2 35790-01 5 19920 00
 4.4784D 01 4.9945D-01-6 5738D-02 6 501CD 00 3 94040 C2 5 0066D-01
                                                                   2 04178-05
   5988D 01-3.4249D 02 5 0745D-01 4 5448D-01 9 1787D GO 1 4385D 03
                                                                     17640-01
                                             1 04620 00-2 12090 00
  1617D-01 3 3529D 01-2,4992D 03 5 31488-01
                                                                   7 40026 03
 5 4620D-01 1.5620D 00 8.7939D 01-5 9338D 03 5 65696-01 2 1569D 06 2 7300D 01
 9 9012D 04 5.77190-01 3.2817D 00 4 7402D 02-2 5086D 04 5 9381D-0: 4 9792D 00
  7354D 02 1,1004D 05 6 1525D-01 7 0567D 00 7 7248D 02 3 6352D 08 6 4429D-01
   1477D 01 1.9520D 03 2 76295 05 0 06240-01 1 3200D 01 3 7432D 03-2 47676 05
  00000 00
 0 0000D-01 0 0000D-01 0.0000D-01 0 0000D-01 1 0000D 00 0 0 000DD-01 0 0000D-01
 0.0000D-01 1 C000D 00 0.0000D-01 0 000D-01 0 0000D-0: : 0000D 00 0 0000D-0:
  00000-01 0 00000-01 1.00000 to 0 00000-01 0 00000-01 0 00000-01 1 00000 00
   00000-01 0.00000-01 0.00000-01 1.00000 00 0.0000-01 0.00000-01 0.00000-01
   00000 00 0.00000-01 0 00000-01 0 00005-01 1 00006 00 0 00000-01 0 00000-01
   00000-01 1.00000 90 0 00000-01 0 00000-01 0 00000-01 0 00000-01
   00000-01 0 00000-01 1 00000 00 0 00000-01 . 06000-6: 6 00000-0: ; 00000 00
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                                                        135 1665
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                                              06.701-6
                                                       1 400 0 00 0 00000-01
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                                  06696 90 0 06601 +01 0 16164 HT
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                                              00001 00 0 00000-01 0 00000-01
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 -00000-01 0 00000-01 1.0000D 00 0 0000-01 0 00000-01 0 00100-01
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 67000 Pt 6 83410-01-7 55020-01 4 93580 00-0 30560 01 5 92670-01-0 09020-01
 80980-62 9 60875 00 5 57765-51-2 74070-01 1 20095 00 2 09755 00 5 37435-61
 16390-01 1 42460 00 1 97350 01 5 33700-01 1 13760-02 2 61690 00 4 77460 01
 37950-01 1.86760-01 4 85470 00 6 036 D 01 5 48100-01 6 00190-01 7 02720 00
 03920 02 5 63320-01 6 48760-01 1 00480,01 3 08660 02 5 82520-01
 7464D 01
          -1 81250 02 6 0576D-0.
                                1 38250 00 2 .6780 01 1 00070 03 6 32750-01
          3 89240 01 7 44440 02 6 63760-01 2.56000 00 4 94430 01 3 78690 03
 .87136 00
          3 75240 00 9 06270 01
                                2 16050 93 7 36210-61 4 30180 30 1 12800 02
 28840-01
                                                      ••
          7
            59610-61 5 1884D DC 2 40275 02-2 44696 03
                                                        10-08-513
 82549 04
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          2 $4140 04 8 06260-01 7 15260 00 3 33510 01 5 75750 00 8 31540-01 5 07370 02 6 49260 04 8 59260-01 1 02370 01 6 93290 02-6 25940 04
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1.00000 00
 0 00000-01 0 00000-01 1 69130 01 7 01670 01 1 30.01 03.5 14260-01 0 00000-01 8 35500 00 8 06750-61-5 001000-01 2 20020 00-1 00121 01 7 10100-01-2 695/0 0 8 47760-02 4.52130 00 7.02350-01-2 16730-01 0 30.00-01 7 71600 00 6 86.20-01 39100-01 1 63030 00 1 23060 01 6 73700-01-4 /1771-01 0 605/0 00 3 81700 01
8.4776D-02 4.5213D 00 7.0235D-01-2 16739-01 0
-1 3910D-01 1 6303D 00 1 2306D 01 6 7433D-5144
                                                                                      28691-01
38701 01
58211 (1
  6 8024D-01 7.7959D-02 3 6031D 00 3 59270 01 6
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7 42320-01
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 8 8254D 01 6.94976-61 4 17310-01
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-70 0 9
-2000 00
 1 2615D 01 1.5072D 02 7.2385D-01 9 4.725-01 1 3145D 00 2 8353D 01 4.4595D 02 7 64236-01
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                                                                                                                            1 52786 87
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     31450 00 2 83530 01 4,45950 02
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    18900D-01 2 37430 00 6 5580D C1 . 16506 02 8 16761-C1 1089D 04 8 32020-01 3 6695D 00 1 7264D 02-5 61511 (3
 7.8900D-01
                                                                                                       3 306 31. 9.
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     5014D 02 2.32840 04 8 6488D-01 5 02630 00 2 41116 0. 3 .44.D 04 8 8264D-01 9746D 00 3 4253D 02 4 6872D 04 9 03355-01 1 16775 00 4 7676D 02-4 30645 04
  5 97460 60 3 42550 02 4 68720 04 9 0:3:6-9)
  1 00000 00
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## APPENDIX E

## LISTING OF SOURCE PROGRAM

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NED PRO DETA
£ $18566 AFEE15
                      normonett veve for afferb effor than a hourst inferi
      461181
      IMPLICIT REALAS (A-H,0-Z)
      INTECEP & 2 FN3 16), FH2(16), FH1(16)
      LOGICAL ANS
      DINEBERGION MAND ANTERN TONES AND TEACHER OF THE PURCH TO A CO. THE STAN ZERVE. TH
      PENERETRA ZZC46 NOCK BOARD (EGIT CO BRANC) PROPICE ABOR Z COPRESSO -
      GENERALIGN NACHRES ST.
      8 M 1 = 1
      K#3=2
      887=3
      医鞭羊氏鼠疫毒素
      हर्म हो। १८७५ के
      17:11年3日日
      HHS= FALSE
      APPROPRIATE ENTER DATA TABLE NAMETER APPREACHAIGANE/FREILNING)
      IF(ERS) GD 19 1000
      WE: TE-1 1691)
1661 FORMAT' JOH ERROR IN OPENING DATA TABLE -
      CALL EXIT
       PEAD IN DATA TABLE INDEX AND AFTICK ARPA
1000 ANS=Runtiactions:
      READ(KT 1002)(FN2(I), I=1, 16), NN, AL9, ALF
      F0F567(20%,1692/10%,15/18%,F10 4/18%,F10 4)
1002
      REAL(81) 1503 (Cacl) / I+1 / He+
1007
      FORMAT(7011 4)
      特別ニガス・サ
      P4=04T4#(1 00-00)
      F2=2 0*F4
      ANS-OPNPSACIENTER OUTPUT FILE HAMEILES ASBRITHASSAME, FRI, LH, KWI)
      IF(ANS) GO TO 1005
      WRITE: 1, 1004)
      FORMATCION FROOR IN OPENING OUTPUT FILEY
1004
      CALL EDAT
1005
      WRITE:: 0.1006) (FR2(I - I=1.16) NH.ALP.ALF
1006
     FORMAT(19H BETRIEVAL FILE NO .18.16A2/18H
                                                         FILE SIZE =, IS/
                                   =,F10.4/7%,48ALFA,6%,18=,F10.4/18.)
                        L.'0
             1 8 H
      KPAGE=5
Û
Ũ
      SELECT DATA EXTRACTION MODE
C
101
      WRITE(1,1010)
     FORMAT(/41HSELECT:
                            (1) PREPARE DATA FOR RSVP INPUT/10%,
     +21H(2) DIMENSIONAL TABLE/10X,23H(3) DIMENSIGNLESS TABLE/10X,
     +8H(4) QUIT)
      READ(1,*)NSEL
      IF (NSEL.EQ.4) CALL EXIT
      WRITE(1,1011)
1011 FORMAT(/28HSELECT: (1) SYNCH FREQ, OR/10%.
     +16H(2) ASYNCH FREQ?)
```

```
READ (1,*) NSYH
      IF (NSYN 50.2) GO TO 1020
      LFREQ = 1
      FRRAT = 1.0
      GO TO 1030
1020
      WRITE(1,1021)
      FORMAT(5%, 28HENTER NUMBER OF FREQ GROUPS?)
1021
      READ (1,*) LFREQ
1030
      GO TO (2100,2100,2200,102), NSEL
      GO TO 101
102
      CALL EXIT
ε
C
      INPUT BEARING DESIGN DATA
C
2100
     WRITE(1,2101)
      FORMAT(25HENTER BEARING LENGTH (IN))
2101
      READ(1,*) ALEN
      WRITE(1,2102)
      FORMAT(27HENTER BEHRING DIAMETER (IN))
2102
      READ(1,*)ADIA
      WRITE(1,2103)
2103
      FORMAT(28HENTER BEARING CLEARANCE (IN))
      READ(1,*) ACLE
      WRITE(1,2104)
      FORMAT(29HENTER VISCOSITY (CENTI-PGISE):
2104
      READ(1,+) ACPO
      WRITE(1,2105)
      FORMAT(24HENTER BEARING LOAD (LBS))
      READ(1,*) ALDA
      IF(NSEL.EQ.2) GO TO 2166
      ANS=OPNF#A('ENTER BEARING COEFS FILE NAME', 20/##WRIT+455AMF,
     +FN3,LN,KW2)
      IF(ANS) GO TO 2106
      WRITE(1,2131)
     FORMAT(36H ERROR IN OPENING BEAPING COEFS FILE)
      CALL EXIT
C
C
     SPEED INDEPENDENT SCALING CONSTANTS
    COR = 2.0+ACLE/ADIA
      AL8=ALEN/ADIA
      AREA=ALEN*ADIA
      REYN=1.450377440-07*ACP0
      ALDAD=REYN+AREA
      BHP=P4/1.1820+07% ALCAS*#Dla*ADIA/4CLE
      ALDAD=ALDAD/COR/COR
      BGPH=AREA*ACLE/2.710+02
      IF(AL8 EQ AL9) GO TO 2107
      WRITE(1,1040) ALP.ALS
     FORMATY/31HSELECT: (1) USE DEFAULT Life = Fig. 4/
     +10X,25H(2) ADJUST DATA FOR LID =,Fid H-1H /
      READ(1, *)NAL
      IF(NAL, E0.1) GO TO 2107
      WFAC=SHORT(AL9,AL8,ALF)
```

```
GO TO 2108
2107
      WFAC=1.0
      AL8=AL9
2108
      ALOAD=ALOAD/WFAC
      BLOAD=ALOAD/60.0
      BSTIF=BLOAD/ACLE
      BDAMP=ALOAD/ACLE/P4/8.0
      BLOA=ALOA/BLOAD
      WRITE(1,2109)
2109 FORMATC/47HSELECT:
                           (1) LOAD VECTOR IN VERTICAL PLANE, OR/10%,
     +25H(2) INCLINED LOAD VECTOR?)
      READ (1,*) NVEC
      IF (NVEC.EQ.1) GO TO 2135
      WRITE(1,2132)
2132
      FORMAT(5%,35HENTER LOAD INCLINATION ANGLE (DEG)?)
      READ (1,*) WANG
      WRAD = WANG*P2/9.00+01
      CC = DEOS(WRAD)
      98 = DSIN(URAD)
      ROY(1) = CC*CC
      POT(2) = CC*S9
      90T(3) = SS*SS
2135
     CALL BASE(NBA, BMA, BST, BDA)
C
Ü
      SPEED LOOP FOR ARCTAN PARAMETER
C
      HMAX=11
      IF(NSEL.EQ.2) NNAX=51
2110
     WPITE (1,2111)
     FORMAT(24HENTER LOWEST SPEED (RPM))
2:11
      REAS(1,+) RPM1
      WPITE(1.2112)
    FORMAT(25HENTER HIGHEST SPEED (RPM))
2112
      READ(1,*) RPM2
      WRITE(1,2113) HMAX
2113
      FORMAT(43HENTER NUMBER OF SPEED POINTS (NOT MORE THAM, 13, 1H))
      READ(1,*) HRPM
      IF(NRPH GT.1) GO TO 2120
      SPRAT=1 0
      GO TO 2130
2120 XP=1 0/(NRPM-1)
      SPRAT=(RPM2/RPM1)**XP
2130 RPH=RPH1
      HIARE=HRPH
      WRITE(XW.2141) ALEN,ADIA,ACLE,ACPO.ALDA.RPM1.PPM2.NRPM
C141 FORMAT(2M, 6HL (IN).3M, 8HDIA (IN).3M, 6HC (IN), 3M, 9HVIEC (CP).2M,
     +78% (LBS)-7%-17HSPEED RANGE (RPM-/2/1%.F8.4-1%)-F9 6-1PD11 3-
     +OPF9 2.2X/F9 2/4H TO /F9.2,15,4H PTS/1H >
      KPAGE = KPAGE+3
      49 TO 2300
2200
     WRITE(1.2201) AL9
     FORMAT/ 214H DEFAULT L/D = .F10 4/27H ENTER DESISED L/D OR O TO .
2201
     *14HACCEPT GEFAULT)
      £En0(1:*) AL8
```

```
IF(AL8, EQ. 0) GO TO 2202
      WEAC=SHORT(AL9, AL8, ALF)
      GO TO 2203
2202
      WFAC=1.0
      AL8 = AL9
2203
      WRITE(1,2204)
2204
      FORMAT(27HENTER LOWEST LOAD PARAMETER)
      READ(1,*) W1
      451=81+WFAC
      WRITE(1,2205)
2205
      FORMAT(28HENTER HIGHEST LOAD PARAMETER)
      READ(1,*) U2
      WW2=W2*WFAC
      WRITE(1,2206)
      FORMAT(46HENTER BUMBER OF DATA POINTS HOT MORE THAN 5177
2206
      READ(, +) NTAB
      IF(P)A8.GT.1) GO TO 2207
      SPSAT = 1.0
      GO TO 2208
      XP = 1 0/(NTAB-1)
2207
      SPRAT=(UU2/UU1)**XP
2208
      186=88
      NEA = 1
      HVEC = 1
2300
      00 2330 TTAB=1, NTAB
      IF(NSEL.EQ.3) GO TO 2310
      WW=BLOAZRPM
      XX(ITAB)=DATAN(W0)/P2
2310
      WO(ITAB)≃WW
      IF(NSEL.EQ 3) GO TO 2320
      XG(ITAB)=RPM
      IF (ITAB EQ.NTAB) GO TO 2330
      RPM=RPM*SPRAT
      GO TO 2330
2320
      XOCITAB)=WW/WFAC
      IF (ITAB, EQ, NTAB) GO TO 2330
      WW=WW*SPRAT
2330
      CONTINUE
C
      RETRIEVAL LOOP
C
      URITE(KW,2331)
2331
      FORMAT(////)
      KPAGE=KPAGE+4
      IF(AL8.EQ.AL9) GO TO 2340
      WRITE(KW,2333) AL8
      FORMAT(37H LOAD PARAMETER IS ADJUSTED FOR L/6 = F10.4/1H )
2333
      KPAGE=KPAGE+2
2340
      IF(NSEL.EQ.3) GO TO 2350
      WRITE(KW, 2341)
      FORMAT(5x,3HRPM,10x,7HECC(IN),5x,9HATT ANGLE,3x,9HFRIST HP //
     +4X, 10HQ-REQ(GPM), 3X, 11HQ-LOST(GPM))
      GO TO 2360
2350 WRITE(KW, 2351)
```

```
FORMAT(5X,4HLOAD,8X,42HECC RATIO ATT SHGAE
                                                        FRICTion
                                                                       BEQ.
     +19H FL0H
                LOST FLOW,
2360 KFAGE=KPAGE+1
Ç
      STATIC PARAMETERS
0
      00 2389 Li=1.5
      READ(KT. 2371) 81,92, An BB: ((22)1), (Y1(N,1), K+1,3), (1+1) HM ((22)) HM;
     FORMST(7011.4)
      ARCTAN LOOP
      DO 2376 ITAB=1. HTAB
      XXX=XX(ITAB)
      HI=NURERE(XXX,X,NK)
      HPLACE: ITAB)=N1
      5X=XXX-X/N1)
      Z=SPL(ZZ,Y1,DX,N1)
      www.WerlTaB)
      Z1=FEFER (AA, BB, 31 - S2 - 99)
      7'=2+21
      IF(HSEL EQ.3) GO TO 2375
      RPM=XO(ITAB)
      GO TO (2372,2375,2373,2374,2374).L1
2372
      YY=YY*ACLE
      GO TO 2375
2373
      YY=YY+BHP+RFM+RFM
      GD TO 2375
2374
      YY=YY*8GPM*RPM
2375
      YO(LI,ITAB)=YY
2376
      CONTINUE
2380
      CCATINUE
      00 2400 ITAB=1,NTAB
      WRITE(KW,2381) X0(ITAB),(Y8(L1,ITAB),L1=1,5)
2381
      FORMAT(6(1X,1PD12.4))
      KPAGE=KPAGE+1
      IF(KPAGE.LT.61.OR.ITAB EQ.NTAB) GO TO 2400
      KPAGE=1
      WRITE(KW,2382)
2382 FORMAT(1H1)
      IF(NSEL.EQ.3) GO TO 2390
      WRITE(KW, 2341)
      GD TO 2400
2390
     WRITE(KW, 2351)
2490
     CONTINUE
C
C
      DYNAMIC PARAMETERS
C
      IF(AL8, EQ. AL9) GO TO 2420
      WRITE(KW.2412) AL8
2412
     FORMAT(////48H LOAD AND DYNAMIC COEFFICIENTS ARE ADJUSTED FOR ,
     +5HL/D =,F8.4/1H >
      KPAGE = KPAGE+6
      IF (KPAGE.GT.60) KPAGE = KPAGE-60
```

```
GO TO 2415
2420
      WRITE(KW, 2331)
      KPAGE = KPAGE+4
      IF (KPAGE.GT.60) KPAGE = KPAGE-60
      IF (NSYN.EQ.2) GD TO 2445
2415
      IF (NSEL.EQ.3) GO TO 2430
      WRITE(KW,2421)
      FORMAT(3X,3HRPM,7X,4HK-XX,6X,4HB-XX,6X,4HK-XY,6X,4HB-X/,6X,4HK-XX,
     +6X,4H8-YX,6X,4HK-YY,6X,4HB-YY)
      GB TO 2440
2430
      WRITE(KW.2431)
      FORMAT(2X,4HLOAD,7X,4HK-XX,6X,4HB-XX,6X,4HK-XY,6X,4HB-XY,6X,
2431
     +4HK-YX,6X,4HB-YX,6X,4HK-YY,6X,4HB-YY)
2440
      KPAGE = KPAGE+1
2445
      DO 2480 L1=6,13
      L=L1-5
      L2=L/2
      LL=2*L2
      REAB(KT.2371) $1,82,AH,BB.((ZZ(1),(\1)).1().1().7:1.3)().1=1.88 ( ZZ()84)
C
C
      ARCTAN LGOP
C
      DO 2480 ITAB=1, NTAB
      XXX=XX(ITAB)
      NI=NPLACE(ITAB)
      DX=XXX-X(N1)
      Z=SPL(ZZ,Y1,DX,H1)
      WW=WO(ITAB)
      Z1=REFER(AA, BB, S1, S2, UV)
      YY=2*21
      1F(NSEL.EQ.3) GO TO 2460
      IF(LL.EQ.L) GO TO 2450
      YY=YY*8STIF*X0(ITAB)
      G9 T0 2470
2450
      YY=YY*BDAMP
      GO TO 2470
2460
      YY=YY/8FAC
2470
      YO(L, ITAB)=YY
      IF(L NE.6) GO TO 2480
      YY=(YY+YO(4,ITAS))/2.0
      YO(L, ITAB)=YY
      Y0(4, [TAB)=YY
2480
      CONTINUE
      00 2490 ITAB=1.HTAB
      IF (NS/N EQ 1) INDX = ITHS
      00 3000 L=1.8
      Y(L \cdot ITAB) = YO(L \cdot ITAB)
3000
      CONTINUE
      IF (NYEC EQ 2) CALL TILT(Y.ROT :TAB!
      00 6030 KFF=1,LFFEQ
      IF (NSYN EQ 27 G5 TO 5010
      HFRE = 1
      IF (NSEL EQ.3) GO TO 5000
      FF = X0(ITHB)
```

```
GO TO 5060
      FF = 1.0
5000
      G0 T0 5060
 5010 [F(KSEL_E0.3) GO in 4010
      WRITE(EW, 4001) XO(ITAB)
 4001 FCPMAT(1H1,6H RPM = ,1PD10 3)
      GO TO 4020
 4010 WRITE(KW.4002) X0(ITAB)
 4902 FORMAT(1H1,7H LOAD =,1P010.3)
4020
      GRITE(KU,5011)
5011
      FORMAT(2X,4HFREQ,7X,4HK-XX,6X,4HB-XX,6X,4HK-XY,6X,4HB-XY,6X,
     +4HK-YX.6%,4HB-YX,6X,4HK-YY.3%,4HB-YY)
      KPAGE=1
      IF (HBEL.EQ 3) GO TO 5015
      WRITE(1,5012) KFF,XO(ITAB)
      FORMATISK, 35HENTER FREQ DATA (HZ) FOR FREQ GROUP, 15, 3H AT, 1PD12.4,
     +4H RPM/10%,12HLOWEST FREQ?)
      GO TO 5020
      WRITE(1,50)6) KFF, XC(ITAB)
      FORMATY 5X-30HENTER FREQ DATA FOR FREQ GROUP, 15.9H AT LOAD
5916
     +11HPARAMETER =,1PD12.4/10X,18HLOWEST FRED RATIO?)
5020 PEAD (1,*) FREQ1
      IF (NSEL.E0.3) GO TO 5021
      FF = 6.00+01*FRE01
      GO TO 5022
5021
      FF = FREQ1
5022
      IF (MSEL.NE 3) WRITE(1,5023)
5023
      FORMAT(10X,13HHIGHEST FRED?)
      IF (HSEL EQ 3) WRITE(1,50236)
50230 FORMAT/10X.19HHIGHEST FREG RATIO?/
      RE40 (1,+) FRE02
      WRITE(1,5024)
      FORMAT(10%,22HNUMBER OF FREG POINTS?)
5024
      READ (1/4) HERE
      IF (NSEL EQ 3) GO TO 5040
      IF (NFRE LE 1) GO TO 5030
      XP = 1 0/(HFRE-1)
      FFRAT = (FRE02/FREQ1)+*XP
      90 TO 5060
5030
      FFRAT = 1.0
      G9 TO 5060
5040
      IF (NFRE LE.1) GO TO 5050
      DERE = (FRER2-FREQ1)/\harmonicsert)
      G0 TO 5080
5050
      0.0 = 3930
5060
      00 5200 IFRE=1, NFRE
      IF (NSYN EQ 2) INOX = TERE
      FRQ(INDX) = FF
      90 5070 L=1.8
      ZEP(L, \{HDX\}) = Y(L, \{TAB\})
5070
      CONTINUE
      IF (MBA, EQ 1 AND MSYN EG.2 AND IFFE GT 1 - GO TO SIGS
      IF (N8a.EQ 1) GC TO 5089
      CALL GOAT (FF.ZEP, INDX. BHA. 85T 804)
```

```
IF (NSEL ED 37 GD TO 5095
      IF (HS:N EQ.1) GO TO 5090
      GG = FF/6.00+01
      GO TO 5100
      1F (HSYN EQ 2) GO TO 5090
5985
      GG = X9(ITAB)
      60 10 5100
5690
      GS = FF
      UPITE(18 5101) 66,426F(1,150X) LTL 1
5100
      KPHGE=FPHGE+3
      FORMAT(199018 3)
5101
      IF THREE ER 19 WESTERNE STRATTERS, CHR. C.
      FORMAT: 198010.7 :
5162
      IF (MBA, 4E 1 OF 8879 HE 21 40 ft 51/4
5:03
      IF KIFRE EG. HESE GO TO 5105
      GO TO 5105
                         46 19 5156
        CIPRE EO HERE
5104
      IF (MSEL EQ 3) GT TO TILL
5:05
         - FREFFRET
      FF
      90 TO 5150
      FF = FF+DFPE
5110
      IF (NSYN EQ 2 AND IFRE EQ HERE
                                        1, 1)
5150
      IF (KPEGE LT.61) GO TO 5290
      WRITE(+8,2382)
      IFCHS78 EQ 17 GO 70 5152
      WRITE(18 5011)
5151
      GO TO 5190
      IF(HSEL EQ.3) GO TO 5153
5:52
      GRITE(FV.2421)
      60 70 5190
      URITE(E4.2431)
5153
5190
      KPAGE = 1
5200
      CONTINUE
      IF (MSYN EQ.1) GO TO 6036
      IF (KPAGE, LT, 57) GO TO 2491
      URITE(KW.2382)
      KF4GE = 0
      GO TO 2492
      WRITE(K4,2331)
2431
      KPAGE - KPAGE+4
      IF (MSEL EQ.3) GO TO 2493
2492
      FF = 6 00+01*FRE01
      GO TO 2494
2493
      FF = FREQ1
      URITE(LV,6001)
2494
      FORMATO37H ASYNCHRONOUS BEARING CHARACTERISTICS (12% ) 45 14
6001
      +10HMINOR MODE,14(1H*),2X-14(1H*),10HMsJOS H50E-14(1H*)/
      +61H FREQUENCY STIFFHESS OR DAKP
                                           ELLIPTIC
                                                       SKIENI
                                 ORIENT:
     +28H CR DAMP ELLIPTIC
       KPAGE = KPAGE+3
      00 6020 IFRE=1, NFRE
       FF = FRQ(IFRE)
       CALL STAB(ZEP, FF, HSEL, HSYN IFRE, FN, KPAGE)
6029
       CONTINUE
```

THE PARTY OF THE P

```
IF (KPAGE.GE.5?) GO TO 6025
      WRITE(KW,2331)
      KPAGE = KPAGE+4
      GO TO 6030
6025
      URITE(ku,2382)
      KPAGE = 0
6030
      CONTINUE
      IF (NSYN.EQ 1) GO TO 2490
      IF (KPAGE LT.61 OR.ITAB-EQ.KTAB) GO TO 2490
      WRITE (KW.2382)
      IF (NSEL EQ.3) GO TO 6031
      WRITE (KW, 2421)
      GO TO 6032
6031
      WRITE(kW,2431)
6032
      KPAGE = KPAGE+1
2490
      CONTINUE
      IF (NSYN.EQ.2) GO TO 7000
      IF (KPAGE LT.57) GO TO 6035
      WRITE (KW, 2382)
      KPAGE = 0
      GO TO 6040
      URITE (KW, 2331)
      KPAGE = KPAGE+4
6040
      WRITE(KW, 2495)
     FORMAT(1X, 15(1H*), 25HSYNCH BRG CHARACTERISTICS, 15(1H*), 2X,
2495
     +21HSTABILITY PARAMETERS)
      IF (NSEL EQ.3) GO TO 2497
      WRITE(KW,2496)
     FORMAT(55H
2496
                  RPH
                             MIN STIFF MIN DAMP
                                                     883
                                                           STIFF
                                                                  Mad BaMP.
     +22H
           ER MASS
                         F RATIO)
      GB TD 2499
2497
      WRITE(KW,2498)
2498
     FORMAT(55H LOAD
                             MIH STIFF MIN DALL
                                                     MAJ
                                                          SYLFF
                                                                  MAJ BAMP
                         F Ratio)
     +22H
           CR MASS
      KPAGE = KPAGE+2
      DO 2500 ITAB=1, NTAB
      FF = X0(ITAB)
      CALL STAB(ZEP, FF, NSEL, HSYN, ITAB, No. 194GE)
2500
      CONTINUE
7000
      IF(NSEL.NE.1) GO TO 2510
      WRITE(1,2501)
2501
      FORMAT( /33H SELECT - (1) HNOTHER SPEED GROUP/10x.88(2) QUIT)
      READ(1,*) NCASE
      IF(NCASE, EQ. 2) GO TO 2510
      ANS=RWHD$A(KW3)
      READ(KT, 1002)(FN2(I), I=1,16), NN, AL9, ALF
      READ(KT, 1003) (X(I), I=1, NN)
      URITE(KW,2382)
      KPAGE = 0
      GO TO 2110
      ANS=CLOS$A(KW1)
      IF(ANS) GO TO 2520
      WRITE(1,2511) KW1
2511
      FORMAT('ERROR IN CLOSING FILE ON UNIT', 15)
```

- 2320 9.5#CLOS#A(KM2) 1F(ANS) 60 TO 2530 #PITC(1,2511; rN2
- \$30. An: #CLOS#ACKW7) IFCAMS\ GO TO 2540 WRITE(1:2511> KW3
- 2540 CALL EXIT

```
C
C
C
      SUBROUTINE STAB(YO, RP, NS, NY, IT, KW, KPAGE)
ε
C
      CALCULATION OF
C
                              MINOR AND MAJOR STIFFNESS COEFFICIENTS
      SYNS1,SYNS2
٤
      SYNB1,SYNB2
                              MINOR AND MAJOR CRITICAL DAMPING FACTORS
C
C
           CRMA
                                    CRITICAL MASS FOR INSTABILITY
€
           FRAT
                                    FREQUENCY RATIO OF INSTABILITY
C
           FOR SYNCHRONOUS DATA (NY.EQ.1)
C
           EE(K)
                                    MINOR/MAJOR RADIUS RATIO
           00(K)
C
                                    ORIENTATION ANGLE OF MAJOR AXIS
           FOR ASYNCHRONOUS DATA (NY.EQ.2)
C
C
      IMPLICIT REAL*8 (A-H,0-Z)
      DIMENSION Y8(8,1), Y(8), EE(2),00(2)
      PI = DATAN(1.00-00)*4.0
      IF (NS.EQ.3) GO TO 1000
      FF = PI/3.00 + 01 *RP
      GO TO 1
1000
     IF (NY.EQ.1) GO TO 1010
      FF = RP
      GO TO 1
      FF = 1.0
1010
      DO 10 I=2.8.2
      J = I - 1
      Y(J) = YO(J,IT)
10
      Y(I) = FF*YO(I,IT)
      Y34 = Y(3)**2+Y(4)**2
      IF (Y34) 20,40,20
20
      Y56 = Y(5)**2+Y(6)**2
      IF (Y56) 30,40,30
C
C
      COEFFICIENTS FOR PRINCIPAL MODE ANALYSIS
30
      SKPA = Y(1)+Y(7)
      SBPA = Y(2)+Y(8)
      DKPA = Y(1)-Y(7)
      DBPA = Y(2)-Y(8)
      XKB = Y(3)*Y(6)+Y(4)*Y(5)
      XKK = Y(3)*Y(5)
      XBB = Y(4)*Y(6)
      DKK=DKPA+DKPA+4.0*XKK
      DKB=2.0*DKPA*DBPA+4.0*XKB
      DBB=DBPA*DBPA+4.0*XBB
      CALL PRINCE(DKK, DKB, DBB, ZK, ZB)
      SYNS1=(SKPA-ZK)/2.0
      SYNS2=SYNS1+ZK.
      SYNB1=(SBPA-ZB)/2.0
      SYNB2=SYNB1+ZB
      SYNB1=SYNB1/DABS(SYNS1)/2.0
```

```
SYNB2=SYNB2/DABS(SYNS2)/2.0
      IF (NY.EQ.2) GO TO 100
      CRMA = (Y(1)*Y(8)+Y(7)*Y(2)-XKB)/SBPA
      IF (CRMA.LE.O.O) GO TO 60
      F2 = ((Y(1)-CRMA)*(Y(7)-CRMA)-XKK)/(Y(2)*Y(8)-XBB)
      IF (F2.LT.0.0) G0 T0 60
      CRMA = CRMA/F2/386.4
      FRAT = DSQRT(F2)
      GO TO 50
40
      SYNS1 = Y(7)
      SYNB1 = Y(8)
      SYNS2 = Y(1)
      SYNB2 = Y(2)
      IF (NY.EQ.2) GO TO 200
60
      WRITE(KW,6001) RP, SYNS1, SYNB1, SYNS2, SYNB2
      GO TO 55
      WRITE(KW,5002) RP, SYNS1, SYNB1, SYNS2, SYNB2, CRMA, FRAT
50
      FORMAT(1PD11.4,2(2D11.4,1X),2D11.4)
5002
5010
      KPAGE=KPAGE+1
      IF(KPAGE.LT.61) GO TO 70
      WRITE(KW,5050)
55
      KPAGE=1
      FORMAT(1H1)
5050
      FORMAT(1PD11.4,2(2D11.4,1X),23H UNCONDITIONALLY STABLE)
6001
70
      RETURN
C
      ASNYCHRONOUS MODAL DATA PRINTOUT OF THE ANISOTROPIC CASE
100
      K = 1
      KK = 1
      GX1 = (DKPA+ZK)/2.0
      GX2 = (DBPA+ZB)/2.0
      GY1 = Y(3)
      GY2 = Y(4)
105
      U0 = DSQRT(GY1*GY1+GY2*GY2)
110
      VD = DSQRT(GX1*GX1+GX2*GX2)
      YY = GX1*U0+GY1*V0
      XX = GX2*U0-GY2*V0
      AA = DATAN2(YY,XX)
      CC = DCBS(AA)
      SS = DSIN(AA)
      X1 = U0 * CC
      X2 = U0*SS
      Y1 = V0 + CC
      Y2 =- V0 + SS
      CALL WHIRL(U1,U2, V1, V2, X1, X2, Y1, Y2)
      CALL SHAPE(E.O.U1.U2.V1.V2)
      IF (E.GT.1.0) GO TO 150
      EE(K) = E
      00(K) = 1.8E + 02 + (1.0 + 0/PI)
      IF (KK.EQ.2) GO TO 266
      IF (K.EQ.2) GO TO 130
      K = 2
      GX1 = GX1-ZK
      GX2 = GX2-ZB
      GO TO 110
```

```
IF (NS.NE.3) FF = RP/6.00+01
130
      WRITE(KW,5100) FF.SYNS1,SYNB1,EE(1),00(1),SYNS2,SYNB2,EE(2),00(2)
5100
      FORMAT(1X, 1P9D10.3)
      GO TO 5010
150
      WRITE(KW,5110)
5110
      FORMAT(19H FAULTY MODAL SHAPE)
      CALL EXIT
      ASYNCHRONOUS MODAL DATA PRINTOUT OF THE ISOTROPIC CASE
      IF (Y34, NE.O.O.OR. Y56, NE.O.O) G0 TO 250
200
      EE(1) = 0.0
      00(1) = 9.00+01
      EE(2) = 0.0
      00(2) = 0.0
      GO TO 130
250
      IF (Y(1).NE.Y(7).OR.Y(2).NE.Y(8)) GO TO 260
      EE(1) = 0.0
      00(1) = 0.0
      IF (NS.NE.3) FF = RP/6.00+01
      WRITE(KW,5200) FF, SYNS1, SYNB1, EE(1), 00(1)
      FORMAT(1X, 1P4D10.3, 16H DEGENERATE MODE)
      GD TO 5010
260
      KK = 2
      IF (Y34.EQ.0.0) GO TO 270
      K \approx 1
      GX1 = Y(3)
      GX2 = Y(4)
      GY1 = Y(1)-Y(7)
      GY2 = Y(2)-Y(8)
      GO TO 105
      IF (K.EQ.2) GO TO 130
266
      EE(2) = 0.0
      00(2) = 0.0
      GO TO 130
270
      EE(1) = 0.0
      00(1) = 9.00+01
      K = 2
      GX1 = Y(7) - Y(1)
      GX2 = Y(8)-Y(2)
      GY1 = Y(5)
      GY2 \approx Y(6)
      GO TO 105
      END
```

```
CCC
      SUBROUTINE PRINCE(DKK, DKB, DBB, ZK, ZB)
      IMPLICIT REAL+8 (A-H,0-Z)
      DRE = DKK-DBB
      DIM = DKB
      DAMP = DSQRT(DRE+DRE+DIM+DIM)
      GD TO 40
      DAMP = DABS(DKK)
30
      DRE = DKK
      DIM = 0.0
40
      DARG = DATAN2(DIM, DRE)
      AMP = DSQRT(DAMP)
      ARG = DARG/2.0
      ZK = AMP+DCOS(ARG)
      28 = AMP*DSIN(ARG)
      RETURN
      END
```

```
C C C SUBROUTINE WHIRL(U1,U2,V1,V2,X1,X2,Y1,Y2) IMPLICIT REAL+8 (A-H,0-Z) C TRANSFORMS COMPLEX CARTESION REPRESENTATION (X1,X2),(Y1,Y2) C INTO COMPLEX ROTATING REPRESENTATION (U1,U2),(V1,V2) U1 = (X1-Y2)/2.0 U2 = (X2+Y1)/2.0 V1 = U1+Y2 V2 = U2-Y1 RETURH END
```

```
C
C
C
      SUBROUTINE SHAPE(E,0,U1,U2,V1,V2)
      IMPLICIT REAL+8 (A-H,0-Z)
      CALCULATES MINOR/MAJOR RADIUS RATIO AND ORIENTATION FROM COMPLEX
C
            WHIRL COMPONENTS
      00 = 01*01*02*02
      VV = V1*V1+V2*V2
      88 = 88+44
      UN = UU/NN
      48 = 44588
      IF (UW.LT.1.0D-20) GO TO 10
      IF (VW.LT.1.0D-20) GO TO 28
      U = DSQRT(UU)
      V = DSQRT(VV)
      A1 = DATAN2(U2,U1)
      A2 = DATAN2(V2,V1)
      0 = (A1-A2)/2.0
      UV = U+V
      E = (U-Y)/UY
      G0 T0 40
10
      IF (VW.LT.1.0D-20) GO TO 30
15
      0 = -4.0 * DATAN2(1.0D-00)
      60 10 40
20
      IF (UW.LT.1.0D-20) GO TO 30
      E = 1.0
      GO TO 15
30
      E = 2.0
      RETURN
40
      END
```

FUNCTION SHORT(ALT, ALD, ALF) IMPLICIT REAL*8 (A-H,0-Z) A2=ALF*ALF73.0 KTIME=1 X=ALF*ALT XX=2.0*X 10 YY=DEXP(-XX) T=(1.0-YY)/(1.0+YY) A=(1.0-T/X)/A2 IF(KTIME EQ.2) GO TO 28 A1=A KTIHE=2 X=ALF*ALD GO TO 10 SHORT=A1/A 20 RETURN END

C C C FUNCTION NUMERE(XX,X,NM) IMPLICIT REAL*8 (A-H,0-Z) DIMENSION X(1) N1=0 MM = NM OO 10 I=2,MM Y=XX-X(I) IF(Y,GT.0.0) GO TO 10 N1=I-1 MM=I CONTINUE IF(N1.EQ.0) N1=MM NUMERE=N1

RETURN END FUNCTION SPL(ZZ,Y1,DX,H)
IMPLICIT REAL+8 (A-H,0-Z)
DIMENSION ZZ(1),Y1(3,1)
SPL=ZZ(N)+DX*(Y1(1,N)+DX*(Y1(2,N)+DX*Y1(3,N)/3.0)/2.0)
RETURN
END

```
C
C
C
      FUNCTION REFER (AA, BB, S1, S2, W)
C
      CALCULATES REFERENCE FUNCTION
C
      IMPLICIT REAL+8 (A-H,0-Z)
      IF ($1.LT.0.0) G0 TO 100
      W1=DABS(AR)
      W2=DABS(BB)
      IF(W.EQ.0.0) GO TO 30
      ¥1=¥1*(¥**S1)
      W2=W2*(W**52)
      S3=S1-S2
      IF($3.GT.0.0) GO TO 10
      IF($3.GT.-1.00-00) W2=W2*W
      REFER=W1+W2
      IF($3.LE.-1.0D-00) GO TO 20
      REFER=REFER/(1.00-00+W)
      GO TO 20
10
      REFER = 41*42*(1.00-00+4)/(4*41+42)
50
      RETURN
30
      IF($1.GT.0.0) GO TO 40
      REFER = W1
      G0 T0 20
      REFER = 0.0
40
      GO TO 20
100
      WRITE(1,1001)
1001
      FORMAT(/35H FAULTY DATA, NEAR FIELD EXPONENT =,1PD12.4)
      CALL EXIT
      END
```

```
C
C
      SUBROUTINE TILT(Y, ROT, IT)
      ROTATIONAL TRANSFROMATION OF IMPEDANCE MATRIX
C
      IMPLICIT REAL+8 (A-H,0-Z)
      DIMENSION ROT(1), Y(8,1)
      YK11 = Y(1,IT)
      YB11 = Y(2,IT)
      YK22 = Y(7,IT)
      Y822 = Y(8,IT)
      YK12 = Y(3,IT)
      Y812 = Y(4,IT)
      YK21 = Y(5,IT)
      YE21 = Y(6,IT)
      DKPA = YK11-YK22
      DBPA = YB11-YB22
      SKXX = YK12+YK21
      SBXX = YB12+YB21
      Y(1,IT) = YK11*R0T(1)-SKXX*R0T(2)+YK22*R0T(3)
      Y(2,IT) = YB11*R0T(1)-SBXX*R0T(2)+YB22*R0T(3)
      Y(3,IT) = DKPA*R0T(2)+YK12*R0T(1)-YK21*R0T(3)
      Y(4,IT) = DBPA*R0T(2)+YB12*R0T(1)-YB21*R0T(3)
      Y(5,IT) = Y(3,IT)-YK12+YK21
      Y(6,IT) = Y(4,IT)-YB12+YB21
       Y(7,IT) = -Y(1,IT) + YK11 + YK22
       Y(8,IT) =-Y(2,IT)+YB11+YB22
```

C

RETURN END

```
C
      SUBROUTINE BASE(HBA, BMA, BST, BDA)
C
      PEDESTAL PROPERTIES
      IMPLICIT REAL+8 (A-H,C-Z)
      DIMENSION BMA(1), BST(1), BDA(1), BIN(2), R2(2)
      URITE(1,1001)
      FORMAT(/29HSELECT: (1) RIGID FOUNDATION/9X,13H(2) ISOTROPIC,
     +26H FOUNDATION COMPLIANCE, OR/9%, 27H(3) ANISOTROPIC FOUNDATION ,
     +11HCOMPLIANCE?)
      READ (1,+) HBA
      IF (NBA.EQ.1) GO TO 1220
      WRITE(1,1002)
     FORMAT(/31HSELECT: (1) RADIAL BEARING, OR/9X,12H(2) ANGULAR,
1002
     ⇒SHBEARIKG?)
      READ (1,+) HTY
      WRITE(1,1003)
1003
      FORMAT(5%, 27HENTER PEDESTAL WEIGHT (LB)?)
      READ (1,*) BUT
      IF (NTY_EQ.2) GG TO 105
      BMA(1) = BVI/386.4
      BMA(2) = BMA(1)
      GD TO 120
105
      WRITE(1,1604)
1084
      FORMAT(5x,32HENTER CG OFFSET OF PEDESTAL (IN)/5x,6HAXIAL?,4x,
     +21HVERTICAL? HORIZONTAL?)
      READ (1,+) 02,0X,0Y
      RRX = 02*0Z
      RRY = 0Z*0Z
      IF (NBA.EQ.2) GO TO 106
      RRX = RRX+0X+0X
      RRY = RRY+0Y+0Y
106
      R2(1) = RRX*BUT
      R2(2) = RRY*BUT
      WRITE(1,1005)
      FORMAT(5x,49HENTER MOMENT OF INERTIA IN VERT PLANE (LB-SQ IN)?)
1005
      READ (1,*) BIN(1)
      IF (NBA.EQ.3) GO TO 1006
      BIN(2) = BIN(1)
      GO TO 1010
1006
      WRITE(1,1007)
1007
      FORMAT(5X,50HENTER MOMENT OF INERTIA IN HORIZ PLANE (LB-SQ IH)?)
      READ (1,+) BIK(2)
1010
      DO 110 I=1.2
      BMA(I) = (BIN(I)+R2(I))/386.4
110
      CONTINUE
      WRITE(1,1201)
120
      FORMAT(5X,43HENTER PEDESTAL STIFFNESS IN VERTICAL PLANE?)
1201
      URITE(1,1202)
      FORMAT(5x,50H(LB/IN) FOR RADIAL BRG, OR (IN-LB/RAD) FOR ANG BRG)
1202
      READ (1,+) BST(1)
      IF (MBA.EQ.3) GO TO 1203
      BST(2) = BST(1)
```

GO TO 1210 1203 WRITE(1,1204) FORMAT(5x,45HENTER PEDESTAL STIFFNESS IN HORIZONTAL PLANE?) 1204 WRITE(1,1202) READ (1,*) BST(2) 1210 WRITE(1,1211) 1111 FORMAT(5X,41HENTER PEDESTAL DAMPING IN VERTICAL PLANE?) URITE(1,1212) 1212 FORMAT(5x,52H(LB-SEC/IN) FOR RAD BRG, OR (IN-LB-SEC/RAD) FOR ANG, +3HBRG) READ (1,*) BDA(1) IF (NBA.EQ.3) GO TO 1213 BDA(2) = BDA(1)GO TO 1220 1213 WRITE(1,1214) FORMAT(5X,43HENTER PEDESTAL DAMPING IN HORIZONTAL PLANE?) 1214 WRITE(1,1212) READ (1,*) 8DA(2) 1220 RETURN END

```
C
C
C
      SUBROUTINE GOAT(FF,Y,IFRE,8M,8S,8D)
C
      SUMMATION OF BEARING AND PEDETAL COMPLIANCES
      IMPLICIT REAL *8 (A-H, 0-Z)
      DIMENSION Y(8:1), BN(1), BS(1), BD(1)
      DIMENSION AB(2,2),BB(2,2),BZ1(2),BZ2(2),DZ1(2),DZ2(2)
      DIMENSION CB(2,2), DB(2,2), EB(2,2), FB(2,2)
      ASSEMBLE BEARING IMPEDANCE
C
C
      RP = FF*DAYAN(1.00-00)/7.50-00
      K = 0
      DO 10 I=1,2
      DO 10 J=1,2
      K = K+1
      KK = 2*K
      K1 = KK-1
      CB(I,J) = Y(K1,IFRE)
      DB(I,J) = Y(KK,IFRE)*RP
10
      CONTINUE
C
      CALCULATE PEGESTAL IMPEDANCE ELEMENTS
C
      RP2 = RP*RP
      DO 20 I=1,2
      BZ1(I) = BS(I)-BM(I)*RP2
      BZ2(I) = BD(I)*RP
20
      CONTINUE
      COMBINE PEDESTAL AND BRG IMPEDANCES
      DO 30 I=1,2
      D0 30 J=1.2
      IF (I.EQ.J) GO TO 25
      AB(I,J) = CB(I,J)
      BB(I,J) = DB(I,J)
      GO TO 36
25
      AB(I,J) = CB(I,J)+BZ1(I)
      BB(I,J) = DB(I,J)+B22(I)
30
      CONTINUE
C
      INVERT
      DD1 = AB(1,1)*AB(2,2)~BB(1,1)*BB(2,2)
           -AB(1,2)*AB(2,1)+BB(1,2)*BB(2,1)
      DD2 = AB(1,1)*BB(2,2)+BB(1,1)*AB(2,2)
           -AB(1,2)*BB(2,1)-BB(1,2)*AB(2,1)
      DD = DD1*DD1+DD2*DD2
      DD1 = DD1/DD
      DD2 = -DD2/DD
      D0 40 I=1.2
      K = 3-I
      EB(I,I) = DD1*AB(K,K)-DD2*BB(K,K)
      FB(I,I) = DD1*BB(K,K)+DD2*AB(K,K)
      EB(I,K) = -DD1*AB(I,K)+DD2*BB(I,K)
      FB(I,K) = -DD1*BB(I,K)-DD2*AB(I,K)
```

```
40
      CONTINUE
C
      POST MULTIPLY BY BRG IMPEDANCE
C
      DO 50 I=1,2
      00 50 J=1,2
      AB(I,J) = 0.0
      88(1,J) = 0.0
      00 50 K=1,2
      AB(I,J) = AB(I,J)+EB(I,K)*CB(K,J)-FB(I,K)*DB(K,J)
      BB(I,J) = BB(I,J)+EB(I,K)*DB(K,J)+FB(I,K)*CB(K,J)
50
      CONTINUE
      PRE MULTIPLY BY PEDESTAL IMPEDANCE AND RESTORE Y
C
      K = 0
      00 60 I-1,2
      DO 60 J=1.2
      K = K+1
      KK = 2*K
      K1 = KK-1
      Y(K1, IFRE) = BZ1(I)*AB(I, J)-BZ2(I)*BB(I, J)
      Y(KK, IFRE) = (BZ1(I)*BB(I,J)*BZ2(I)*AS(I,J))/RP
60
      CONTINUE
      RETURN
      END
```

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